# GOSAT-2/TANSO-CAI-2 <br> Level 1 Data Description Document 

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## GOSAT-2/TANSO-CAI-2 Level 1 Data Description Document

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## 1. Introduction

### 1.1. Outline

The GOSAT-2 mission is aimed at continuing and advancing GOSAT mission and continuously providing useful information that contributes to environmental decision making for global warming.

GOSAT-2 project is promoted under the cooperation between the Ministry of the Environment (MOE), the Japan Aerospace Exploration Agency (JAXA) and the National Institute for Environmental Studies (NIES).
JAXA implements Level 1 processing of GOSAT-2 data. The Level 1 products based on the observation data of GOSAT-2 is processed by GOSAT-2 Mission Operations System.

This document describes the format of TANSO-CAI-2 Level 1 following products generated by GOSAT-2 Mission Operations System.

- Level 1A product
- Level 1A calibration product

TANSO-CAI-2 Level 1 products are stored in HDF5 (Hierarchical Data Format Version 5). They are produced with HDF5 library.

### 1.2. Baseline Documents

Following documents give the baseline for the design of products:
(1) HDF5

- HDF5 Reference Manual (Release 1.8.18)
- HDF5 User's Guide(Release 1.8.18)
(2) Engineering Specification Document (ESPC)
- ESPC for global earth observation data processing system (for GOSAT-2), in Japanese
- Definition of GOSAT-2 Level 1 Products, in Japanese


## 2. Overview of products

### 2.1. Definition of processing level

Processing of TANSO-CAI-2 level 1 product is defined as follows:
Level 1A processing:

Level 1A products contains uncorrected image data of TANSO-CAI-2, which is stored as digital number together with telemetry of geometric information at observation point, orbit and attitude data, temperature, etc. Uncorrected image data in the product is digital value output from the sensor

Table $2.1-1$ shows definitions of TANSO-CAI-2 products.
Table 2.1-2 shows correspondence table of between product and mode and band.

## Table 2.1-1 Definition of TANSO-CAI-2 Product

| Type | Definition | Operation Mode | Appended information |
| :---: | :---: | :---: | :---: |
| Level 1A | Level 1A products contain uncorrected image data of TANSO-CAI-2, which is stored as digital number, together with geometric information at observation point and telemetry of temperature, etc. <br> Every scene, the following 4 files are produced. <br> - Common file <br> Common information for both Forward looking and Backward looking is stored. <br> - Forward looking band file <br> Information for Forward looking is stored <br> (The observation data of band $1-5$, etc.). <br> Backward looking band file <br> Information for Backward looking is stored <br> (The observation data of band $6-10$, etc.). <br> TANSO-CAI-2 L1 processing result file <br> Quality information, geometric information of representative point and etc. are stored as XML format. | Observation <br> Mode | - Point number <br> - Line exposure time <br> - Gain, various sensor temperatures and exposure duration <br> - Latitude and longitude at representative point <br> - Satellite orbit data at representative point (ECI, ECR) <br> - Satellite attitude data at representative point <br> - Sensor zenith and azimuth angles at representative point <br> - Sun position at representative point (ECI, ECR) <br> - Moon position at representative point (ECI, ECR) <br> - Quality information |
| Level 1A <br> Calibration | Same as Level 1A. | Electrical <br> Calibration <br> Dark Calibration <br> Lunar <br> Calibration | Same as Level 1A. <br> In this mode, sensor doesn't observe on the earth surface, therefore information of geolocation, etc. at representative point is not calculated. |

Table 2.1-2 Correspondence table of product and mode/band

|  |  | TANSO-CAI-2 <br> Observation or Calibration <br> mode | Stored Band |
| :--- | :--- | :--- | :--- |
| TANSO-CAI-2 <br> L1A Product | Observation <br> (day) | Observation Mode | Forward looking (Band1-5) |
|  |  | Backward looking (Band6-10) |  |
| TANSO-CAI-2 <br> L1A <br> Calibration <br> Product | Dark <br> Calibration | Electrical <br> Calibration | Electrical Calibration mode |

### 2.2. Unit of product

Unit of TANSO-CAI-2 level 1 product is described as follows:
(1) One scene product is defined as 1 satellite revolution data starting from ascending node to the next ascending node. If the observation points (satellite position) cross ascending node, the product should be divided into separate products.
(2) Common file, Forward looking band file (for band 1-5) and Backward looking band file (for band 6-10) are produced for both of Observation mode and Calibration mode. Calibration product has plural calibration mode in a product (Night, Electrical, Lunar calibration mode). These calibration data are united together for every calibration mode and stored.

### 2.3. Data contents

Basic observation modes of TANSO-CAI-2 are shown in Table 2.3-1.
In nominal operation phase, TANSO-CAI-2 sensor of Forward looking observes with band 1-5 and Backward looking observes with band 6-10 during the daytime on earth surface.
Lunar calibration uses the reflected solar irradiance from moon. In this calibration, TANSO-CAI-2 is oriented the moon during night and CAI-2's FOV catches the reflected solar irradiance after attitude maneuver.

Table 2.3-1 Basic observation modes of TANSO-CAI-2

| Observation Mode |  | Description |
| :--- | :--- | :--- |
| Observation Mode 1 | Nominal Observation |  |
| Observation Mode 2 | In the situation that the power supply level of the satellite becomes lower <br> and the satellite cannot keep observation mode1, the observation <br> continues under the condition that a part of function of TANSO-CAI-2 is <br> suspended depending on the power level. |  |
| Calibration <br> Mode | Lunar <br> Calibration | Once a month <br> Same as Observation Mode 1 and 2. |
|  | Electric <br> Calibration | Every path <br> Performs calibration of signal processing after the analogue-signal <br> processing system, by inputting a reference voltage signal. |
|  | Dark <br> Calibration | Once a month as needed. <br> Same as Observation Mode 1 and 2. <br> Calibrates the offset level during nighttime. |

Data contents for each processing level and mode are shown in Table 2.3-2.

Table2.3-2 Contents of TANSO-CAI-2 L1 Products

| Processing Level | Observation Mode | Used Band | Data Contents (for one observation point) | Data size | Note |
| :---: | :---: | :---: | :---: | :---: | :---: |
| 1A | Observation Mode | Forward looking band(Band 1-5) <br> Backward looking band (Band 6-10) | Band 1-4, Band 6-9: <br> 2056 pixels. <br> $1^{\text {st- }}$ 8th $^{\text {th }}$ pixel : dark <br> Band 5, Band 10: <br> 1024 pixels <br> $1^{\text {st- }} 6^{\text {th }}$ pixel: dark <br> $7-66^{\text {th }}$ pixel: invalid | The nominal data size of daytime satellite in a revolution is about 40,000 lines <br> (In case of Band 5,10 are about a half of above, 20,000 lines) | - |
| $1 \mathrm{~A}$ <br> Calibration | Calibration Mode <br> (Electric Calibration) | ibid | ibid | A series of Electric and Dark calibration data | Input reference voltage signal |
|  | Calibration Mode <br> (Dark Calibration) |  |  |  | Observe earth in night time for acquiring the dark offset level. Acquire the offset level by observing during night time. |
|  | Calibration Mode <br> (Lunar Calibration) | ibid | ibid | A series of Lunar calibration data | Observe Lunar |

## 3. Product format

### 3.1. File name convention

3.1.1. File name convention of TANSO-CAI-2 L1A (HDF5 format)

Table $3.1^{-1}$ shows the file name convention of TANSO-CAI-2 L1A products.
Table3.1-1 File name convention of TANSO-CAI-2 L1A products
 GOSAT 2TCAI 2/YYYYMMDDHHmmPPPISS. LLBARCOOOOOOAAABBB. h 5

Convention for each item is shown below

- Satellite Name : GOSAT2 (Fixed)
- Sensor Name : TANSO-CAI-2: TCAI2 (Fixed)
- Observation time at the first line of scene (year • month • day • hour • minute) : YYYYMMDDHHmm (UT)
Compared with the first line of time of Forward and Backward band, the former line time is nominally stored. Because in nominal case, the start of Forward looking is ahead of that of Backward looking.
- Path No. : PPP(001-089)
- Scene No. : 00(Fixed)
- Processing Level : 1A(Fixed)
- Band : B

Common file : C
Forward looking band (Band1-5) : F
Backward looking band (Band6-10) : B

- Orbit data used for processing : R

Using predicted orbit data : P
Using GPS or determined orbit data: D

- Correction coefficients used for processing : C

Using nominal coefficients: N
Using updated coefficients: U

- Reserved : 00
- Operation Mode : OOOO

Observation Mode (day) : OBSM
Dark calibration mode : NCAL
Electric calibration mode: ECAL
Lunar calibration mode : LCAL

- Algorithm Version : AAA (000-999)
- Parameter Version : BBB (000-999)
- Extension : h5 (Fixed)


### 3.1.2. File name convention of processing result (XML format)

Table 3.1-2 shows File name convention of TANSO-CAI-2 L1 processing result file.

Table3.1-2 File name convention of TANSO-CAI-2 L1 processing result file

| 1 | 2 | 3 | 4 | 5 | 6 | 7 | 8 | 9 | 10 | 11 | 12 | 13 | 14 | 15 | 16 | 17 | 18 | 19 | 20 | 21 | 22 | 23 | 24 | 25 | 26 | 27 | 28 | 29 | 30 | 31 | 32 | 33 | 34 | 35 | 36 | 37 | 38 | 39 | 40 | 41 | 42 | 43 | 44 | 45 | 46 |
| :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | :--- | $\mathbf{4 7}$| 48 |
| :--- |



The convention for each item is shown below.

- Satellite Name : GOSAT2 (Fixed)
- Sensor Name : TANSO-CAI-2: TCAI2 (Fixed)
- Start Time of Observation (year • month • day • hour • minute) : YYYYMMDDHHmm (UT)
- Path No. : PPP (001-089)
- Scene No. : 00 (Fixed)
- Processing Level : 1A (Fixed)
- Band : B

Common file : C
Forward looking band (Band1-5) : F
Backward looking band (Band6-10) : B

- Orbit data used for processing : R

Using predicted orbit data : P
Using GPS or determined orbit data: D

- Correction coefficients used for processing : C

Using nominal coefficients : N
Using updated coefficients : U

- Reserved : 00
- Operation Mode : OOOO

Observation Mode (day) : 0BSM
Dark calibration mode : NCAL
Electric calibration mode : ECAL
Lunar calibration mode : LCAL

- Algorithm Version : AAA (000-999)
- Parameter Version : BBB (000-999)
- Extension : XML (Fixed)


### 3.2. Dataset Structure

TANSO-CAI-2 L1 product consists of Common file (consists of common information to Forward looking and Backward looking), Forward looking band file (for band 1-5) and Backward looking band file (for band 6-10).
Common file contains metadata, orbit and attitude data, ephemeris data (sun and moon), housekeeping telemetry data such as temperature, status of instruments.
Forward/Backward looking band file contain metadata, Scene Attribute (information of number of pixels, etc.), Line Attribute (line observation time, etc.), Geometric information, Image (digital number for each band), these are Forward/Backward specific information.
Dataset structure of TANSO-CAI-2 L1 product is shown in Table 3.2-1.
Table 3.2-1 Dataset Structure of TANSO-CAI-2 Level 1 Product (1/2)

| File | Group | Outline |
| :---: | :---: | :---: |
| Common file | Metadata | Items below are stored as explanation of product type, contents, etc. <br> - Granule ID <br> - Operation Mode <br> - Date of product creation <br> - Processing Level <br> - Processing Algorithm/Parameter version <br> - Start and end time of observation <br> - Quality information |
|  | SpacecraftTimeError | Parameter for correcting spacecraft time error is stored. |
|  | SiderealTimeInfo | Parameter for calculating Greenwich sidereal time is stored. |
|  | TransMatrixInfo | Transform matrix which convert from J2000.0 to TOD and true ECR corresponding polar motion is stored. |
|  | OnboardOrbitData | Onboard orbit data is stored. |
|  | KinematicOrbitDataPredicted | Predicted kinematic orbit data is stored. |
|  | KinematicOrbitDataDetermined | Determined kinematic orbit data is stored. |
|  | AttitudeData | Onboard attitude data is stored. |
|  | SolarEphemeris | Solar position and velocity data is stored. |
|  | LunarEphemeris | Lunar position and velocity data is stored. |
|  | TemperatureTelemetry_1sec | Temperature telemetry of 1 sec period is stored. |
|  | TemperatureTelemetry_32sec | Temperature telemetry of 32 sec period is stored. |
|  | HK_Telemetry_1sec | Housekeeping telemetry of 1sec period is stored. |
| Forward / <br> Backward looking band file | Metadata | Items below are stored as explanation of product type, contents, etc. <br> - Granule ID <br> - Operation Mode <br> - Date of product creation <br> - Processing Level <br> - Processing Algorithm Name and its version <br> - Start and end time of observation |
|  | SceneAttribute | Number of bands, pixels and lines are stored as information about observation point data. |

Table 3.2-1 Dataset Structure of TANSO-CAI-2 Level 1 Product (2/2)

| File | Group | Outline |
| :---: | :---: | :---: |
|  | LineAttribute_500 | Observed time, missing flag, etc. for each line are stored as information of band 1-4(for Forward looking band file) or 6-9(for Backward looking band file). |
|  | LineAttribute_1km | Observed time, missing flag, etc. for each line are stored as information of band 5(for Forward looking band file) or 10(for Backward looking band file). |
|  | ImageData | Image data is stored. |
|  | GeometryAttribute | Standard band No, sampling interval of both pixel and line direction, number of the sample of both pixel and line direction, etc. are stored as geometric information for reference band |
|  | ImageGeometry | Latitude, longitude, sensor zenith and azimuth angle, solar zenith and azimuth angle, etc. for each sample are stored as geometric information for reference band. |
|  | SatelliteGeometry | Satellite position, velocity and attitude for each line are stored as geometric information for reference band. |
|  | SolarGeometry | Solar position and velocity for each line are stored as geometric information for reference band. |
|  | LunarGeometry | Lunar position and velocity for each line are stored as geometric information for reference band. |

### 3.3. Notes for definition of data group

(1) Definition of data type

Table 3.3-1 describes definition of data type stored in TANSO-CAI-2 L1A products.
Table 3.3-1 Definition of data type

| HDF5 type | Data type |
| :---: | :---: |
| H5T_STRING | more than 1 byte string |
| H5T_STD_I8LE | signed 1byte integer |
| H5T_STD_U8LE | unsigned 1byte integer |
| H5T_STD_I16LE | signed 2byte integer |
| H5T_STD_U16LE | unsigned 2byte integer |
| H5T_SSTD_I32LE | signed 4byte integer |
| H5T_STD_U32LE | unsigned 4byte integer |
| H5T_IEEE_F32LE | signed 4byte float |
| H5T_IEEE_F64LE | signed 8byte double |

(2) Expression of time

UTC date is expressed as "YYYY-MM-DDThh:mm:ss.ffffffZ" with string data. "YYYY-MM-DD" means year, month and day. "hh:mm" means hour and minute. "ss.ffffffZ" means second with microsecond accuracy.

