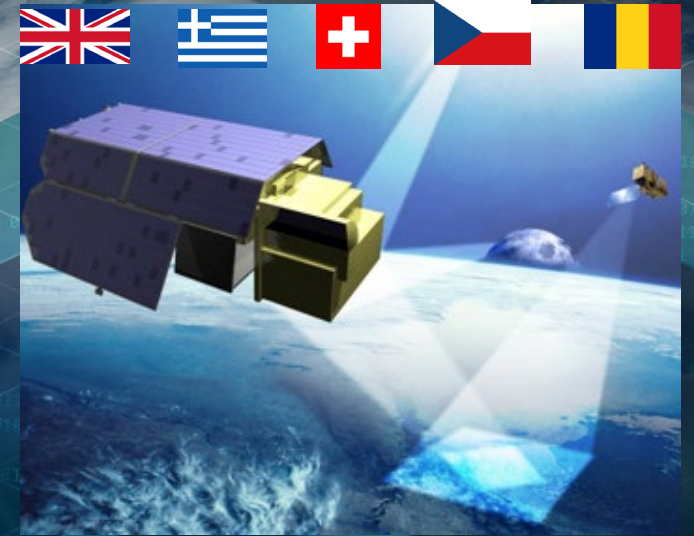


living planet symposium | BONN

23-27 May
2022

TAKING THE PULSE
OF OUR PLANET FROM SPACE



Traceable Radiometry Underpinning Terrestrial- and Helio- Studies (TRUTHS) – A ‘gold standard’ reference for an integrated space-based observing system for environment and climate action

Nigel Fox, Paul Green, Samuel Hunt, NPL,
Andrea Marini, Thorsten Fehr, ESA
Kyle Palmer, Airbus

27 May 2022

ESA UNCLASSIFIED – For ESA Official Use Only

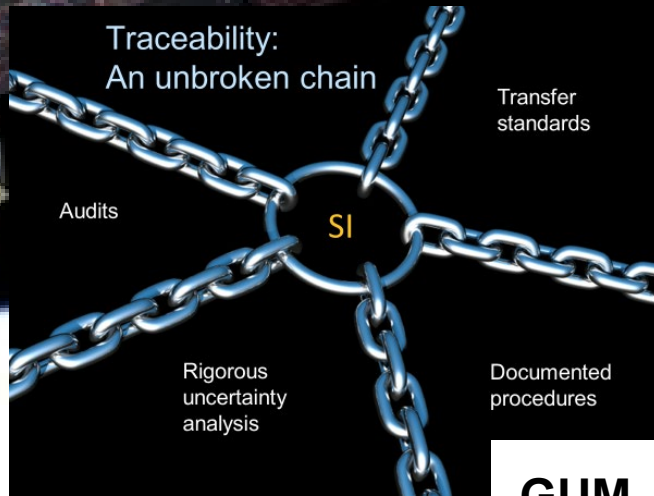
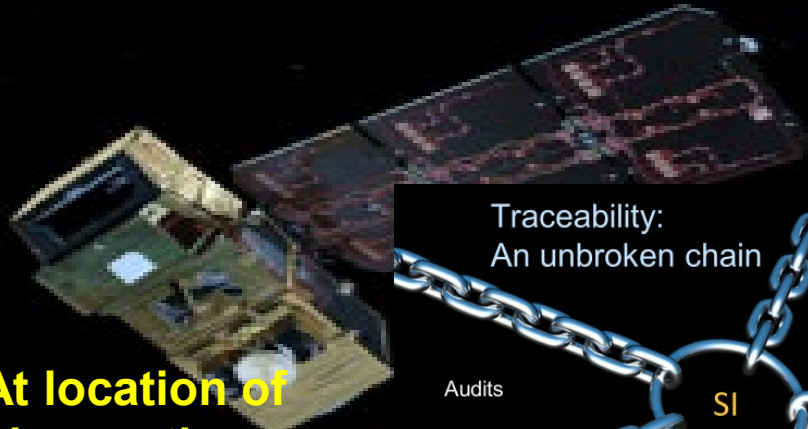


Long-time-base & societal/economic critical applications like Climate require robust 'trustworthy' data:



e.g detection of small signals of change out of a noisy background of nat variability

At location of observation

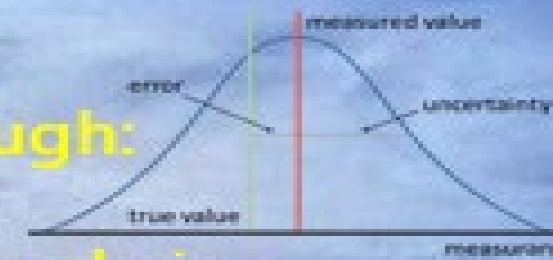


- Interoperable
- Century-long stability
- Coherence
- Absolute accuracy

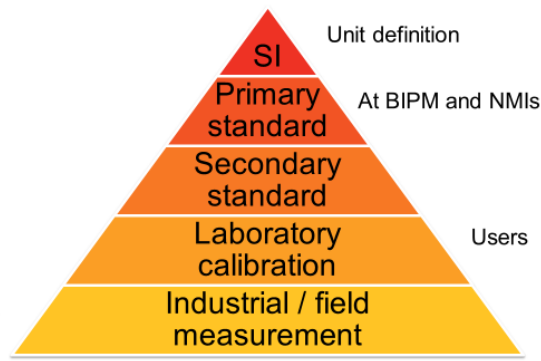
Achieved through:

- Traceability
- Uncertainty Analysis
- Comparison

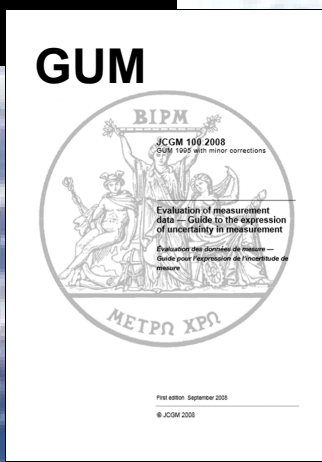
All documented and accessible



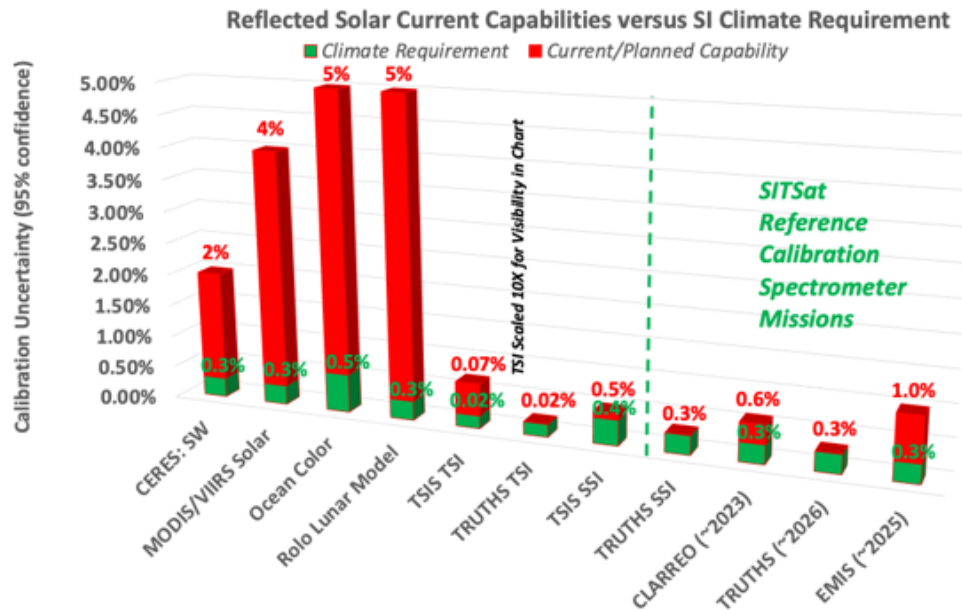
Traceability



GUM



Climate Need & observation challenges



Libya 4, a CEOS PICS site is a desert with a stable long-term reflectance. Vegetation index measured below should be horizontal red line i.e. none!

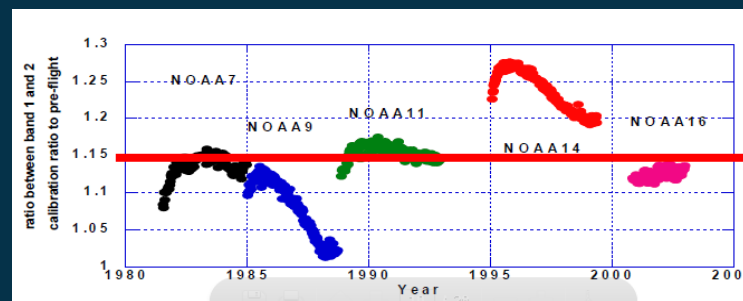
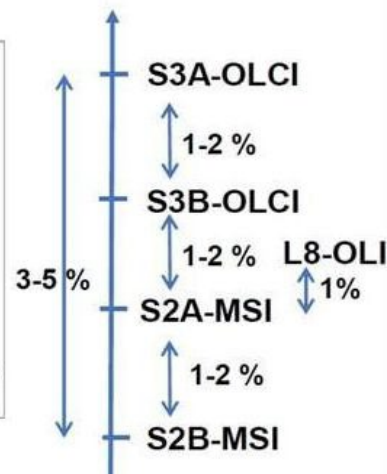
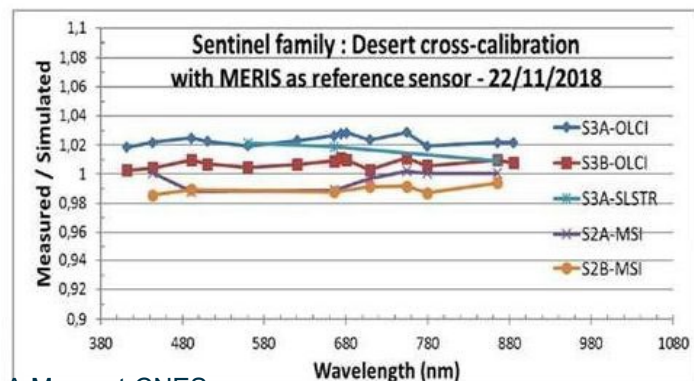
CEOS
SI-Traceable Space-based Climate Observing System:
a CEOS and GSICS Workshop
National Physical Laboratory,
London, UK,
9-11 Sept. 2019

SITSOS Workshop Report



Editors: Nigel Fox, Tim Hewison, Greg Kopp, Bruce Wielicki
<https://doi.org/10.47120/npl.9319>

Most satellites not designed for climate:
performance to suit application



Trustable harmonised time series
require stable/understood
sensors anchored to invariant
references

<http://calvalportal.ceos.org/report-and-actions>

<https://doi.org/10.47120/npl.9319>

What is the Truth?

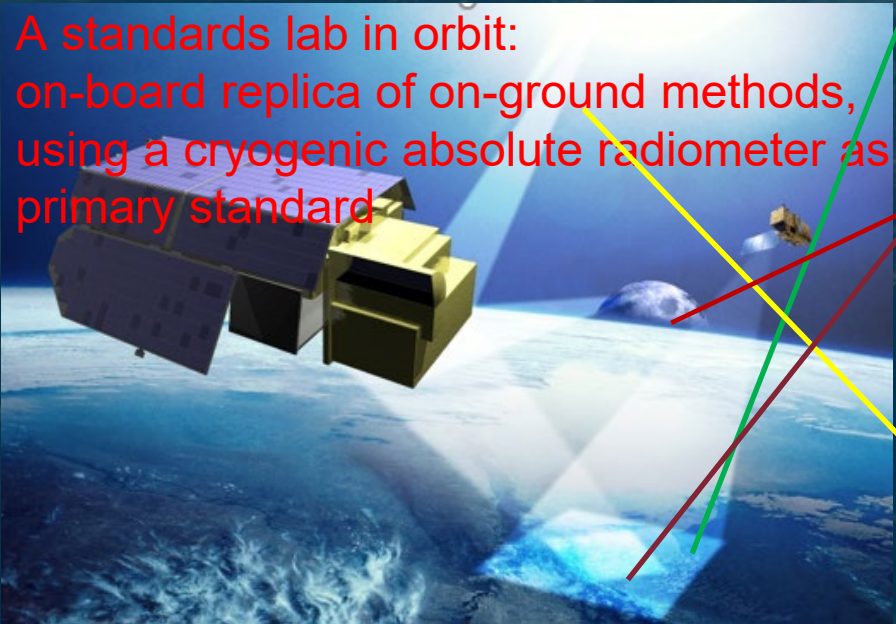


SITSats and TRUTHS Mission Objectives

What is a SITSat?: 'Space borne missions specifically designed, characterised and documented to provide **high accuracy SI-Traceable** 'reference' measurements.' (Evidencing comprehensive uncertainty to SI, 'in-space', of all contributors to observations made from the satellite)