Spacecraft time is defined as follows:
Spacecraft Time (s) = GPS Time (s) - 1,041,033,615(s),
where GPS Time (s) is total seconds since 00:00:00 UTC, Jan 6, 1980.

## (3) Definition of coordinates

Table 3.3-2 describes definition of coordinates used for dataset.

Table 3.3-2 Definition of coordinates

| Name | Abbreviated name | The origin/Axis | Definition |  |
| :---: | :---: | :---: | :---: | :---: |
| Inertial coordinate system (J2000.0 coordinate ) | $\Phi_{1}$ | The origin: $\mathrm{O}_{\text {I }}$ | Earth centered | $\begin{aligned} & \text { EPOCH } \\ & \text { 2000/01/01 } \\ & \text { 12:00:00 TT(Earth time) } \end{aligned}$ |
|  |  | $\mathrm{X}_{\mathrm{I}}$ | Direction of mean vernal equinox of EPOCH |  |
|  |  | $Y_{1}$ | $\mathrm{Z}_{\mathrm{I}} \times \mathrm{X}_{\mathrm{I}}$ |  |
|  |  | Z | Vertical direction of mean equatorial plain of EPOCH (Direction of the north pole is + ) |  |
| Coordinate Reference Systems in Orbit | $\Phi_{R}$ | The origin: $\mathrm{O}_{\mathrm{R}}$ | Ascending node |  |
|  |  | $\mathrm{X}_{\mathrm{R}}$ | Coincide with ascending node of orbit coordinate |  |
|  |  | $Y_{R}$ |  |  |  |
|  |  | $\mathrm{Z}_{\mathrm{R}}$ |  |  |  |
| Orbit coordinate | $\Phi_{0}$ | The origin:Oo | Center of the mass of satellite | Defined by orbit model of AOCE inInertial coordinate system. |
|  |  | $\mathrm{X}_{0}$ | $Y_{0} \times Z_{0}$ |  |
|  |  | $Y_{0}$ | Opposite direction of vector of orbit plane |  |
|  |  | $\mathrm{Z}_{0}$ | Direction of center of the earth |  |
| Coordinate Reference <br> Systems in STT <br> (Reference point for determination of satellite attitude ) | $\Phi_{\text {STT1 }}$ | The origin: $\mathrm{O}_{\text {STT1 }}$ | Reference mirror in STT | Defined after early operations phase on orbit |
|  |  | $\mathrm{X}_{\text {STT1 }}$ | Roll axis in orbit |  |
|  |  | $Y_{\text {STT1 }}$ | Pitch axis in orbit |  |
|  |  | $\mathrm{Z}_{\text {STT1 }}$ | Yaw axis in orbit |  |
| Satellite coordinate | $\Phi_{\text {B }}$ | The origin: $\mathrm{O}_{\mathrm{B}}$ | Center of the mass of satellite | Coincide with the orbit coordinate except for attitude error |
|  |  | $\mathrm{X}_{\mathrm{B}}$ | Parallel to each axis of coordinate reference systems in STT |  |
|  |  | $Y_{B}$ |  |  |
|  |  | $\mathrm{Z}_{\mathrm{B}}$ |  |  |
| Coordinate Reference Systems in TANSO-CAI-2 | $\Phi_{\text {CAI-2 }}$ | The origin: $\mathrm{O}_{\mathrm{CAI}-2}$ | Transform matrix to convert Satellite coordinate $\Phi_{B}$ from Coordinate Reference Systems in TANSO-CAI-2 is provided as separate file. Ideally, this is stored in unit matrix. <br> The origin is same as satellite coordinate. |  |
|  |  | $\mathrm{X}_{\text {CAI-2 }}$ |  |  |  |
|  |  | $\mathrm{Y}_{\text {CAI-2 }}$ |  |  |  |
|  |  | $\mathrm{Z}_{\text {CAI-2 }}$ |  |  |  |
| Satellite-fixed coordinate | $\Phi_{\text {S }}$ | The origin: $\mathrm{O}_{\text {S }}$ | Intersection point of center line of and satellite separation plain |  |
|  |  | $\mathrm{X}_{\mathrm{S}}$ | Roll axis in machine |  |
|  |  | $Y_{S}$ | Pitch axis in machine |  |
|  |  | $\mathrm{Z}_{\text {S }}$ | Yaw axis in machine |  |
| Earth-fixed coordinate | $\Phi_{\text {WGS84 }}$ | The origin: $\mathrm{O}_{\text {WGS84 }}$ | the gravity center of the earth | GPSR gives absolute position and absolute velocity on this corrdinate |
|  |  | $X_{\text {WGS84 }}$ | Coincide with X axis which is defined byBIH for calculation of earth rotation paramerter |  |
|  |  | $\mathrm{Y}_{\text {WGS84 }}$ | $Z_{\text {WGS84 }} \times \mathrm{X}_{\text {WGS84 }}$ |  |
|  |  | $\mathrm{Z}_{\text {WGS84 }}$ | Parallel to $Z$ axis which is defined byBIH for calculation of earth rotation paramerter. <br> $Z$ axis is the direction of CTP. |  |
| TOD coordinate | $\Phi_{\text {TOD }}$ | The origin: $\mathrm{O}_{\text {TOD }}$ | Earth centered |  |
|  |  | $\mathrm{X}_{\text {TOD }}$ | Direction of vernal equinox at the present time | Inertial coordinate system (J2000.0 coordinate) ФI with taking into precession and nutation. |
|  |  | $Y_{\text {TOD }}$ | $\mathrm{Z}_{\text {TOD }} \times \mathrm{X}_{\text {TOD }}$ |  |
|  |  | $\mathrm{Z}_{\text {TOD }}$ | Vertical direction of equatorial plain at the present time (Direction of the north pole is + ) |  |

(4) Definition of latitude/longitude

Unless otherwise specifically noted, latitude and longitude in this document means geographic latitude and longitude.

### 3.4. Definition of common file

### 3.4.1. Metadata group

Each dataset in Metadata describes product type, contents, etc which are related to this product file.

Metadata group in common file contains productQualityFlag. productQualityFlag refers to number of missing lines and evaluates quality of product in four levels (Good, Fair, Poor, NG ). The criteria are shown below.
When productQualityFlag is Good, there is no missing lines.
Fair/Poor/NG is determined by the threshold value defined in this system. When productQualityFlag is evaluated "NG", product isn't provided to users.

### 3.4.2. SpacecraftTimeError group

SpacecraftTimeError group contains the information to correct the gap between satellite and the ground station time. If the status of time system is normal, this correction is not required to use.
Formula to correct the time gap is as follows:
Spacecraft time (after correction)
$=$ periodCount * $\{$ spacecraft time(before correction) - refCount $\}+$ groundTime

### 3.4.3. SiderealTimeInfo group

SiderealTimeInfo group contains information of Greenwich sidereal time. Using this information, TOD can be converted to pseudo earth-fixed coordinates (Polar motion is not considered.).

Using Greenwich sidereal time $\theta_{g o}$ at the baseline time $t_{0}$ and Deviation of Greenwich sidereal time $d \theta_{g} / d t$ Greenwich sidereal time $\theta_{g}$ at the arbitrary time $t$ is expressed as follows:

$$
\theta_{g}=\theta_{g 0}+d \theta_{g} / d t \times\left(t-t_{0}\right)
$$

Transform matrix conversion from TOD to pseudo earth-fixed coordinate $\mathbf{M}_{\text {Tod-PECR }}$ is as follows:

$$
\mathbf{M}_{\text {TOD-PECR }}=\left(\begin{array}{ccc}
\cos \theta_{g} & \sin \theta_{g} & 0 \\
-\sin \theta_{g} & \cos \theta_{g} & 0 \\
0 & 0 & 1
\end{array}\right)
$$

### 3.4.4. Definition of TransMatrixInfo data group

TransMatrixInfo data group contains PN matrix which can convert from J2000.0 coordinates to TOD coordinates and XY matrix which can convert from pseudo earth-fixed coordinates to ECR coordinates.
Data interval is 60 sec . But in case of the leap second, data interval is 61 sec .

### 3.4.5. OnboardOrbitData group

OnboardOrbitData group contains onboard orbit data (expressed in ECR coordinates) and orbit data converted to TOD coordinates from onboard orbit data.
Data interval is 1 sec . But in case of data missing, there can be some gap in data interval.
The Conversion method from position vector $\mathbf{P}_{\text {ECR }}$ and velocity vector $\mathbf{V}_{\text {ECR }}$ of onboard in ECR to position vector $\mathbf{P}_{\text {TOD }}$ and velocity vector $\mathbf{V}_{\text {ToD }}$ in TOD is described below.
First, $\mathbf{P}_{\text {ECR }}$ and $\mathbf{V}_{\text {ECR }}$ convert to position vector and velocity vector in pseudo earth-fixed coordinates (This coordinate doesn't consider polar motion) using XY matrix:

$$
\begin{align*}
& \mathbf{P}_{\mathbf{P E C R}}=\mathbf{X} \mathbf{Y}^{t} \times \mathbf{P}_{\mathrm{ECR}} \\
& \mathbf{V}_{\mathbf{P E C R}}=\mathbf{X} \mathbf{Y}^{t} \times \mathbf{V}_{\mathrm{ECR}}
\end{align*}
$$

(The superscript $t$ denotes transpose. Since XY matrix is unitary, transpose of it is the same as its inverse matrix.)
Next, $\mathbf{P}_{\text {Pecr }}$ and $\mathbf{V}_{\text {Pecr }}$ convert to $\mathbf{P}_{\text {tod }}$ and $\mathbf{V}_{\text {tod }}$ using Greenwich sidereal time $\boldsymbol{\theta}_{g}$ and Deviation of Greenwich sidereal time $d \theta_{g} / d t$

$$
\begin{align*}
& \mathbf{P}_{\text {TOD }}=\left(\begin{array}{ccc}
\cos \left(-\theta_{g}\right) & \sin \left(-\theta_{g}\right) & 0 \\
-\sin \left(-\theta_{g}\right) & \cos \left(-\theta_{g}\right) & 0 \\
0 & 0 & 1
\end{array}\right) \times \mathbf{P}_{\mathbf{P E C R}} \\
& \mathbf{V}_{\text {TOD }}=\left(\begin{array}{ccc}
\cos \left(-\theta_{g}\right) & \sin \left(-\theta_{g}\right) & 0 \\
-\sin \left(-\theta_{g}\right) & \cos \left(-\theta_{g}\right) & 0 \\
0 & 0 & 1
\end{array}\right) \times\left[\mathbf{V}_{\mathbf{P E C R}}+\left(\begin{array}{c}
0 \\
0 \\
\dot{\theta}_{g}
\end{array}\right) \otimes \mathbf{P}_{\mathrm{PECR}}\right]
\end{align*}
$$

Eq. 3.4.5-4
where $\otimes$ denotes outer product.

### 3.4.6. KinematicOrbitDataPredicted group

KinematicOrbitDataPredicted group contains predicted kinematic orbit data in ECR and TOD coordinates, this data is distributed from the kinematic orbit system.
In all cases (includes the leap second is inserted), data interval is 60 sec .

### 3.4.7. KinematicOrbitDataDetermined group

KinematicOrbitDataDetermined group contains determined kinematic orbit data in ECR and TOD coordinates, this data is distributed from the kinematic orbit system.
In all cases (includes the leap second is inserted), data interval is 60 sec .

### 3.4.8. AttitudeData group

AttitudeData group contains onboard attitude data and yaw steering flag which shows
yaw steering operation status.
The data interval is not constant. But, in case of data missing, there can be some gap in data interval.

Attitude data is given in quaternion $Q=\left(q_{0}, q_{1}, q_{2}, q_{3}\right)$ in $J 2000.0 . q_{0}$ is scalar component and ( $q_{1}, q_{2}, q_{3}$ ) are vector components.
Interpolation is needed to determine attitude data at the given time.

Transform matrix $\mathbf{M}_{\text {I2000-body }}$ which convert satellite coordinates from J2000.0 coordinates is expressed as follows:

$$
\mathbf{M}_{\text {J2000-body }}=\left(\begin{array}{ccc}
q_{0}^{2}+q_{1}^{2}-q_{2}^{2}-q_{3}^{2} & 2\left(q_{1} q_{2}+q_{0} q_{3}\right) & 2\left(q_{1} q_{3}-q_{0} q_{2}\right) \\
2\left(q_{1} q_{2}-q_{0} q_{3}\right) & q_{0}^{2}-q_{1}^{2}+q_{2}^{2}-q_{3}^{2} & 2\left(q_{2} q_{3}+q_{0} q_{1}\right) \\
2\left(q_{1} q_{3}+q_{0} q_{2}\right) & 2\left(q_{2} q_{3}-q_{0} q_{1}\right) & q_{0}^{2}-q_{1}^{2}-q_{2}^{2}+q_{3}^{2}
\end{array}\right)
$$

Transform matrix $\mathbf{M}_{\text {body-J2000 }}$ which converts J2000.0 coordinates from satellite coordinates is transpose matrix of $\mathbf{M}_{\mathbf{J 2 0 0 0} \text {-body. }} \mathbf{M}_{\text {body-J2000 }}$ is expressed as follows (The superscript $t$ denotes transpose):
$\mathbf{M}_{\text {body-J2000 }}=\left(\mathbf{M}_{\text {J2000-body }}\right)^{t}=\left(\begin{array}{ccc}q_{0}^{2}+q_{1}^{2}-q_{2}^{2}-q_{3}^{2} & 2\left(q_{1} q_{2}-q_{0} q_{3}\right) & 2\left(q_{1} q_{3}+q_{0} q_{2}\right) \\ 2\left(q_{1} q_{2}+q_{0} q_{3}\right) & q_{0}^{2}-q_{1}^{2}+q_{2}^{2}-q_{3}^{2} & 2\left(q_{2} q_{3}-q_{0} q_{1}\right) \\ 2\left(q_{1} q_{3}-q_{0} q_{2}\right) & 2\left(q_{2} q_{3}+q_{0} q_{1}\right) & q_{0}^{2}-q_{1}^{2}-q_{2}^{2}+q_{3}^{2}\end{array}\right)$
Eq. 3.4.8-2

### 3.4.9. SolarEphemeris group

SolarEphemeris group contains the kinematic solar position and velocity data in ECR and TOD coordinates distributed from kinematic orbit system. In all cases (includes the leap second is inserted), data interval is always 60 sec .
Solar position and velocity data are true position and velocity at the time. The light propagation time from sun to earth is not taken in account. However, time data has been recorded since about 10 minutes before start of observation. Thus, solar position and velocity data with taking account of light propagation time can be calculated.

### 3.4.10. LunarEphemeris group

LunarEphemeris group contains kinematic lunar position and velocity data in ECR and TOD coordinates distributed from kinematic orbit system. In all cases (includes the leap second is inserted), data interval is always 60 sec .

### 3.4.11. TemperatureTelemetry_1sec group

TemperatureTelemetry_1sec group contains temperature telemetry.
Data interval is 1 sec but there can be some gap in data interval in case of data missing.
The evaluation result about the range of temperature is stored for each data.
Data of sensorTemp, preAmpTemp and ampTemp in this group are used for radiometric conversion. (For the radiometric conversion, see Chapter 4.2)
Temperature data except the above are used for checking sensor condition.

### 3.4.12. TemperatureTelemetry_32sec group

TemperatureTelemetry_32sec group contains temperature telemetry.

Data interval is 32 sec but there can be some gap in data interval in case of data missing. The evaluation result about the range of temperature is stored for each data.
The temperature data are used for checking sensor condition.

### 3.4.13. HK_Telemetry_1sec group

HK_Telemetry_1sec group contains housekeeping telemetry.
Data interval is 1 sec but there can be some gap in data interval in case of data missing. The telemetry data are used for checking sensor condition.

### 3.5. Definition of Forward/Backward looking band file

### 3.5.1. Metadata group

Each dataset in Metadata describes product type, contents, etc are related to this product file.

### 3.5.2. SceneAttribute group

SceneAttribute group contains number of bands, pixels and lines for each resolution in this product are stored, which are related to this product file.

CAI-2 has 2 resolutions, one is 500 m and 1000 m is the other.
In Forward looking band file, 500 m spatial resolution bands are Band 1-4, total 4 bands and 1 km spatial resolution band is only Band 5 .

In Backward looking band file, 500 m spatial resolution bands are Band 6-9,total 4 bands, and 1 km spatial resolution band is only Band 10 .

### 3.5.3. LineAttribute_500 group

LineAttribute_500 group contains the following information for 500 m spatial resolution bands.

- missingFlag
- observationTime
- satTime
- satTimeStatusFlag
- fineobservationCounter
- integrationNum
- exposureTime
observationTime is the center of exposureTime considering the duration of exposure. In case satellite time system status is anomaly, observationTime is corrected by time correction information.
satTime and exposureTime are used for calculation of observationTime. observationTime is expressed as follows:

```
observationTime
    \(=\) satTime \(+(\) Fixed-delay time \()+\) exposureTime \(\times 0.5\)
```

(Fixed-delay time) is a fixed parameter and it is not stored in product.

### 3.5.4. LineAttribute_1km group

LineAttribute_1km group contains observationTime, etc. for 1 km spatial resolution bands. The contents are the same as LineAttribute_500 group.

### 3.5.5. ImageData group

ImageData group contains the digital number of CAI-2 data (Effective digit are 12 bits). Each band includes dark pixels, invalid pixels and valid pixel. The dark pixels are the pixel to store data during dark. The valid pixels are the pixel to store during observing earth surface data. The invalid pixels are not used in processing.

Table 3.5.5-1 shows the pixel number of dark, invalid and valid pixels for each band.

Table 3.5.5-1 Pixel number of Dark, invalid and valid pixels

| Band | Number of <br> pixels | Dark pixel No. | Invalid pixel <br> No. | Valid pixel No. |
| :---: | :---: | :---: | :---: | :---: |
| Band $1 \sim 4$ <br> Band $6 \sim 9$ | 2056 | $1 \sim 8$ | - | $9 \sim 2056$ |
| Band 5,10 | 1024 | $1 \sim 6$ | $7 \sim 66$ | $67 \sim 1024$ |

### 3.5.6. GeometryAttribute group

GeometryAttribute group contains information about pixel which has geometry information. The pixels are sampled from reference band every 10 pixels and 10 lines as Figure $3.5 .6-1$ shows. The reference band is any one of band 1 to 4 in the case of forward looking and any one of band 6 to 9 in the case of backward looking and stdBand shows the band number.

In pixel direction, the pixel are sampled every 10 pixels start from the pixel number 9 to pixel number 2056. The last pixel is 2056 .

In line direction, the lines are sampled from the beginning of the lines and every 10 lines. The last line will be a sampled line. Only information of the pixel which has valid orbit data will be stored in the group.


Figure 3.5.6-1 The sample pixels

### 3.5.7. ImageGeometry group

ImageGeometry group contains latitude and longitude on the standard band image, sensor zenith and azimuth angle, solar zenith and azimuth angle, solar distance from the observation point and lunar to satellite to solar angle.

The calculation of the values in this group related solar position (for example, solar zenith angle, etc.) is used by the apparent position in consideration of light propagation time from sun to earth (fixed value).

The Definition of sensor and solar zenith/azimuth angle and the angle between lunar to satellite vector and solar- to satellite vector in this product are described below.
(1) The Definition of sensor and solar zenith/azimuth angle

When geographic latitude/longitude is defined as $\lambda / \varphi$ in observation point $\mathbf{p}_{\text {obs }}=\left(p_{\text {obs_ }}\right.$, $\left.p_{\text {obs_y }}, p_{\text {obs_z }}\right)^{t}$, unit vector of zenith direction $\mathbf{z}$, unit vector of north direction $\mathbf{n}$ and unit vector of east direction $\mathbf{e}$ are expressed as follows:

$$
\begin{align*}
& \mathbf{z}=\left(\begin{array}{c}
\cos \varphi \cos \lambda \\
\cos \varphi \sin \lambda \\
\sin \varphi
\end{array}\right) \\
& \mathbf{n}=\left(\begin{array}{c}
-\sin \varphi \cos \lambda \\
-\sin \varphi \sin \lambda \\
\cos \varphi
\end{array}\right) \\
& \mathbf{e}=\left(\begin{array}{c}
-\sin \lambda \\
\cos \lambda \\
0
\end{array}\right)
\end{align*}
$$

Eq. 3.5.7-3

Using sensor or solar position vector in ECR $\mathbf{p}_{\text {ECR }}$, zenith angle $\theta_{z}$ and azimuth angle $\varphi_{A z}$ are expressed as follows:

$$
\begin{align*}
& \theta_{z}=\operatorname{acos}\left(\frac{\left(\mathbf{p}_{\mathrm{ECR}}-\mathbf{p}_{\mathbf{o b s}}\right) \cdot \mathbf{z}}{\left|\mathbf{p}_{\mathrm{ECR}}-\mathbf{p}_{\mathrm{obs}}\right|}\right) \\
& \varphi_{\mathrm{Az}}=\operatorname{atan} 2\left(\left(\mathbf{p}_{\mathrm{ECR}}-\mathbf{p}_{\mathrm{obs}}\right) \cdot \mathbf{e}, \quad\left(\mathbf{p}_{\mathrm{ECR}}-\mathbf{p}_{\mathrm{obs}}\right) \cdot \mathbf{n}\right)
\end{align*}
$$

$\left(\mathbf{p}_{\text {ECR }}-\mathbf{p}_{\text {obs }}\right)$ is the direction to sun from observation point or satellite.

As the azimuth angle $\varphi_{A z}$ is defined from 0 to $2 \pi$ [rad] ( 0 to 360 [deg]). If $\varphi_{A z}$ is negative value at Eq.3.5.7-5 add $2 \pi$ to $\varphi_{A z}$. The definition of atan 2 function, please refer to Chapter 5.


Figure 3.5.7-1 The definition of sensor and solar zenith/azimuth angle.
(2) The Definition of angle between lunar to satellite vector and solar to satellite vector Using sensor position vector $\mathbf{p}_{\text {sat }}$, solar position vector $\mathbf{p}_{\text {sud }}$ and lunar position vector $\mathrm{p}_{\text {moon }}$, the angle between lunar to satellite vector and solar to satellite vector $\theta_{e l}$ is expressed as follows:

$$
\theta_{e l}=\operatorname{acos}\left(\frac{\left(\mathbf{p}_{\text {MOON }}-\mathbf{p}_{\text {sat }}\right) \cdot\left(\mathbf{p}_{\text {sun }}-\mathbf{p}_{\text {sat }}\right)}{\left|\mathbf{p}_{\text {MOON }}-\mathbf{p}_{\text {sat }} \| \mathbf{p}_{\text {SUN }}-\mathbf{p}_{\text {sat }}\right|}\right)
$$



Figure 3.5.7-2 The angle between lunar-satellite vector and solar-satellite vector.

## (3) The definition of scatter angle

The scatter angle is defined as the angle between the progress direction of scattered light and the direction of incident light. When the scatted light progresses to the same direction of incident light ( $\varphi_{\text {SCAT }}=0$ degree) , the scatter is called the forward scatter and when it progresses to the opposite direction ( $\varphi_{\text {SCAT }}=180$ degree) , it is called the backward scatter. The definition is as follows.

$$
\varphi_{\text {SCAT }}=\operatorname{acos}\left(\Phi_{\text {SCAT }}\right)
$$

$\Phi_{\text {SCAT }}$ is defined as follows and others like $\theta_{Z_{-S a r}}$ refers to section(1) and Figure3.5.8-1

$$
\begin{aligned}
\Phi_{\text {SCAT }}=- & \sin \theta_{z_{-} S U N} \sin \varphi_{A z_{-} S U N} \sin \theta_{z_{-} \text {sat }} \sin \varphi_{A z_{-} \text {sat }} \\
& -\sin \theta_{z_{-} S U N} \cos \varphi_{A z_{-} S U N} \sin \theta_{z_{-} \text {sat }} \cos \varphi_{A z_{-} \text {sat }} \\
& -\cos \theta_{z_{-} S U N} \cos \theta_{z_{-} \text {sat }}
\end{aligned}
$$

Eq.3.5.7-8


Figure 3.5.7-3 The definition of scattered angle

### 3.5.8. SatelliteGeometry group

SatelliteGeometry group contains satellite position/velocity (in ECR and TOD) and attitude in the sample lines of standardBand and transformation matrix (satToECR_Matrix) which can convert to ECR (WGS84) coordinate from satellite coordinate.

Satellite attitude is stored as quaternion in J2000.0 and roll, pitch and yaw angles. Definition and usage of quaternion are the same as Chapter 3.4.8. Roll, pitch, and yaw angles are calculated by using quaternion, etc. The algorithm is shown later.