TRUTHS is an **operational climate mission**, aiming to:

A standards lab in orbit:
on-board replica of on-ground methods,
using a cryogenic absolute radiometer as
primary standard



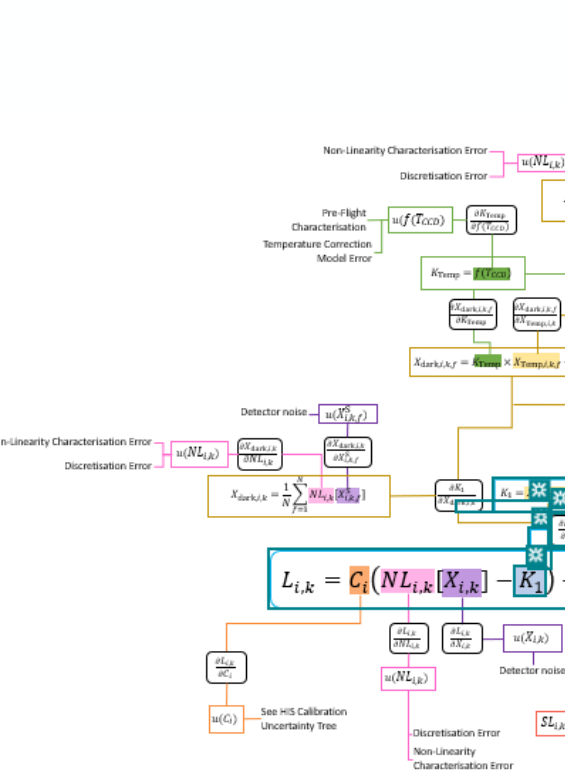
1. **Climate benchmarking:** enhance by an order-of-magnitude our ability to estimate the **Earth Radiation Budget** (and attributions) through direct measurements of incoming & outgoing energy and reference calibration.

2. **Satellite cross-calibration:** establish a 'standards laboratory in space' to create a '**gold standard**' reference data set to cross-calibrate other sensors and improve the quality and interoperability of their data-simultaneous observations, surface reference sites and the moon

3. provide SI-traceable measurements of the **solar spectrum (incoming & reflected)** to address its impact on climate and interactions with the atmosphere and surface

A **benchmark measurement** is one with characteristics (documentation, SI-Traceable uncertainty, representative sampling) that allows it to be unequivocally considered a 'reference' of the specified measurand against which future measurements of the same measurand, can be compared.

Open access data with full transparency of uncertainties and traceability – a ‘metrology lab in space’



CSAR power correction						
Name of effect	Cavity temperature characterisation error (Shuttered)	Cavity temperature characterisation error (open)	Sensor reference resistor voltage fluctuation error	Reference resistor characterisation error	Cavity sensitivity characterisation error	
Affected term in measurement function	$T_{cav, c}$	$T_{cav, i}$	$V_{r_cav, i}$	$R_{r_h_cav}$	S_{cav}	
Image error-correlation type, form and scale	Across track (k)	Systematic	Systematic	Random	Systematic	Systematic
	Along track (f)	Systematic	Systematic	Systematic	Systematic	Systematic
	Between spectral pixels (i)	Systematic	Systematic	Systematic	Systematic	Systematic
Uncertainty	PDF shape	Rectangular	Rectangular	Gaussian	Rectangular	Rectangular
	units	K	K	V	Ω	KV^{-1}
	magnitude	TBD	TBD	TBD	TBD	TBD
Source						
Sensitivity coefficient	$\frac{\partial L_{i,k}}{\partial T_{cav, c}}$	$\frac{\partial L_{i,k}}{\partial T_{cav, i}}$	$\frac{\partial L_{i,k}}{\partial V_{r_cav, i}}$	$\frac{\partial L_{i,k}}{\partial R_{r_h_cav}}$	$\frac{\partial L_{i,k}}{\partial S_{cav}}$	

Neighbouring Pixel Detector Noise

Fahy & Hunt

TRUTHS HIS Earth Radiance Uncertainty Tree

‘effects table’ for each error source

- Upto ~4 Tbytes a day download
- UK ‘Data processing hub’
- Baseline ‘archived’ Level 1B products
 - ToA Earth, Moon, Sun
- Toolbox to enable customisable ‘on-demand’ processing of Level 1B, 1C and 2 (BoA reflectance) with pixel level uncertainties
- Coefficients to facilitate sensor SI-traceability and/or recalibration
- Unequivocal litigation quality data

Fiduceo like analysis of end to end traceability and uncertainties – an exemplar for other missions



What does TRUTHS do?

Measures incoming and earth reflected radiation from the sun

- 320 to 2400 nm @ ~4 nm intervals (1 nm for solar UV)
- Global nadir @ 50 m ground resolution with 100 km swath (capability)
- Target uncertainty of 0.3% (k=2)

Establishing a benchmark of the state of the planet at ToA & BoA surface reflectance to enable:

Calibration

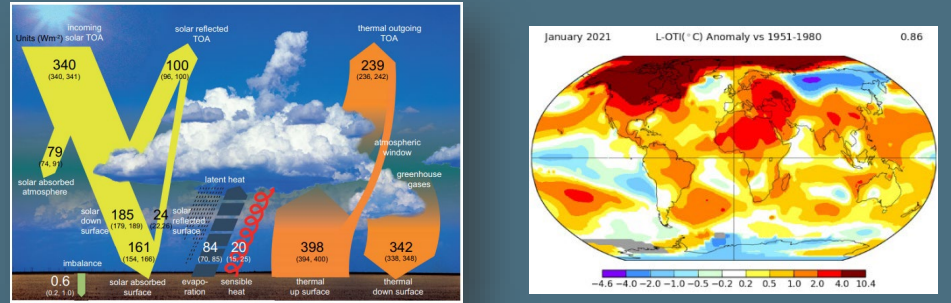
- Interoperability
- data-gaps
- performance
- Utility