Transformation matrix (satToECR_Matrix) can transform coordinate from satellite coordinate to ECR (WGS84) directly. The matrix includes all coordinate transformation from satellite coordinate to J2000, J2000 to TOD, and TOD to ECR(WGS84). For usage of this matrix, see Chapter 5.

The calculation of roll, pitch and yaw angles are described below.
The first step is to make a transform matrix from orbit coordinate to TOD by using $\mathbf{p}_{\text {Tod }}$ satellite position and velocity $\mathbf{v}_{\text {Tod }}$ vectors in TOD .

$$
\mathbf{E}_{\text {orbit-TOD }}=\left(\begin{array}{lll}
E_{11} & E_{12} & E_{13} \\
E_{21} & E_{22} & E_{23} \\
E_{31} & E_{32} & E_{33}
\end{array}\right)
$$

Each element of the matrix is defined as follows:

$$
\begin{align*}
& \mathbf{E}_{\mathbf{z}}=\left(\begin{array}{l}
E_{13} \\
E_{23} \\
E_{33}
\end{array}\right)=-\frac{\mathbf{p}_{\text {TOD }}}{\left|\mathbf{p}_{\text {TOD }}\right|} \\
& \mathbf{E}_{\mathrm{y}}=\left(\begin{array}{l}
E_{12} \\
E_{22} \\
E_{32}
\end{array}\right)=-\frac{\mathbf{p}_{\text {TOD }} \otimes \mathbf{v}_{\text {TOD }}}{\left|\mathbf{p}_{\text {Tod }} \otimes \mathbf{v}_{\text {ToD }}\right|} \\
& \mathbf{E}_{\mathrm{x}}=\left(\begin{array}{l}
E_{11} \\
E_{21} \\
E_{31}
\end{array}\right)=\mathbf{E}_{\mathrm{y}} \otimes \mathbf{E}_{\mathrm{z}}
\end{align*}
$$

Eq. 3.5.8-4
where $\otimes$ denotes outer product.
Next step is to make a transform matrix from orbit coordinate to satellite body $\mathbf{M}_{\text {orbit-body }}$ by using E $_{\text {orbit-Tod }}$, quaternion, and PN matrix,

$$
\mathbf{M}_{\text {orbit-body }}=\mathbf{M}_{\text {t2000-body }} \times \mathbf{P N}^{t} \times \mathbf{E}_{\text {orbit-TOD }},
$$

where the superscript $t$ denotes transpose, and $\mathbf{M}_{\mathbf{J 2 0 0 0} \text {-body }}$ is defined by Eq. 3.4.8-1.
The same matrix $\mathbf{M}_{\text {I2000-body }}$ can be defined by roll $\varphi$, pitch $\theta$ and yaw $\psi$, where each angle is defined as the rotation (Euler) angle between orbit coordinate and satellite body:

$$
\begin{aligned}
\mathbf{M}_{\text {orbit-body }} & =\left(\begin{array}{lll}
M_{11} & M_{12} & M_{13} \\
M_{21} & M_{22} & M_{23} \\
M_{31} & M_{32} & M_{33}
\end{array}\right) \\
& =\left(\begin{array}{ccc}
1 & 0 & 0 \\
0 & \cos \varphi & \sin \varphi \\
0 & -\sin \varphi & \cos \varphi
\end{array}\right)\left(\begin{array}{ccc}
\cos \theta & 0 & -\sin \theta \\
0 & 1 & 0 \\
\sin \theta & 0 & \cos \theta
\end{array}\right)\left(\begin{array}{ccc}
\cos \psi & \sin \psi & 0 \\
-\sin \psi & \cos \psi & 0 \\
0 & 0 & 1
\end{array}\right) \\
& =\left(\begin{array}{ccc}
\cos \theta \cos \psi & \cos \theta \sin \psi & -\sin \theta \\
\sin \varphi \sin \theta \cos \psi-\cos \varphi \sin \psi & \sin \varphi \sin \theta \sin \psi+\cos \varphi \cos \psi & \sin \varphi \cos \theta \\
\cos \varphi \sin \theta \cos \psi+\sin \varphi \sin \psi & \cos \varphi \sin \theta \sin \psi-\sin \varphi \cos \psi & \cos \varphi \cos \theta
\end{array}\right)
\end{aligned}
$$

Eq. 3.5.8-6
By equation Eq. 3.5.8-5 and Eq. 3.5.8-6, roll $\varphi$, pitch $\theta$ and yaw $\psi$ can be obtained as follows:

$$
\begin{align*}
& \varphi=\operatorname{atan} 2\left(M_{23}, M_{33}\right) \\
& \theta=\operatorname{asin}\left(-M_{13}\right) \\
& \psi=\operatorname{atan} 2\left(M_{12}, M_{11}\right)
\end{align*}
$$

### 3.5.9. SolarGeometry group

SolarGeometry group contains apparent solar position and velocity (in ECR and TOD) in the sample lines of standardBand. The values take account of the light propagation time from sun to earth ( 8 minutes 19 seconds/ fixed value/ specified by parameters).

### 3.5.10. LunarGeometry group

LunarGeometry group contains true lunar position and velocity (in ECR and TOD) in the sample lines of standard band.

## 4. Image Processing

This chapter describes how to convert the digital number stored in ImageData group to the luminance with band-to-band registration

### 4.1. Processing flow

### 4.1.1. Band $2,3,4,7,8$ and 9

Figure $4-1$ shows the image processing for band $2,3,4,7,8$ and 9 .


Figure 4 -1 Image processing for Band $2,3,4,7,8$ and 9

The image is processed in following steps.
Radiometric Convert 12bit observation data to the luminance ( $\mathrm{W} / \mathrm{m}^{2} / \mu \mathrm{m} / \mathrm{str}$ ). conversion Refer to section 4.2.

Band-to-band Transform the image to be overlaid on the base band.
registration Refer to section 4.3

### 4.1.1. Band 1,6

Figure 4-2 shows the image processing for band 1 and 6.


Figure 4-2 Image processing for Band 1 and 6

The image is processed in following steps.

Radiometric Convert 12bit observation data to the luminance (W/m ${ }^{2 /} \mu \mathrm{m} / \mathrm{str}$ ).
conversion Refer to section 4.2.

Saturation Correct saturated pixel value with the estimated value of around correction corresponding pixel of band 2,7.

Refer to section 4.3 .

Stray light $\quad$ Correct stray light in 1 or 6 band.
correction Refer to section 4.5.

Out-of-band $\quad$ Correct out-of-band stray light from band 2,3,4 or 7, 8,9 .
stray light Refer to section 4.6.
correction
Band 1 and 6 Correct crosstalk between band 1 and band 6.
crosstalk Refer to section 4.7.
correction

Band-to-band Transform the image to be overlaid on the base band.
Registration Refer to section 4.3.

### 4.1.1. Band 5,10

Figure 4-3 shows the image processing for band 5 and 10 .


Figure 4-3 Image processing for Band 5 and 10

The image is processed in following steps.
Inter-channel Correct inter-channel crosstalk within the band.
crosstalk Refer to section 4.8.
correction

Radiometric Convert 12bit observation data to the luminance ( $\mathrm{W} / \mathrm{m}^{2} / \mu \mathrm{m} / \mathrm{str}$ ).
conversion Refer to section 4.2.
Stray light Correct stray light in band5 or band10.
correction Refer to section 4.9.
Band-to-band Transform the image to be overlaid on the base band.
registration Refer to section 4.3.

### 4.2. Radiometric conversion

Radiometric conversion method calculating from the digital values, DN to the brightness
( $\mathrm{W} / \mathrm{m}^{2} / \mu \mathrm{m} / \mathrm{str}$ ) is described below. In follows, temperatures are expressed in degrees Celsius.
(1) Correction of the DN value for valid pixels using pre-amplifier temperature $T_{1}$ and amplifier temperature $T_{2}$
The DN value of valid pixels $X(m, n . l)$ are corrected with pre-amplifier temperature $T_{1}$ and amplifier temperature $T_{2}$.

$$
Z_{1}(m, n, l)=\frac{X(m, n, l)}{C_{1}\left(m, T_{1}(m, l)\right) C_{2}\left(m, T_{1}(m, l)\right)}
$$

$m, n$ and $l$ are band number, pixel number and line number, respectively.

The coefficient $C_{1}$ and $C_{2}$ is calculated as follows:

$$
\begin{align*}
& C_{1}\left(m, T_{1}(m, l)\right)=\sum_{k=0}^{3} a(m, k) T_{1}^{k}(m, l) \\
& C_{2}\left(m, T_{2}(m, l)\right)=\sum_{k=0}^{3} b(m, k) T_{2}^{k}(m, l)
\end{align*}
$$

Eq. 4-3

Pre-amplifier temperature $T_{1}$ and amplifier temperature $T_{2}$ are stored in TemperatureTelemetry_1sec group of Common file. The coefficients of polynomial $a$ and $b$ are unique to each band. These are provided in separate file.
(2) The correction term $Z_{21}$ calculated by dark pixel

The correction term $Z_{21}$ can be calculated with average of dark pixels $X_{d k 1}$, pre-amplifier temperature $T_{1}$ and amplifier temperature $T_{2}$.

$$
\begin{array}{ll}
Z_{21 \_O D D}(m, n, l)=\frac{X_{d k 1 \_O D D}(m, l)}{C_{1}\left(m, T_{1}(m, l)\right) C_{2}\left(m, T_{2}(m, l)\right)} & \text { if } m \neq 5,10 \text { and } n=2 k-1(k=1,2, \ldots) \\
Z_{21_{-} E V E N}(m, n, l)=\frac{X_{d k 1 \_E V E N}(m, l)}{C_{1}\left(m, T_{1}(m, l)\right) C_{2}\left(m, T_{2}(m, l)\right)} & \text { if } m \neq 5,10 \text { and } n=2 k(k=1,2, \ldots) \\
Z_{21}(m, n, l)=\frac{X_{d k 1}(m, l)}{C_{1}\left(m, T_{1}(m, l)\right) C_{2}\left(m, T_{2}(m, l)\right)} & \text { if } m=5,10
\end{array}
$$

Eq. 4-4
Here $m, n$ and $l$ are band number, pixel number and line number, respectively.
$X_{\text {dkI_ODD }}(m, l)$ is the average of up to 4 dark pixels selected from pixel number1,3,5,7 in the l'
$\left(l-p w \leq l^{\prime} \leq l+p w, p w\right.$ is the parameter) near line $l$.
$X_{\text {dk1_EVEN }}(m, l)$ is the average of up to 4 dark pixels selected from pixel number2, $4,6,8$ in the $l$ ' $\left(l-p w \leq l^{\prime} \leq l+p w, p w\right.$ is the parameter $)$ near line $l$.
$X_{d k l}$ is the average of up to 6 dark pixels selected from pixel number 1 to 6 in the l' $\left(l-p w \leq l^{\prime} \leq l+p w, p w\right.$ is the parameter) near line $l$.

The coefficient $C_{1}$ and $C_{2}$ are the same as Eq. 4-2 and 4-3.
(3) The correction term of $\mathrm{Z}_{22}$ calculated by dark earth observation data, pixel temperature $T_{3}$ and exposure duration time $t_{\text {int }}$.
The correction term $Z_{22}$ can be calculated using the night observation data (the average of time series of $X_{d k 2}(m, n)$, pre-amplifier temperature $T^{\prime}{ }_{1}(m)$, amplifier temperature $T^{\prime}{ }_{2}(m)$ and pixel temperature $T_{3}{ }_{3}(m)$ and the exposure time $\left.t_{\text {int. }}(m, l)\right)$ band $m$, and exposure time tint $(m, l)$ at line $l$.

Eq. 4-5
Here $m, n$ and $l$ are band number, pixel number and line number, respectively.
$X_{d k 3 \_O D D}(m): X_{d k 2}$ is the average of up to 4 dark pixels selected from pixel $X_{d k 2}$ number1,3,5,7. $X_{d k 3 \_E V E N}(m): X_{d k 2}$ is the average of up to 4 dark pixels selected from pixel $X_{d k 2}$ number2,4,6,8. $X_{d k 3}(m) \quad: X_{d k 2}$ is the average of up to 6 dark pixels selected from pixel $X_{d k 2}$ number 1 to 6 . $X_{d k 2}$ is provided as the parameter file.

The coefficient $C_{1}$ and $C_{2}$ is calculated by substituting temperatures of pre amplifier $T_{1}{ }_{1}$, $T^{\prime}{ }_{2}$ into Eq.4-2, and Eq.4-3 at the acquisition of $X_{d k 2} . T^{\prime}{ }_{1}, T^{\prime}{ }_{2}$ are provided together with $X_{d k 2}$.