Observations

- Benchmark
- monitoring
- Litigation
- algorithm improvement

Climate action: Supporting 'Net Zero'

Climate sensitivity/response



Adaptation/sustainability



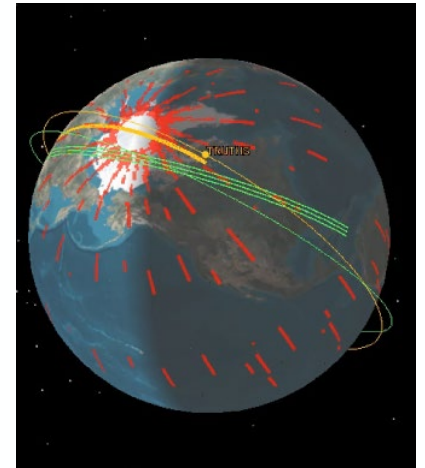
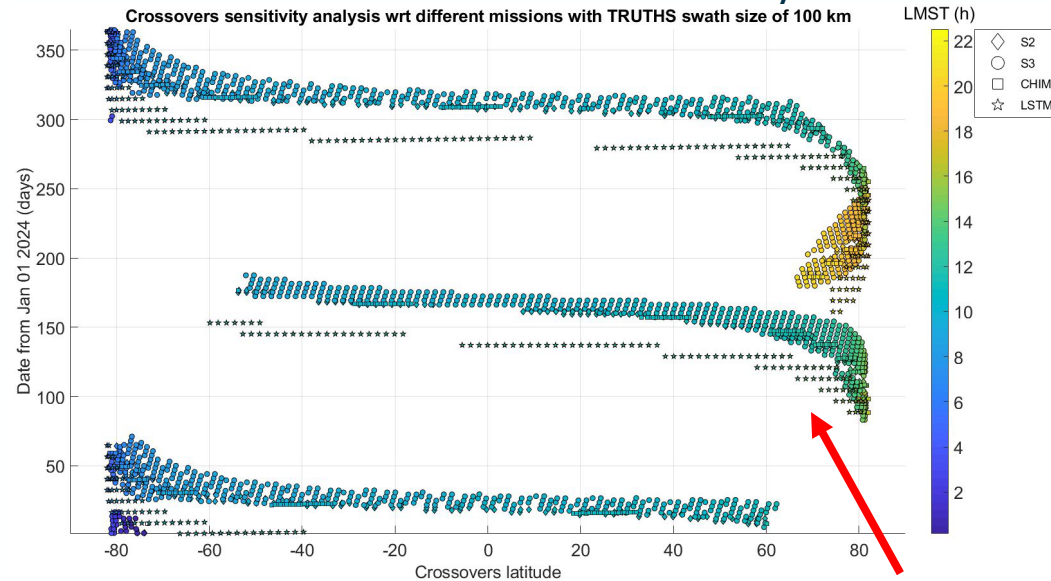
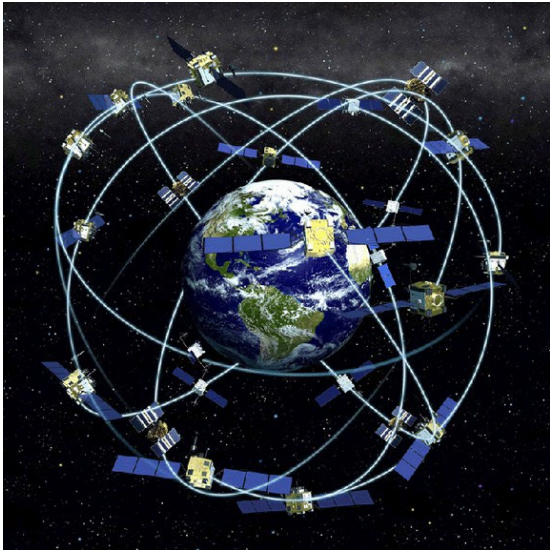
Climate action/mitigation



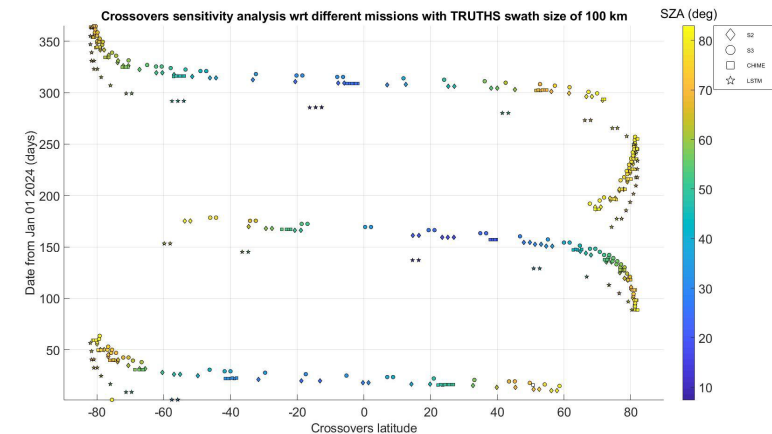
Reference Calibration

- Enables interoperability & Harmonisation
 - Prospect of 'certified calibration'

TRUTHS 90° pole to pole orbit, observing through the diurnal cycle, allows many opportunities to overpass orbit of sun-synchronous sensors



Summary after 6 months



TRUTHS provides the means to transform global EO system, including constellations of micro-sats so they deliver traceable scientific/climate quality observations -

<30 s time difference Swath overlap

1 year of near perfect nadir overlaps for TRUTHS & satellite under test

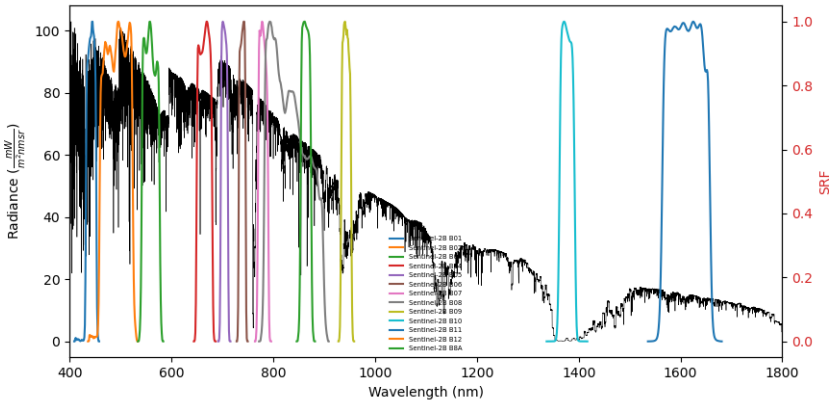
(<1° (no pointing) <30 s time difference

Transferring TRUTHS accuracy to other Sensors

(Fahy, Hunt, Stedman, Gorrone)

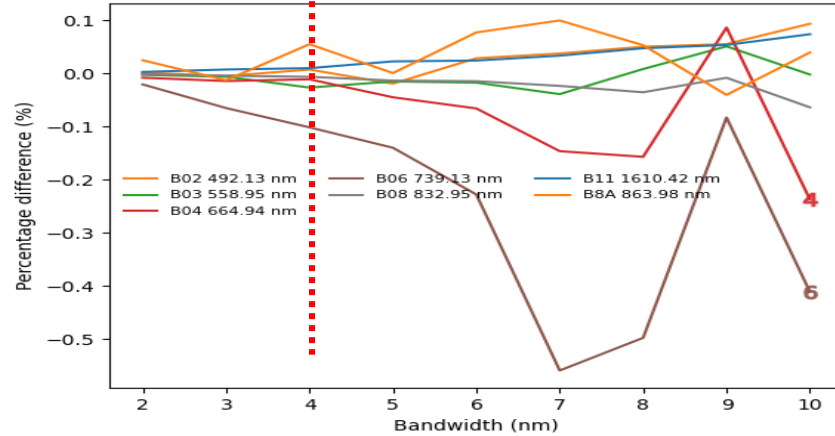


Graph of Sentinel-2B msi SRF overlaid on simulated TOA radiance spectrum (from libRadtran)

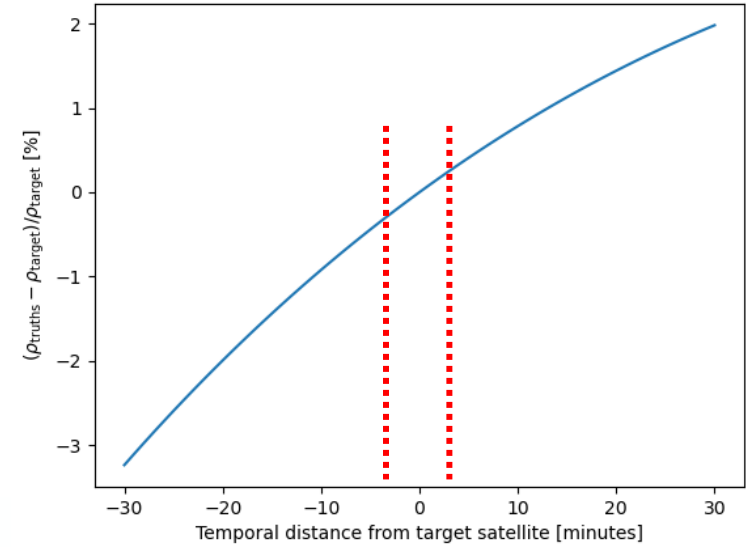


Spectral Bands of Sentinel 2

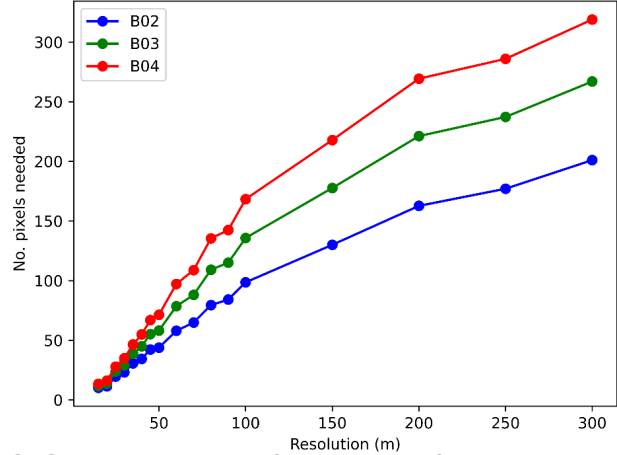
Graph of percentage difference between Sentinel-2B msi band integrated radiances using TRUTHS radiances vs libradtran simulated radiances (BW = SSI)



Impact of TRUTHS bandwidth

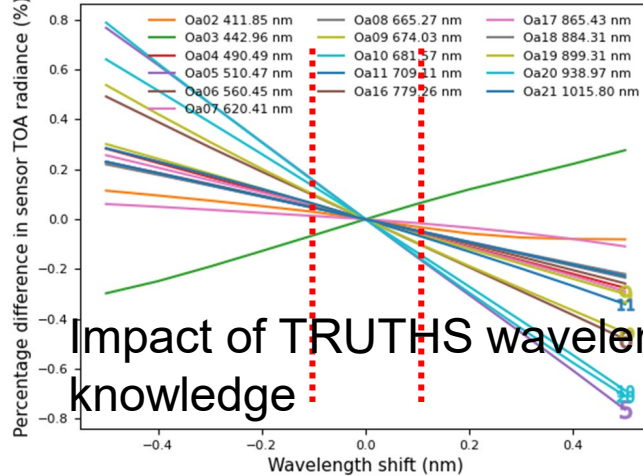


No. pixels needed to achieve u=0.2% for Libya ROI (5490 x 5490)

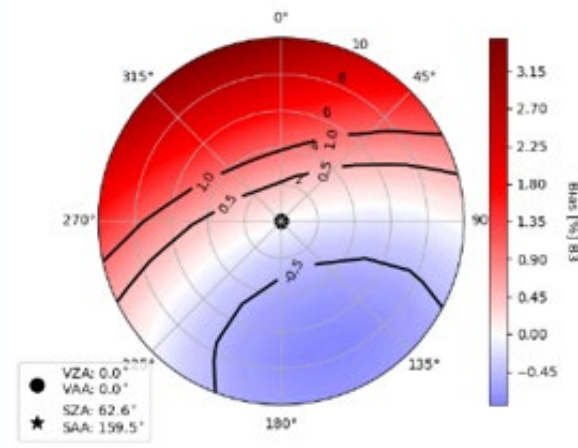


Uniformity of Cal Target (Libya 4) area to be sampled (2.5 km to achieve 0.2% @ 50 m)