The coefficient $C_{3}$ and $C_{4}$ can be calculated as follows:

$$
\begin{align*}
& C_{3}\left(m, n, T_{3}^{\prime}(m)\right)=\sum_{k=0}^{3} c(m, n, k) T_{3}^{\prime k}(m) \\
& C_{4}\left(m, t_{\mathrm{int}}(m, l), t_{\mathrm{int}}^{\prime}(m)\right)=\sum_{k=0}^{3} d(m, k)\left(\frac{t_{\mathrm{int}}(m, l)}{t_{\mathrm{int}}^{\prime}(m)}\right)^{k}
\end{align*}
$$

Here $T_{3}^{\prime}$ is the pixel temperature at the acquisition of $X_{d k 2}$ and is provided together with $X_{d k 2}$.
Exposure time $t_{\text {int }}$ (ms) is stored for each line in LineAttribute group of forward/backward looking band file. $t^{\prime}$ (int $(\mathrm{ms})$ is the integration time at the acquisition of $X_{d k 2}$ and provide together with $X_{d k 2}$.
The coefficients of polynomial $c$ and $d$ are provided as the parameter file.
(4) The calculation of corrected pixel digital value $Z$

The corrected pixel digital value $Z$ can be calculated with $Z_{1}, Z_{21}$ and $Z_{22}$ which are defined in above (1) to (3),
$Z(m, n, l)$
$= \begin{cases}Z_{1}(m, n, l)-Z_{21 \_O D D}(m, n, l)-Z_{22 \_O D D}(m, n, l) & \text { if } m \neq 5,10 \text { and } n=2 k-1(k=1,2 \ldots) \\ Z_{1}(m, n, l)-Z_{21 \_E V E N}(m, n, l)-Z_{22 \_E V E N}(m, n, l) & \text { if } m \neq 5,10 \text { and } n=2 k(k=1,2 \ldots) \\ Z_{1}(m, n, l)-Z_{21}(m, n, l)-Z_{22}(m, n, l) & \text { if } m=5,10\end{cases}$
(5) Conversion to radiance $R a d$

The conversion from the corrected pixel digital value Z to radiance $\operatorname{Rad}\left[\mathrm{W} / \mathrm{m}^{2} / \mu \mathrm{m} / \mathrm{str}\right]$ is expressed as follows:

$$
\operatorname{Rad}(m, n, l)=R(m, n, 0)+\frac{1}{C_{5}\left(m, t_{\mathrm{int}}(m, l)\right) C_{6}\left(m, T_{3}(m, l)\right)} \sum_{k=1}^{3} R(m, n, k) Z^{k}(m, n, l)
$$

Eq. 4-9

The coefficients $C_{5}$ and $C_{6}$ are calculated as follows:

$$
\begin{align*}
& C_{5}\left(m, t_{\mathrm{int}}(m, l)\right)=\sum_{k=0}^{3} e(m, k) t_{\mathrm{int}}^{k}(m, l) \\
& C_{6}\left(m, T_{3}(m, l)\right)=\sum_{k=0}^{3} f(m, k) T_{3}^{k}(m, l)
\end{align*}
$$

The pixel temperature is stored in the TemperatureTelemetry_1sec group in the common file. The polynominal coeffficent $e, f$ and $R(m, n, k)$ are provided as the parameter file.

Basically $R(m, n, k)$ is constant value because is calculated from sensor gain and the sensor gain will not be changed in space. If sensor gain was changed for some reason, the $R(m, n, k)$ parameter file will be updated and the provided.

### 4.3. Band-to-band registration

Because the view area of the band is slightly different each other, we transform images by pre-calculated lookup table to be overlaid on the base band. The following equation shows the transformation.

$$
I M G_{i, B}(n, l)=I M G_{i}\left(n_{B}^{\prime}(i, n), l_{B}^{\prime}(i, n, l)\right)
$$

Where
$I M G_{i, B}(n, l):$ The image data of band $i$ overlaid on the band $B$ (pixel $n$, line $l$ )
$I M G_{i}(u, v)$ : The image data of band $i$ (pixel $u$, line $v$ )
$n^{\prime}{ }_{B}(i, n)$ : The pixel number of band $i$ corresponding to the band $B$ pixel $n$
$l_{B}^{\prime}(i, n, l):$ The line number of band $i$ corresponding to the band $B$ pixel $n$ line $l$

Figure 4-4 shows the example of the band-to-band registration.


Figure 4-4 Band-to-band registration
$n_{B}^{\prime}(i, n)$ is lookup table in the parameter file..
$l_{B}^{\prime}(i, n, l)$ is defined by following equation.

$$
l_{B}^{\prime}(i, n, l)=l_{s B}(i, l)+\Delta l_{B}(i, n)
$$

Where
$l_{s B}(i, l):$ The line number which was observed at the nearest observation time to band B
line $l$.
$\Delta l_{B}(i, n):$ The error of line number of band $i$.
$\Delta l_{B}(i, n)$ is lookup table in the parameter file..

### 4.4. Saturation correction

The observation data is 12 bit digital number (DN) so it will be saturated with 4095. For band1 and 6, the value before saturation is estimated using pixels around the saturated pixel and other bands. The following equations show the saturation correction.

Band 1

$$
I M G_{1}(n, l)=A_{1,2}(n, l) \cdot I M G_{2}(n, l)
$$

Band 6

$$
I M G_{6}(n, l)=A_{6,7}(n, l) \cdot I M G_{7}(n, l)
$$

Where
$I M G_{i}(n, l)$ : The luminance image data of band $i(i=1,6)$ (pixel $n$, line $\left.l\right)$
$A_{i, r}(n, l)$ : The coefficient for band $i$ (pixel $n$, line $l$ ), predicted by band $r$.

Figure $4-5$ shows the saturation correction (example of band1)


Figure 4-5 Example of saturation correction (in case of band1)
$A_{i, r}(n, l)$ is defined as the following average value of $I M G_{i} / I M G_{r}$ around the saturated pixel $n$. Where $n^{\prime}$ is the unsaturated nearest neighbor of the saturated pixel $n$.

$$
\begin{align*}
& A_{i, r}(n, l)=\frac{1}{N} \sum_{v=l-\Delta l}^{l+\Delta l} \sum_{u=n_{\min }^{\prime}}^{n_{\text {max }}^{\prime}} \frac{I M G_{i}(u, v)}{I M G_{r}(u, v)} \\
& n_{\min }^{\prime}= \begin{cases}n^{\prime}-\Delta n & \text { if } n^{\prime}<n \\
n^{\prime} & \text { otherwise }\end{cases} \\
& n_{\max }^{\prime}= \begin{cases}n^{\prime} & \text { if } n^{\prime}<n \\
n^{\prime}+\Delta n & \text { otherwise }\end{cases} \\
& N=(2 \Delta l+1)(\Delta n+1)
\end{align*}
$$

Eq.4.4-6

The both $\Delta n$ and $\Delta l$ are parameters which mean the range to calculate $A_{i, r}(n, l)$. No correction will be applied for $\operatorname{IMG}(\mathrm{n}, \mathrm{l})$, when data is missing or $\operatorname{IMG}(\mathrm{n}, \mathrm{l})$ is 0 .

### 4.5. Stray light correction for band 1 and 6

The observation data of band 1 and band 6 are contaminated by stray light component. We estimate the stray light component by the convolution of observation data and "Point Spread Function (PSF)", and remove it from the observation data.

The following equations show the stray light correction for band 1 and 6 .

Band 1

$$
I M G_{1}(n, l)=A_{1}(n)\left(I M G_{1}(n, l)-H_{1, a}(u, v) \otimes I M G_{1}(n, l)\right)
$$

Band 6

$$
I M G_{6}(n, l)=A_{6}(n)\left(I M G_{6}(n, l)-H_{6, a}(u, v) \otimes I M G_{6}(n, l)\right)
$$

Where
$I M G_{i}(n, l)$ : The luminance image data of band $i(i=1,6)$ (pixel $n$, line $\left.l\right)$
$H_{i, a}(u, v)$ : The stray light PSF for band1 $i(i=1,6)$ (pixel $u$, line $v$ )
$\otimes$ : The convolution operator
$A_{i}(n)$ : The coefficient to correct signal reduction by subtraction of the stray light component. (Pixel $n$ )

Figure 4-6 shows the example of stray light correction.


Figure 4-6 Stray light correction (example of band1)

Because stray light depends on the pixel position in image, we use five PSF $H_{i, a}(u, v)$ from $H_{i, 1}(u, v)$ to $H_{i, 5}(u, v)$ for the correction. Figure 4-7 shows example of PSF and applying area.


Figure 4-7 PSF and applying area (example of band1)
$H_{i, a}(u, v)$ and the applying area is defined in the parameter file.
PSF applying area is defined within the valid pixel area. We apply the border processing to out of the valid pixel area. The border processing is described in section 4.11.
Before the convolution, the missing line in $\operatorname{IMG}_{i}(n, l)$ must be filled by the interpolation. The interpolation is described in section 4.10.
$A_{i}(n)$ is the coefficient in the parameter file.

### 4.6. Out-of-band stray light correction for band 1 and 6

The observation data of band 1 and band6 also are contaminated with stray light component coming from out-of-band. As same as section 4.5, we calculate the stray light component by the convolution of observation data and PSF, and remove it from the observation data.
The following equations show the out-of-band stray light correction for band 1 and 6 .

## Band 1

$$
I M G_{1}(n, l)=I M G_{1}(n, l)-\sum_{j=2,3,4} H_{1, j}(u, v) \otimes I M G_{j}(n, l)
$$

Band 6

$$
I M G_{6}(n, l)=I M G_{6}(n, l)-\sum_{j=7,8,9} H_{6, j}(u, v) \otimes I M G_{j}(n, l)
$$

Where
$I M G_{i}(n, l):$ The luminance image data of band $i(i=1,2,3,4,6,7,8,9)$ (pixel $n$, line $l$ )
$H_{i, j}(u, v)$ : The band $j(j=2,3,4,7,8,9)$ out-band-stray light PSF for band $i(i=1,6)$ (pixel $u$, line $v$ )
$\otimes$ : The convolution operator

Figure $4-8$ shows the out-of-band stray light correction.

PSF applying area is defined within the valid pixel area. We apply the border processing to out of the valid pixel area. The border processing is described in section 4.11.

Before the convolution, the missing line in $I M G_{j}(n, l)(j=2,3,4,7,8,9)$ must be filled by the interpolation. The interpolation is described in section 4.10.


Band 1 (before out-of-band stray light correction ) $I M G_{1}$
Band 1 (after out-of-band stray light correction) IMG $_{1}$


Band 2 IMG $_{2}$


Band 2 stray light PSF $H_{1,2}$


Band $3 \mathrm{IMG}_{3}$


Band $4 \mathrm{IMG}_{4}$


$$
H_{1,4}
$$

Figure 4-8 Out-of-band stray light correction (example of band1)

### 4.7. Band 1 and 6 crosstalk correction

The observation data of band 1 and band6 also are contaminated with crosstalk of each other. As same as section 4.5, we calculate the crosstalk component by the convolution of observation data and PSF, and remove it from the observation data.
The following equations show the band 1 and 6 crosstalk correction.

Band 1

$$
I M G_{1}(n, l)=I M G_{1}(n, l)-H_{6}(u, v) \otimes I M G_{6,1}(n, l)
$$

Band 6

$$
I M G_{6}(n, l)=I M G_{6}(n, l)-H_{1}(u, v) \otimes I M G_{1,6}(n, l)
$$

Where
$I M G_{i}(n, l)$ : The luminance image data of band $i(i=1,6)($ pixel $n$, line $l)$
$\operatorname{IMG}_{r, i}(n, l)$ : The luminance image data of band $r(r=1,6)$ which is extracted by observation time of band $i$
$H_{i}(u, v)$ : The band $i(i=1,6)$ crosstalk PSF
$\otimes$ : The convolution operator

Figure 4-9 shows the example of the band-to-band registration for crosstalk correction between band 1 and 6 . For each line of band $i, \operatorname{IMG}_{r, i}(n, l)$ is defined by the band $r$ which observation time is nearest time to band $i$. If the time difference between band $i$ line $l$ and the its nearest line of band $r$ is greater than the double of nominal observation cycle, we assume the line of $I M G_{i}(n, l)$ is a missing line. Before the convolution, the missing line in $I M G_{r, i}(n, l)$ must be filled by the interpolation. The interpolation is described in section 4.10.

Only forward or backward looking data are filled with 0 in $I M G_{r, i}(n, l)$
PSF applying area is defined in the valid pixel area. We apply the border processing to out of the valid pixel area. The border processing is described in section 4.11.


Figure 4-9 Band-to-band registration for band 1 and 6 crosstalk correction.
4.8. Inter-channel crosstalk correction for band 5 and 10

The observation data of band 5 and band10 are contaminated with the electrical inter-channel crosstalk
between the data reading channels (CH). Figure 4-10 shows the relation of pixels and channels. Figure 4-11 shows the example of inter-channel crosstalk.