Graph of percentage difference in Sentinel-3A olci TOA radiance due to wavelength shift in TRUTHS (SSI = 4 nm, BW = 4 nm): Ocean



Impact of TRUTHS wavelength knowledge



(d) MODIS B3 (460nm); AOT (550nm) = 0.4

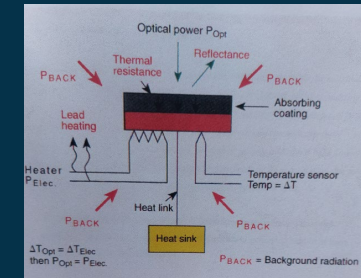
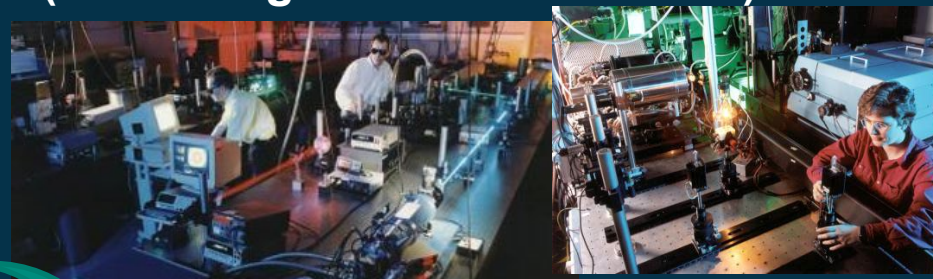


Uncertainty budget for TRUTHS – satellite comparisons (Gorrone et al)

Uncertainty	Best S2 bands	Worst S2 bands
Spectral resolution TRUTHS	0.1 %	0.6 %
Spectral accuracy TRUTHS	0.1 %	0.2 %
Spatial co-alignment mismatch	0.1 % (Libya) 0.12 % (La Crau)	0.1 % (Libya) 0.5 % (La Crau)
30 minute time difference (atmospheric effects)	0.1 % (if corrected) 0.3 % (if atmosphere not known)	0.1 % (if corrected) 2 % (if atmosphere not known)
30 minute time difference (surface BRF)	0.2 %	0.4 %
Combined with reasonable corrections	0.4 % - 0.5 %	0.7 %

Metrology laboratory in-space

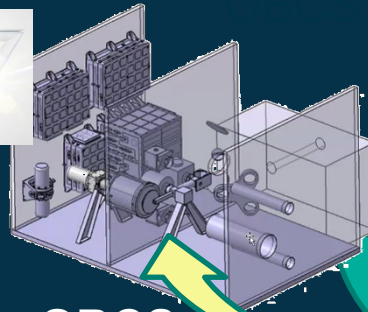
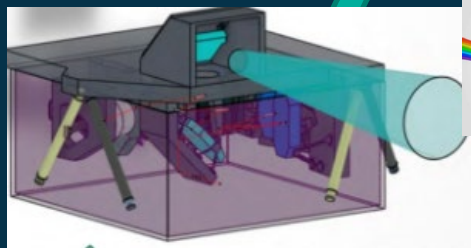
- Measuring energy from the sun, providing the direct traceability to International Standards (CSAR)
 - **Compares heating effect of optical power with electrical power (Volt)**
- ‘Camera’ (Hyperspectral Imaging Spectrometer, HIS) observing the direct incoming and Earth reflected sunlight at high spectral and spatial resolution
- Novel on-board calibration system (OBCS) ensuring traceability to the absolute reference (Cryogenic, 60K, Solar Absolute Radiometer, CSAR) (mimicking terrestrial methods)



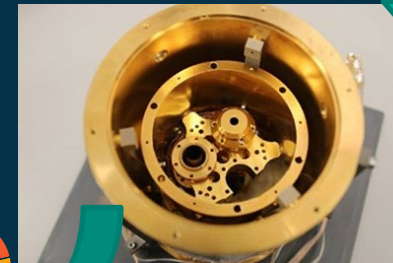
Link to SI
= Volt



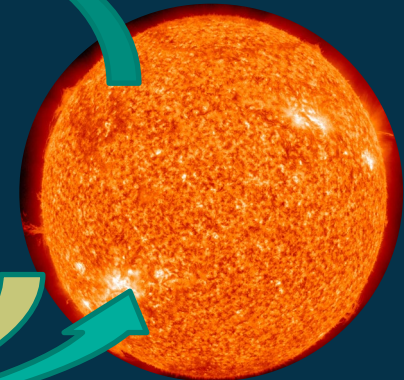
HIS



OBCS



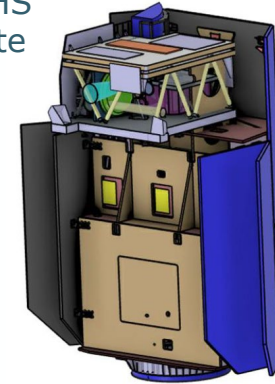
CSAR



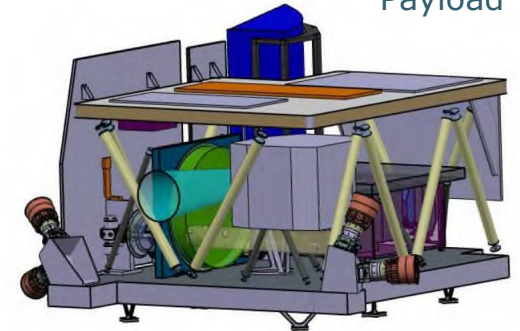
Completing Phase A/B1 to provide a mature concept

- Agile Platform recurrent from **CRISTAL**
- Payload key technical features:
 - ✓ **HIS**: Four-mirror anastigmatic telescope, Offner (two-prisms) spectrometer, single MCT detector at 150 K, thermally stable optical bench.
 - ✓ **CSAR**– three high-absorbance cavities, operated at 60 K with cryocooler, design heritage of NPL (UK) and PMOD/WRC (CH)
 - ✓ **OBCS** (On-Board Calibration System) – traceable set of absolute wavelength anchors (solar monochromator + TBD filter), high-dynamic transfer radiometer, precise and stable wavelength scanning mechanism, relay optics, diffuser to HIS
 - ✓ **Calibration process**: novel methodology, heritage of metrology lab (NPL), rigorous traceability of uncertainties, need for demanding on-ground calibration
- Pre-developments running for all critical items (detector, coating, CSAR, mirror, calibration detectors...)
- Phase A/B1 led by Airbus UK progressing as planned with final review, the ISRR, started End-April 22 – completes Mid June.
- Gate review (as defined in the EW element) July 22 to confirm technical, scientific and programmatic maturity of the proposed solution ready for subscription in CM 22

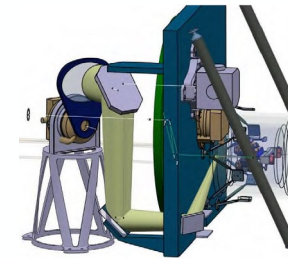
TRUTHS satellite



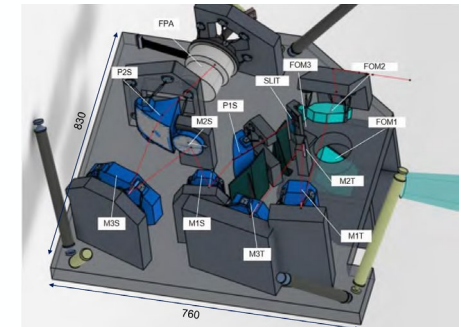
TRUTHS Payload



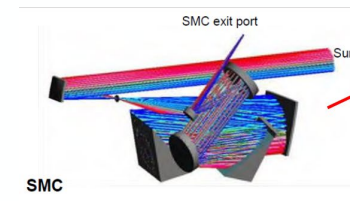
CSAR layout/OBCS I/F



HIS layout

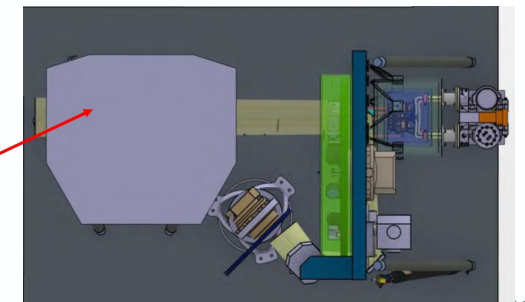


SMC - OBCS layout

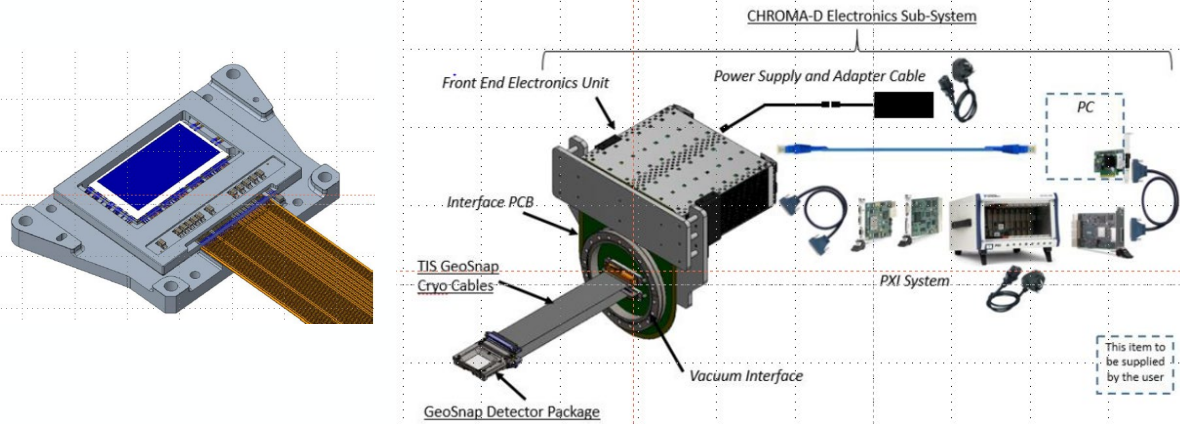


SMC

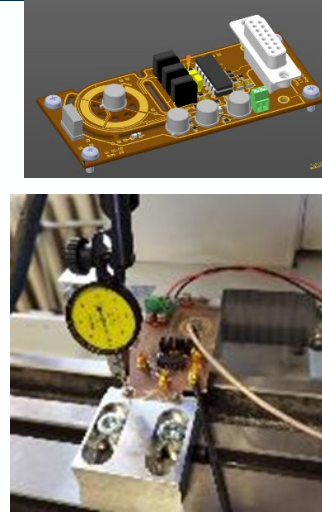
All images courtesy of Airbus, PMOD, NPL



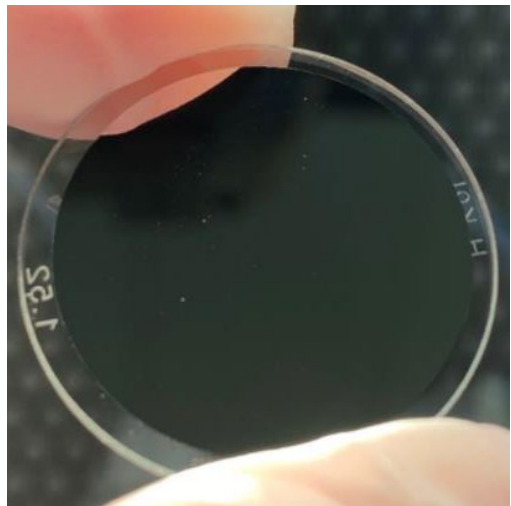
T-E2V Detector layout and FEE test assembly