Figure 4-10 Band 5 and 10 pixels and channels


Following equation shows to remove the crosstalk component.
Where $C H_{i}(n, l)$ is the observation data as the digital number of band $i$ pixel $n$, line $l$.

$$
\begin{aligned}
C H_{1}\left(n_{1}, l\right)= & C H_{1}\left(n_{1}, l\right) \\
& -\left(a_{31} \times C H_{3}\left(n_{3}, l\right)+a_{51} \times C H_{5}\left(n_{5}, l\right)+a_{71} \times C H_{7}\left(n_{7}, l\right)\right) \\
& -\left(b_{31} \times\left|\frac{\partial C H_{3}\left(n_{3}, l\right)}{\partial n}\right|+b_{51} \times\left|\frac{\partial C H_{5}\left(n_{5}, l\right)}{\partial n}\right|+b_{71} \times\left|\frac{\partial C H_{7}\left(n_{7}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq. 4.8-1

$$
\begin{aligned}
C H_{3}\left(n_{3}, l\right)= & C H_{3}\left(n_{3}, l\right) \\
& -\left(a_{13} \times C H_{1}\left(n_{1}, l\right)+a_{53} \times C H_{5}\left(n_{5}, l\right)+a_{73} \times C H_{7}\left(n_{7}, l\right)\right) \\
& -\left(b_{13} \times\left|\frac{\partial C H_{1}\left(n_{1}, l\right)}{\partial n}\right|+b_{53} \times\left|\frac{\partial C H_{5}\left(n_{5}, l\right)}{\partial n}\right|+b_{73} \times\left|\frac{\partial C H_{7}\left(n_{7}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq. 4.8-2

$$
\begin{aligned}
C H_{5}\left(n_{5}, l\right)= & C H_{5}\left(n_{5}, l\right) \\
& -\left(a_{15} \times C H_{1}\left(n_{1}, l\right)+a_{35} \times C H_{3}\left(n_{3}, l\right)+a_{75} \times C H_{7}\left(n_{7}, l\right)\right) \\
& -\left(b_{15} \times\left|\frac{\partial C H_{1}\left(n_{1}, l\right)}{\partial n}\right|+b_{35} \times\left|\frac{\partial C H_{3}\left(n_{3}, l\right)}{\partial n}\right|+b_{75} \times\left|\frac{\partial C H_{7}\left(n_{7}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq. 4.8-3

$$
\begin{aligned}
C H_{7}\left(n_{7}, l\right)= & C H_{7}\left(n_{7}, l\right) \\
& -\left(a_{17} \times C H_{1}\left(n_{1}, l\right)+a_{37} \times C H_{3}\left(n_{3}, l\right)+a_{57} \times C H_{5}\left(n_{5}, l\right)\right) \\
& -\left(b_{17} \times\left|\frac{\partial C H_{1}\left(n_{1}, l\right)}{\partial n}\right|+b_{37} \times\left|\frac{\partial C H_{3}\left(n_{3}, l\right)}{\partial n}\right|+b_{57} \times\left|\frac{\partial C H_{5}\left(n_{5}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq. 4.8-4

$$
\begin{array}{ll}
n_{1}=2 m+1 & \left(n_{1}=1,3,5, \ldots, 255\right) \\
n_{3}=256+2 m+1 & \left(n_{3}=257,259,261, \ldots, 511\right) \\
n_{5}=512+2 m+1 & \left(n_{5}=513,515,517, \ldots, 767\right) \\
n_{7}=768+2 m+1 & \left(n_{7}=769,711,713, . ., 1023\right)
\end{array}
$$

$$
(0 \leq m<128)
$$

Eq. 4.8-5

$$
\begin{aligned}
C H_{2}\left(n_{2}, l\right)= & C H_{2}\left(n_{2}, l\right) \\
& -\left(a_{42} \times C H_{4}\left(n_{4}, l\right)+a_{62} \times C H_{6}\left(n_{6}, l\right)+a_{82} \times C H_{8}\left(n_{8}, l\right)\right) \\
& -\left(b_{42} \times\left|\frac{\partial C H_{4}\left(n_{4}, l\right)}{\partial n}\right|+b_{62} \times\left|\frac{\partial C H_{6}\left(n_{6}, l\right)}{\partial n}\right|+b_{82} \times\left|\frac{\partial C H_{8}\left(n_{8}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq. 4.8-6

$$
\begin{aligned}
C H_{4}\left(n_{4}, l\right)= & C H_{4}\left(n_{4}, l\right) \\
& -\left(a_{24} \times C H_{2}\left(n_{2}, l\right)+a_{64} \times C H_{6}\left(n_{6}, l\right)+a_{84} \times C H_{8}\left(n_{8}, l\right)\right) \\
& -\left(b_{24} \times\left|\frac{\partial C H_{2}\left(n_{2}, l\right)}{\partial n}\right|+b_{64} \times\left|\frac{\partial C H_{6}\left(n_{6}, l\right)}{\partial n}\right|+b_{84} \times\left|\frac{\partial C H_{8}\left(n_{8}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

$$
\begin{aligned}
C H_{6}\left(n_{6}, l\right)= & C H_{6}\left(n_{6}, l\right) \\
& -\left(a_{26} \times C H_{2}\left(n_{2}, l\right)+a_{46} \times C H_{4}\left(n_{4}, l\right)+a_{86} \times C H_{8}\left(n_{8}, l\right)\right) \\
& -\left(b_{26} \times\left|\frac{\partial C H_{2}\left(n_{2}, l\right)}{\partial n}\right|+b_{46} \times\left|\frac{\partial C H_{4}\left(n_{4}, l\right)}{\partial n}\right|+b_{86} \times\left|\frac{\partial C H_{8}\left(n_{8}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq.4.8-8

$$
\begin{aligned}
C H_{8}\left(n_{8}, l\right)= & C H_{8}\left(n_{8}, l\right) \\
& -\left(a_{28} \times C H_{2}\left(n_{2}, l\right)+a_{48} \times C H_{4}\left(n_{4}, l\right)+a_{68} \times C H_{6}\left(n_{6}, l\right)\right) \\
& -\left(b_{28} \times\left|\frac{\partial C H_{2}\left(n_{2}, l\right)}{\partial n}\right|+b_{48} \times\left|\frac{\partial C H_{4}\left(n_{4}, l\right)}{\partial n}\right|+b_{68} \times\left|\frac{\partial C H_{6}\left(n_{6}, l\right)}{\partial n}\right|\right)
\end{aligned}
$$

Eq.4.8-9

$$
\begin{array}{ll}
n_{2}=2 m+2 & \left(n_{2}=2,4,6, \ldots, 256\right) \\
n_{4}=256+2 m+2 & \left(n_{4}=258,260,262, \ldots, 512\right) \\
n_{6}=512+2 m+2 & \left(n_{6}=514,516,518, \ldots, 768\right) \\
n_{8}=768+2 m+2 & \left(n_{8}=770,712,714, \ldots, 1024\right) \\
& (0 \leq m<128)
\end{array}
$$

Eq. 4.8-10
The coefficients $a$ and $b$ are defined in the parameter file.
$\partial C H_{i}(n, l) / \partial n$ is defined in following equation.

$$
\frac{\partial C H_{i}(n, l)}{\partial n}=\left\{\begin{array}{ll}
C H_{i}(n+2, l)-C H_{i}(n-2, l) & \text { if } 0<n-2 \\
0 & \text { otherwise }
\end{array} \text { and } n+2 \leq 1024\right.
$$

Eq. 4.8-11

### 4.9. Stray light correction for band 5 and 10

The observation data of band 5 and band10 are also contaminated with stray light component. As same as section 4.5 , we estimate the stray light component by the convolution of observation data and PSF, and remove it from the observation data.
The following equation shows the stray light correction for band 5 and 10 .

Band 5

$$
I M G_{5}(n, l)=A_{5}(n)\left(I M G_{5}(n, l)-H_{5, a}(u, v) \otimes I M G_{5}(n, l)\right)
$$

Band 10

$$
I M G_{10}(n, l)=A_{10}(n)\left(I M G_{10}(n, l)-H_{10, a}(u, v) \otimes I M G_{10}(n, l)\right)
$$

Where
$I M G_{i}(n, l)$ : The luminance image data of band $i(i=5,10)$ (pixel $n$, line $l$ )
$H_{i, a}(u, v)$ : The stray light PSF for band1 $i(i=5,10)$ (pixel $u$, line $v$ )
$\otimes$ : The convolution operator
$A_{i}(n)$ : The coefficient to correct signal reduction by subtraction of the stray light component.

Figure 4-12 shows the example of stray light correction.


Figure 4-12 Stray light correction (example of band 5)

Because stray light depends on the pixel position in image, we use five $H_{i, a}(u, v)$ from $H_{i, 1}(u, v)$ to $H_{i, 5}(u, v)$. Figure 4-13 shows example of PSF and applying area.


Figure 4-13 PSF and applying area (example of band5)
$H_{i, a}(u, v)$ and the applying area is defined in the parameter file.
PSF applying area is defined within the valid pixel area. We apply the border processing to out of the valid pixel area. The border processing is described in section 4.11.

Before the convolution, the missing line in $\operatorname{IMG}_{i}(n, l)$ must be filled by the interpolation. The interpolation is described in section 4.10.
$A_{i}(n)$ is the coefficient in the parameter file.

### 4.10. Filling missing line by interpolation

As the convolution operation needs to be applied to the spatially continuous image data, the missing lines in the image must be filled by interpolation. The following equation shows the interpolation.

$$
I M G_{i}(n, l)=\frac{I M G_{i}\left(n, l_{\max }\right)-I M G_{i}\left(n, l_{\min }\right)}{l_{\max }-l_{\min }}\left(l-l_{\min }\right)+I M G_{i}\left(n, l_{\min }\right)
$$

Where

$$
\begin{aligned}
& I M G_{i}(n, l) \text { : The luminance image data of band } i(i=1,6)\left(\text { pixel } n \text {, line } l: l_{\min }<l<l_{\max }\right) \\
& l_{\min }, l_{\max }: \text { The pre/post non-missing line number of the missing line. }
\end{aligned}
$$

Figure 4-14 shows the interpolation of the missing line.


Figure 4-14 Interpolation of the missing line

As the result of band-to-band registration, the first and/or last line of the target image might be missing line. If the first line of the target image is missing line, the lines will be filled with zero between the first line and the first normal line. If the last line of the target image is missing line, the lines will be filled with zero between the last normal line and the last line.

### 4.11. Image border processing

PSF and its applying area to the image are provided as the parameter file.
The followings are normal range.

Band 1,2,3,4,6,7,8,9 : Pixel 9 to 2056
Band 5, 10 : Pixel 7 to 1024 (include 7 to 66 pixel)

Figure 4-14 shows the PSF applying area.
The pixels out of applying area will be filled with the value of the border area.
For pixel direction, the border filling width is specified by parameter. The parameter specifies the physical width of sensor receiving potentially extra light. The pixels out of the border range are filled with zero. Figure $4-15$ shows the image border processing.


Figure 4-15 PSF applying area


White box shows the original image size of target image.
$\mathrm{W}_{\text {outside }}$ is a parameter. $\mathrm{W}_{\text {PSF }}$ and HPSF are width/height of PSF.

## 5. Geometric conversion

Setting of viewing vector, coordinate conversion to ECR coordinate and calculation of observation point are described below.
(1) Definition of view vector in sensor (CAI-2) coordinate

View vector in sensor (CAI-2) coordinate $\mathbf{v}_{\text {sensor }}(m, n)$ is calculated below formula as a unit vector for each band $m$ and Pixel number $n$.

$$
\begin{aligned}
& \mathbf{v}_{\text {sensor }}(m, n)=\frac{1}{\sqrt{x^{2}(m, n)+y^{2}(m, n)+z^{2}(m, n)}}\left(\begin{array}{l}
x(m, n) \\
y(m, n) \\
z(m, n)
\end{array}\right) \\
& x(m, n)=\sum_{j=0}^{10} g_{j x}(m) \cdot p^{j}(m, n) \\
& y(m, n)=\sum_{j=0}^{10} g_{j y}(m) \cdot p^{j}(m, n) \\
& z(m, n)=\sum_{j=0}^{10} g_{j z}(m) \cdot p^{j}(m, n) \\
& p(m, n)=p_{\text {det }}(m) \cdot\left(n-p_{c}(m)\right)
\end{aligned}
$$

Eq.5-1
Here,
$g_{j k}(m)$ the coefficient of view vector of band $m$, order $j . k(k=x, y, z)$ means the axis.
$p(m, n)$ the pixel position of pixel $n$, band $m$ from the reference position on the detector. (Unit:mm).
$p_{\text {det }}(m)$ the pixel pitch of band $m$ of the detector (Unit:mm).
$p_{\mathrm{c}}(m)$ the reference pixel position of band m , on the detector (Unit :pixel)
$g_{j k}(m), p_{\operatorname{det}}(m)$ and $p_{\mathrm{c}}(m)$ are provided in the parameter file.
(2) Conversion from sensor (CAI-2) coordinate to satellite coordinate

Conversion from view vector in sensor (CAI-2) coordinate $\mathbf{v}_{\text {sensor }}$ to view vector in satellite-fixed coordinate $\mathbf{v}_{\text {body }}$ is expressed as follows:

$$
\mathbf{v}_{\text {body }}=\mathbf{M}_{\text {sensor-body }} \times \mathbf{v}_{\text {sensor }}
$$

$\mathbf{M}_{\text {sensor-body }}$ is coordinate transformation matrix convert from sensor (CAI-2) coordinate to satellite coordinate and provided in another file.
(3) Conversion from satellite coordinate to ECR coordinate

Using transformation matrix (satToECR_Matrix) $\mathbf{M}_{\text {body-ECR }}$ which is stored in SatelliteGeometry group, conversion from view vector in satellite coordinate $\mathbf{v}_{\text {body }}$ to view vector in ECR(WGS84) $\mathbf{v}_{\text {ECR }}$ is expressed as follows:

$$
\mathbf{v}_{\text {ECR }}=\mathbf{M}_{\text {body-ECR }} \times \mathbf{v}_{\text {body }}
$$

Conversion from satellite coordinate to J2000.0, TOD, pseudo earth-fixed and ECR coordinate without using $\mathbf{M}_{\text {body-ECR }}$ is described below.
(4) Conversion from satellite coordinate to J2000.0 coordinate