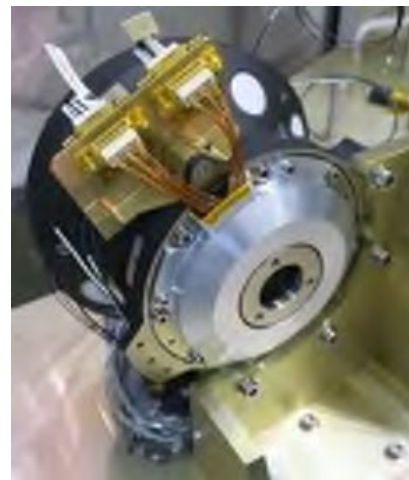
CSAR Voltage Ref: LTZ1000 Board



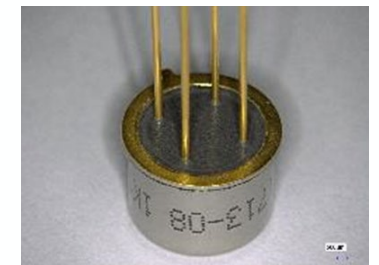
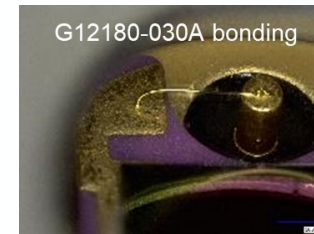
Mirror coatings



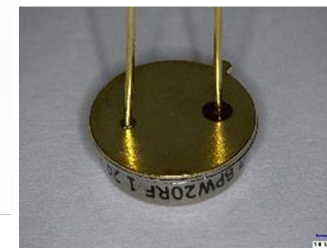
WSM actuator: TRISHNA EM



CA OBCS TR photodiodes



Visual Inspection



K -1713-08 dual-colour



X-ray

CPF & TRUTHS: Similarities & differences



Characteristic	CPF	TRUTHS (some characteristics may change as design develops)	Comment
Platform/orbit	Int Space Station/ 52°	Own sat / 90°	Both asynchronous TRUTHS full globe
Operational / lifetime	2024 (1 yr seeking 5)	2030 (5 – 8 yrs)	Potential overlap if CPF life extended (5+ yrs needed for benchmark)
Spectral range	350 -2300 nm	320 – 2400 nm	
Spectral Resolution	3-6 nm	2–6 nm Earth 1-8 nm Sun	Bandwidths to facilitate cross-calibration
Radiometric accuracy (goal)	0.6 % (k=2)	0.3 % (k=2) 0.02 % (k= 2) (total solar irradiance)	Spectrally resolved and band integrated
IFOV / Swath	500 m / 70 km	50 m / 100 km	50 m to additionally target land imagers, new space, EO applications
Measurands (direct)	Earth/lunar spec reflectance	Earth spec (radiance/reflectance) Solar/lunar Spec (irradiance) Total Solar Irradiance	CPF can obtain radiance/irradiance through solar values from TSIS mission
Route to SI-Traceability	Ratio (sun to earth) with attenuation using apertures & time integration	On-board calibration system mimicking ground including primary standard	Comparison of different methods provides opportunity for rigorous confirmation of uncertainties



Phase AB1 team



Agencies

Swiss Confederation
 Confédération suisse
 Confederaziun Svizra
 Confederaziun svizra

CSO
 COSMIC SPACE OFFICE

EAAO
 European Agency for the Space Programme

rosa

esa

UK SPACE AGENCY

AIRBUS

deimos
 elecno group

ThalesAlenia
 Space

SONOVISION

INASCO

RAL Space

pmod wrc

ISD SA
 Integrated Systems Development

planetek
 hellas

CGI

NPL

SURREY

UNIVERSITY OF
 LEICESTER

GOONHILLY
 SPACE ESTABLISHMENT

toptec

SWISSOPTIC
 BERLINER GLAS GROUP

TELESPIAZIO
 a LEONARDO and THALES company

TELEBYNE Q2V
 Evrything you do look for in the Space Group/Group

Science institutes

National Centre for Earth Observation
 NATURAL ENVIRONMENT RESEARCH COUNCIL

Swansea University
 Prifysgol Abertawe

Universität Zürich

Imperial College London

UCL

Industrial Consortium

ThalesAlenia Space
 Mission System & Instrument tasks

deimos
 elecno group

ISD SA
 Integrated Systems Development

NPL
 RAL Space

planetek

OPSI/E2EMS

NPL
 RAL Space

Calibration Activities

NPL
 RAL Space

CGI
 TELESPIAZIO

Ground Segment

Skills & capability Development

Phase A/B1 SFS

esa

AIRBUS
 TRUTHS Mission System & Instrument Prime & Ground Segment lead

AIRBUS

SURREY

Optics Centre of Competence support

ThalesAlenia Space
 PMCS/OBCS/CSAR B1 ex

NPL
 E2EMS ABTDs B1 ex

planetek
 OPSI Software B1 ex

Mandatory Option B1 Extension

TELEBYNE Q2V
 EBB and Flight Detector Development

INASCO
 EBB structure & MGSE

toptec
 Mirror engineering

Optical elements (to be selected post KO)

pmod wrc
 CSAR pre-development

NPL
 S2CS simulator

NPL
 CSAR/OBCS consultancy and predevelopments

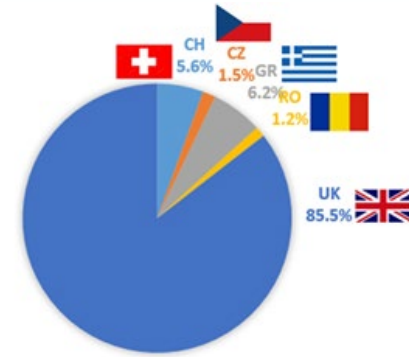
Pre-development activities

Pre-Development Activities



TRUTHS Timeline

- TRUTHS was proposed by UKSA in May 2019 as a new Earth Watch (EW) Element.
- TRUTHS Phase A/B1 has been fully subscribed at Space19+ by 5 Participating Countries: UK (85.5%), GR (6.2%), CH (5.6%), CZ (1.5%), RO (1.2%)
- Industrial Phase A/B1 system studies and technology predevelopments initiated in Oct-20.
 - Phase-A kicked-off in Oct-20 and completed at end-July 2021
 - Phase B1 on-going, to be completed in Q2-2022.
- Mission Advisory Group (MAG): Science/Engineering/User expertise primarily from Europe (not limited to funding nations) inc NASA CLARREO Pathfinder
- TRL and SRL assessments in May/June 2022
- Programmatic “Gate Review”: go/no-go decision, in July-22, to submit program to CM-22
- Phase B2/C/D/E to be funded at CM-22/-25 -> Program plan being currently prepared



TRUTHS A/B1 SUBSCRIPTION - @SPACE19+