Conversion from view vector in satellite coordinate $\mathbf{v}_{\text {body }}$ to view vector in J2000.0 coordinate $\mathbf{v}_{\mathbf{J 2 0 0 0}}$ is expressed as follows:

$$
\mathbf{v}_{\text {J2000 }}=\mathbf{M}_{\text {body }-\mathbf{J} 2000} \times \mathbf{V}_{\text {body }}
$$

$\mathbf{M}_{\text {body-J2000 }}$ is coordinate transformation matrix converting from satellite coordinate to J2000.0 coordinate using satellite attitude data (quaternion) which is stored in AttitudeData group of Common file (see Chapter 3.4.8).
(5) Conversion from J2000.0 coordinate to TOD coordinate

Using PN matrix which is stored in TransMatrixInfo group of Common file, conversion from view vector in J2000.0 coordinate $\mathbf{v}_{\mathbf{J} 2000}$ to view vector in TOD coordinate $\mathbf{v}_{\text {TOD }}$ is expressed as follows:

$$
\mathbf{v}_{\text {TOD }}=\mathbf{P N} \times \mathbf{v}_{\text {J2000 }}
$$

(6) Conversion from TOD coordinate to pseudo earth-fixed coordinate

Conversion from view vector in TOD coordinate $\mathbf{v}_{\text {TOD }}$ to view vector in pseudo earth-fixed coordinate (without considering the polar motion) $\mathbf{V}_{\text {Pecr }}$ is expressed as follows:

$$
\mathbf{v}_{\text {PECR }}=\mathbf{M}_{\text {TOD-PECR }} \times \mathbf{v}_{\text {TOD }}
$$

$\mathbf{M}_{\text {TOD-PECR }}$ is coordinate transformation matrix convert from TOD coordinate to pseudo earth-fixed coordinate and calculated using Greenwich sidereal time SiderealTimeInfo group of Common file (see Chapter 3.4.3).
(7) Conversion from pseudo earth-fixed coordinate to ECR coordinate

Using XY matrix is stored in TransMatrixInfo group of Common file, conversion from view vector in pseudo earth-fixed coordinate (without considering polar motion) $\mathbf{v}_{\text {Pecr }}$ to view vector in ECR $\mathbf{v}_{\text {ECR }}$ is expressed as follows:

$$
\mathbf{v}_{\mathrm{ECR}}=\mathbf{X Y} \times \mathbf{v}_{\mathbf{P E C R}}
$$

(8) Calculation of observation point on the earth ellipsoid

Using view vector in ECR coordinate $\mathbf{v}_{\text {ECR }}=\left(v_{x}, v_{y}, v_{z}\right)^{t}$ (the superscript $t$ denotes transpose), sensor position vector $\mathbf{p}_{\text {sat }}=\left(p_{\text {sat } x}, p_{\text {sat } y}, p_{\text {sat }}\right)^{t}$ and observation point vector on the earth ellipsoid $p_{\text {obs }}=\left(p_{\text {obs }} \text {, }, p_{\text {obs_ } y,}, p_{\text {obs }}^{-} \text {}\right)^{t}$, observation point is expressed as follows:

$$
\left(\begin{array}{c}
p_{\text {obs }_{-x} x} \\
p_{\text {obs }_{-y}} \\
p_{\text {obs }_{-} z}
\end{array}\right)=\left(\begin{array}{l}
p_{\text {sat }_{-} x} \\
p_{\text {sat }_{-} y} \\
p_{\text {sat }_{-} z}
\end{array}\right)+k\left(\begin{array}{l}
v_{x} \\
v_{y} \\
v_{z}
\end{array}\right)
$$

$k$ is intermediate variable.

When the equatorial radius and polar radius on the earth ellipsoid is $R e$ and $R p$, $\mathrm{p}_{\text {obs }}=\left(p_{\text {obs_ }}, p_{\text {obs }}^{y}, \mathrm{y}, p_{\text {obs }}^{-} \text {}\right)^{t}$ satisfies the following relational expression :

$$
\frac{p_{o b s_{-}-x}^{2}+p_{o b s_{-} y}^{2}}{R_{e}^{2}}+\frac{p_{o b s_{-} z}^{2}}{R_{p}^{2}}=1
$$

Assigning Eq.5-7 to Eq.5-8, quadratic equation for $k$ is obtained as follows:

$$
a k^{2}+2 b k+c=0
$$

where

$$
\left\{\begin{array}{l}
a=R_{p}^{2}\left(v_{x}^{2}+v_{y}^{2}\right)+R_{e}^{2} v_{z}^{2} \\
b=R_{p}^{2}\left(p_{\text {sat }-x} v_{x}+p_{\text {sat }-y} v_{y}\right)+R_{e}^{2} p_{\text {sat }-z} v_{z} \\
c=R_{p}^{2}\left(p_{\text {sat }-x}^{2}+p_{\text {sat }-y}^{2}\right)+R_{e}^{2} p_{\text {sat }}^{-z}
\end{array}-R_{e}^{2} R_{p}^{2} .\right.
$$

Then, Eq. 5-9 is solved for $k$.

$$
k=\frac{-b-\sqrt{b^{2}-a c}}{a}
$$

In case $b^{2}-a c<0$ and $k<0$, observation point is outside of earth surface.
Assigning $k$ to Eq.5-7, observation point vector $\mathrm{p}_{\text {obs }}=\left(p_{\text {obs_x }}, p_{\text {obs } y}, p_{\text {obs_ }}\right)^{t}$ can be calculated.

Figure 5-1 shows each vector.


## Figure 5-1 Calculation of observation point

(9) Calculation of geographic latitude/longitude

Geographic longitude $\lambda$ corresponding observation point vector of the earth ellipsoid $\mathbf{p}_{\text {obs }}=\left(p_{\text {obs }}, p_{\text {obs }}^{-y}, p_{o b s_{-}}\right)^{t}$ is expressed as follows:

$$
\lambda=\operatorname{atan} 2\left(p_{o b s_{-} y}, p_{o b s_{-} x}\right)
$$

Using geocentric latitude $\psi$, geographic latitude $\varphi$ can be calculated as follows:

$$
\begin{align*}
& \psi=\operatorname{asin}\left(\frac{p_{o b s_{-} z}}{\sqrt{p_{o b s_{-} x}^{2}+p_{o b s_{-} y}^{2}+p_{o b s_{-} z}^{2}}}\right) \\
& \varphi=\operatorname{atan} 2\left(\sin \psi, \frac{R_{p}^{2}}{R_{e}^{2}} \cos \psi\right)
\end{align*}
$$

Eq. 5-14

Definition of atan2 function is described below.

$$
\operatorname{atan} 2(y, x)=\left\{\begin{array}{cc}
\tan ^{-1}\left(\frac{y}{x}\right) & (x>0) \\
\tan ^{-1}\left(\frac{y}{x}\right)+\pi & (x<0, y \geq 0) \\
\tan ^{-1}\left(\frac{y}{x}\right)-\pi & (x<0, y<0) \\
\frac{\pi}{2} & (x=0, y>0) \\
-\frac{\pi}{2} & (x=0, y<0) \\
\text { undefined } & x=0, y=0
\end{array}\right.
$$

## 6.Format Details

The details of product (HDF5) and L1 processing result file (XML) format are described below.

Table 6-1 shows the format details of Common file (HDF5) and Table 6-2 shows the format details of Forward/Backward looking band file (L1A/ HDF5).

Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (1/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (2/12)

| Group Path/Dataset Name |  | Data size |  | Data Type | Dataset Name | Explanation (Format) | Unit | Significant digit | Invalid Value | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | endDateFwd | Dimension | Size | H5T_STRING | End date of scene of Forward looking band | End date of scene of Forward looking band (UTC) Time format: YYYY-MM-DDThh:mm:ss. ffffffZ (28bytes, the last 1 byte is a null terminated string) <br> If there is no forward looking data in this scene, "(2bytes, the last 1 byte is a null terminated string) is stored. | UTC | Sigmicant disit | Value | The latest observation time of forward looking without integration time is stored. The time here is the integration start time and common to all bands, because the integration time depends on bands. |
|  | startDateBwd | 1 | 1 | H5T_STRING | Start date of scene of Backward looking band | Start date of scene of Backward looking band (UTC) Time format: YYYY-MM-DDThh:mm:ss. ffffffZ (28bytes, the last 1 byte is a null terminated string) <br> If there is no backward looking data in this scene, ""(2bytes, the last 1 byte is a null terminated string) is stored. | UTC | - | "-" | The oldest observation time of backward looking without integration time is stored. The time here is the integration start time and common to all bands, because the integration time depends on bands. |
|  | endDateBwd | 1 | 1 | H5T_STRING | End date of scene of Backward looking band | End date of scene of Backward looking band (UTC) Time format : YYYY-MM-DDThh:mm:ss.ffffffZ (28bytes, the last 1 byte is a null terminated string) <br> If there is no backward looking data in this scene, ""(2bytes, the last 1 byte is a null terminated stringe) is stored. | UTC | - | "-" | The latest observation time of backward looking without integration time is stored. The time here is the integration start time and common to all bands, because the integration time depends on bands. |
|  | geodeticDatum | 1 | 1 | H5T_STRING | Geodetic datum | "WGS84/ WGS84": Reference EII ipsoid Model/Frame of Reference <br> Fixed (14bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | satelliteName | 1 | 1 | H5T_STRING | Satellite Name | "GOSAT-2": Greenhouse gases Observing SATellite-2 Fixed (8bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | sensorName | 1 | 1 | H5T_STRING | Sensor Name | "TANSO-CAI-2": Cloud and Aerosol Imager-2 Fixed (12bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | process ingLevel | 1 | 1 | H5T_STRING | Processing Level | "L1A": Level 1A (4bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | algor ithmVersion | 1 | 1 | H5T_STRING | Algorithm Version | Algorithm vertsion in processing control information is stored. <br> (4bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | parameterVersion | 1 | 1 | H5T_STRING | ParameterVersion | Parameter vertsion in processing control information is stored. (4bytes, the last 1 byte is a null terminated string) | - | - | - |  |
|  | processingFacility | 1 | 1 | H5T_STRING | Processing facility name | "G2MDP": Mission Operations System Data Processing "JSS": JAXA Super computer System "EORC": Earth Observation Research Center (the size is the length of string above plus 1byte) | - | - | - |  |

Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (3/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (4/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (5/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (6/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (7/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (8/12)

| Group Path/Dataset Name |  | Data size |  | Data Type | Dataset Name | Explanation (Format) | Unit | Significant digit | Invalid Value | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\square$ SolarEphemer is |  |  |  |  |  |  |  |  |  |  |
|  | numData | 1 | 1 | H5T_STD_I32LE | Number of data | Number of solar ephemeris data is stored. |  | - | - | 0 |  |
|  | startDate | 1 | 1 | H5T_STRING | Start date of solar ephemer is data (UTC) | Start date of solar ephemeris data (UTC) is stored. Time format: YYYY-MM-DDThh:mm:ss.ffffffZ (28bytes, the last 1 byte is a null terminated string) | UTC | - | - | There is no dataset if numData is 0 . |
|  | startDate_Cont inuousTime | 1 | 1 | H5T_IEEE_F64LE | Start date of solar ephemer is data (seconds) | Total seconds of start date of solar ephemer is data since 23:59:59 UTC, Dec 31, 2012. | sec | - | - | There is no dataset if numData is 0 . |
|  | time | 1 | numData | H5T_IEEE_F64LE | Elapse time from Start time start date of solar ephemer is data | The elapsed seconds from start date of solar ephemeris data. | sec | 10 | - | There is no dataset if numData is 0 . |
|  | posECR | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{array}{\|l} \hline \text { Solar Position } \\ \text { Vector (ECR) } \end{array}$ | Solar Position Vector in ECR is stored. (x, y, z) ECR (WGS84) | km | 10 | - | There is no dataset if numData is 0 . |
|  | velECR | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Solar Velocity } \\ & \text { Vector (ECR) } \end{aligned}$ | Solar Velocity Vector in ECR is stored. (u, v, w) ECR (WGS84) | km/s | 10 | - | There is no dataset if numData is 0 . |
|  | posECI | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{array}{\|l} \hline \text { Solar Position } \\ \text { Vector (ECI) } \end{array}$ | Solar Position Vector in ECI is stored. ( $\mathrm{x}, \mathrm{y}, \mathrm{z}$ ) ECI (TOD) | km | 10 | - | There is no dataset if numData is 0 . |
|  | velECI | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{array}{\|l} \hline \text { Solar Velocity } \\ \text { Vector (ECI) } \end{array}$ | Solar Velocity Vector in ECI is stored. (u, v, w) ECI (TOD) | km/s | 10 | - | There is no dataset if numData is 0 . |
| , | LunarEphemer is |  |  |  |  |  |  |  |  |  |
|  | numData | 1 | 1 | H5T_STD_I32LE | Number of data | Number of lunar ephemeris data is stored. | - | - | 0 |  |
|  | startDate | 1 | 1 | H5T_STRING | Start date of lunar ephemer is data (UTC) | Start date of lunar ephemer is data (UTC) is stored. Time format : YYYY-MM-DDThh:mm:ss.ffffffZ <br> (28bytes, the last 1 byte is a null terminated string) | UTC | - | - | There is no dataset if numData is 0 . |
|  | startDate_Cont inuousTime | 1 | 1 | H5T_IEEE_F64LE | Start date of lunar ephemer is data (seconds) | Total seconds of start date of lunar ephemer is data since 23:59:59 UTC, Dec 31, 2012. | sec | - | - | There is no dataset if numData is 0 . |
|  | time | 1 | numData | H5T_IEEE_F64LE | Elapse time from start date of lunar ephemeris data | The elapsed seconds from start date of lunar ephemer is data. | sec | 10 | - | There is no dataset if numData is 0 . |
|  | posECR | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{array}{\|l} \hline \text { Lunar Position } \\ \text { Vector (ECR) } \end{array}$ | Lunar Position Vector in ECR is stored. ( $x, y, z$ ) ECR(WGS84) | km | 10 | - | There is no dataset if numData is 0 . |
|  | ve IECR | 2 | numData, 3 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Lunar Velocity } \\ & \text { Vector (ECR) } \end{aligned}$ | Lunar Velocity Vector in ECR is stored. ( $u, v, w)$ ECR(WGS84) | km/s | 10 | - | There is no dataset if numbata is 0 . |

Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (9/12)

| Group Path/Dataset Name |  | Data size |  | Data Type | Dataset Name | Explanation (Format) | Unit | Significant digit | Invalid Value | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | posECI | Dimension | Size | H5T_IEEE_F64LE | Lunar Position Vector (ECI) | Lunar Position Vector in ECI is stored. ( $x, y, z$ ) ECI (TOD) | km | \| | Value | There is no dataset if numData is 0 . |
|  | velECI | 2 | numData, 3 | H5T_IEEE_F64LE | Lunar Velocity <br> Vector (ECI) | Lunar Velocity Vector in ECI is stored. ( $u, v, w)$ ECI (TOD) | km/s | 10 | - | There is no dataset if numData is 0 . |
| DTemperatureTelemetry_1sec (Temperature telemetry of 1 sec period) |  |  |  |  |  |  |  |  |  |  |
|  | numData | , | 1 | H5T_STD_I32LE | Number of data | Number of data is stored. | - | - | 0 |  |
|  | startDate | 1 | 1 | H5T_STRING | $\begin{aligned} & \text { Start date of data } \\ & \text { (UTC) } \end{aligned}$ | Start date of data (UTC) is stored. Time format : YYYY-MM-DDThh:mm:ss.ffffffZ <br> (28bytes, the last 1 byte is a null terminated string) | UTC | - | - | There is no dataset if numData is 0 . |
|  | startDate_ContinuousTime | 1 | 1 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Start date of data } \\ & \text { (seconds) } \end{aligned}$ | Total seconds of start date of data since 23:59:59 UTC, Dec 31, 2012. | sec | - | - | There is no dataset if numData is 0 . |
|  | time | 1 | numData | H5T_IEEE_F64LE | Elapse time from <br> start date of data | The elapsed seconds from start date of data. | sec | 10 | - | There is no dataset if numData is 0 . |
|  | sensor Temp | 2 | numData, 10 | H5T_IEEE_F64LE | Sensor temperature | Sensor temperature is stored for each band. | ${ }^{\circ} \mathrm{C}$ | 10 | - | There is no dataset if numData is 0 . |
|  | sensorTempQual ity | 2 | numData, 10 | H5T_STD_I8LE | Quality flag of sensor temperature | $\begin{aligned} & \text { Quality flag of sensor temperature is stored. } \\ & \text { 0: Normal } \\ & 1: \text { Abnormal (outside the acceptable range) } \\ & \text { 2: Quality is unknown due to data loss and so on } \end{aligned}$ | - | - | - | There is no dataset if numData is 0 . |
|  | preAmpTemp | 2 | numData, 10 | H5T_IEEE_F64LE | Pre-amplifier <br> temperature | Pre-amplifier temperature is stored for each band. | ${ }^{\circ} \mathrm{C}$ | 10 | - | There is no dataset if numData is 0 . |
|  | preAmpTempQual ity | 2 | numData, 10 | H5T_STD_I8LE | Quality flag of Pre-amplifier temperature | Quality flag of Pre-amplifier temperature is stored. <br> 0 : Normal <br> 1: Abnormal (outside the acceptable range) <br> 2: Quality is unknown due to data loss and so on | - | - | - | There is no dataset if numData is 0 . |
|  | Amp Temp | 2 | numData, 10 | H5T_IEEE_F64LE | $\begin{array}{\|l} \hline \begin{array}{l} \text { Amplifier } \\ \text { temperature } \end{array} \end{array}$ | Amplifier temperature is stored for each band. | ${ }^{\circ} \mathrm{C}$ | 10 | - | There is no dataset if numData is 0 . |
|  | AmpTempQual ity | 2 | numData, 10 | H5T_STD_I8LE | $\begin{aligned} & \text { Quality flag of } \\ & \text { Amplifier } \\ & \text { temperature } \end{aligned}$ | $\begin{aligned} & \text { Quality flag of Amplifier temperature is stored. } \\ & 0: \text { Normal } \\ & \text { 1: Abnormal (outside the acceptable range) } \\ & \text { 2: Quality is unknown due to data loss and so on } \end{aligned}$ | - | - | - | There is no dataset if numbata is 0 . |
|  | IensM_Temp | 2 | numData, 5 | H5T_IEEE_F64LE | lens (M) temperature | Lens ( $M$ ) temperature is stored for each lens. | ${ }^{\circ} \mathrm{C}$ | 10 | - | There is no dataset if numData is 0 . |
|  | IensM_TempQuality | 2 | numData, 5 | H5T_STD_I8LE | Quality flag of lens (M) temperature | $\begin{aligned} & \text { Qual ity flag of lens }(\mathrm{M}) \text { temperature is stored. } \\ & \text { 0: Normal } \\ & \text { 1: Abnormal (outside the acceptable range) } \\ & \text { 2: Quality is unknown due to data loss and so on } \end{aligned}$ | - | - | - | There is no dataset if numData is 0 . |

Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (10/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (11/12)


Table 6-1 Dataset definition of CAI-2 L1A common file (HDF5) (12/12)


Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (1/7)

| Group Path/Dataset Name | Forward/Backward | Observation/Cal i ration mode | ension] | Data size ${ }_{\text {Size }}$ | Data Type | Dataset Name | Explanation (Format) | Unit | $\begin{array}{\|c\|} \hline{ }^{\text {Significant }} \\ \text { digit } \end{array}$ | $\begin{array}{\|c\|c\|} \hline \text { Invalid } \\ \text { Value } \end{array}$ | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\square$ \|Netadata |  |  |  |  |  |  |  |  |  |  |  |
|  | Forward, Backward | OBS, CAL | 1 | 1 | H5t_String | $\begin{aligned} & \text { File Identifier } \\ & \text { (Granule ID) } \end{aligned}$ | Granule ID (47bytes, the last 1 byte is a null terminated string) <br> - Satellite Name : GOSAT2 (Fixed) <br> - Start Time of Observation (year • month • day • hour • minute) : <br> YYYYMMDDHHmm (UTC) <br> Path No. : PPP (001-089) <br> Processing Level: 1A(Fixed) <br> Band <br> Forward looking band (Band1-5) : F <br> Backward looking band (Band6-10): B <br> Orbing predicted orbit data: P <br> Using GPS or determined orbit data: D <br> Correction coefficients used for processing: <br> Using nominal coefficients: $N$ <br> Using updated coefficients: U <br> Reserved : 00 <br> - Operation Mode : 0000 <br> Observation Mode (day) : OBSM <br> Dark calibration mode : NCAL <br> Electric calibration mode : ECAL <br> Lunar calibration mode: LCAL <br> Al gor ithm Version: AAA (0000999) Parameter Version: BBB (000-999) | - | - | - |  |
| operationlode | Forward, Backward | OBS, CAL | 1 | 1 | H5T_String | $\begin{array}{\|l} \text { Sensor Operation } \\ \text { liode } \end{array}$ |  | - | - | - |  |
| process ingate | Forward, Backward | OBS, CAL | 1 | 1 | H5t_String | Processing date | Date of product creation (UTC) <br> Time format: YYYY-MM-DDThh:mm:ss.ffffffz <br> (28bytes, the last 1 byte is a null terminated string) | UTC | - | - | $\begin{array}{\|l\|} \text { Time when creation job } \\ \text { started is stored. } \end{array}$ |
| startDate | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STRINa | $\begin{aligned} & \text { Start date of CAI-2 } \\ & \text { data } \end{aligned}$ |  | utc | - | - |  |
| endate | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STRING |  |  | UTC | - | - |  |
| geodeticDatum | Forward, Backward | OBS, CAL | 1 | 1 | H5T_String | Geodetic datum | "WGS84/ WGS84": Reference EII ipsoid Model/Frame of Reference Fixed (14bytes, the last 1 byte is a null terminated string) | - | - | - |  |
| satel I iteName | Forward, Backward | OBS, CAL | 1 | 1 | H5t_String | Satell ite Name | "GOSAT-2" . Greenhouse gases Observing SATellite-2 <br> Fixed (8bytes, the last 1 byte is a null terminated string) | - | - | - |  |
| sensorName | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STRING | Sensor Name | "TANSO-CAI-2": Cloud and Aerosol Imager-2 Fixed (12bytes, the last 1 byte is a null terminated string) | - | - | - |  |
| process inglevel | Forward, Backward | OBS, CAL | 1 | 1 | H5T_StRINa | Process ing Level | "L1A": Level 1A (4bytes, the last 1 byte is a null terminated string) | - | - | - |  |
| al gor ithnvers ion | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STRING | Al gori ithm Vers ion | Algorithm vertsion is stored in processing control information (4bytes, the last 1 byte is a null terminated string | - | - | - |  |
| parameterVersion | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STRING | ParaneterVersion | Parameter vertsion is stored in processing control information (4bytes, the last 1 byte is a null terminated string) | - | - | - |  |

Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (2/7)

|  | Group Path/Dataset Name | Forward/Backward | Observation/Cal i ration mode | Dimension | $\frac{\text { Data size }}{\text { Size }}$ | Data Type | Dataset Name | Explanation (Format) | Unit | $\begin{array}{\|c\|} \hline \begin{array}{c} \text { Significant } \\ \text { digit } \end{array} \\ \hline \end{array}$ | $\begin{array}{\|c} \hline \text { Invalid } \\ \text { Value } \end{array}$ | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | granul elicommon | Forward, Backward | OBS, CAL | 1 | 1 | H5t_String | $\begin{array}{\|l\|l\|} \hline \text { Granule ID of } \\ \text { common file } \end{array}$ | Granule ID of common file is stored (47bytes, the last 1 byte is a null terminated string) | - | dist | - |  |
|  | process ingFaci l ity | Forward, Backward | OBS, CAL | 1 | 1 | ht__tring | Process ing facility <br> name |  | - | - | - |  |
| ScanAttri iute (Synchronize with observation point) |  |  |  |  |  |  |  |  |  |  |  |  |
|  | bands_500 | Forward, Backward | OBS, cal | 1 | 1 | H5T_STD_132LE | $\left\lvert\, \begin{aligned} & \text { Number of } 500 \mathrm{~m} \\ & \text { spatial } \\ & \text { bands }\end{aligned}\right.$ | Number of high resolution bands is stored (4bands). | - | - | - |  |
|  | pixels_500 | Forward, Backward | OBS, CAL | 1 | 1 | H5T_ST_I32LE | Number of pixel Is of <br> Som spatiax <br> resolution band | Number of pixels including dark pixels and invalid pixels in a line of high resolution bands (Forward looking band:1-4/Backward looking band:6-9) is stored | - | - | - |  |
|  | 1 ines_500 | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STD_132LE | $\|$Number of lines of <br> 500 m spatial <br> resol lution band | Number of lines in a product of high resolution bands (Forward looking band:1-4/Backward looking band:6-9) is stored. | - | - | - |  |
|  | mi ssingLi ine_500 | Forward, Backward | OBS, CAL | 1 | bands_500 | H5T_ST_I32LE | Number of missing <br> linesof of 500m <br> spatial <br> besond | Number of missing lines in a product of high resolution bands (Forward stored. | - | - | - |  |
|  | bands_1km | Forward, Backward | OBS, CAL | 1 | 1 | H5T_ST_I32LE | band <br> vanber of 1 km <br> sanatial resol ution <br> bands | Number of low resolution bands is stored (lbands). | - | - | - |  |
|  | pixel s_1km | Forward, Backward | OBS, CAL | 1 | 1 | H5T_STD_132LE | Number of pixel $/$ of of <br> Lm spatial <br> resolution band | Number of pixels including dark pixels and invalid pixels in a line of low resolution bands (Forward looking band:5/Backward looking band:10) is stored. | - | - | - |  |
|  | 1 ines _1km | Forward, Backward | OBS, cal | 1 | 1 | H5T_STD_32LE | Number of <br> 1. ines spatial <br> resol of of ion band | Number of lines in a product of low resolution bands (Forward looking band:5/Backward looking band:10) is stored. | - | - | - |  |
|  | missingLines_1km | Forward, Backward | OBS, CAL | 1 | bands_1km | H5T_STD_132LE | Number of missing <br> lines of <br> spatial <br> sent <br> band | Number of missing lines in a product of low resolution bands (Forward looking band:5/Backward looking band:10) is stored. | - | - | - |  |
|  | LineAtribute_500 (Information for each line of band1-4 and 6-9) |  |  |  |  |  |  |  |  |  |  | There is no dataset if I ines_500 is 0 . |
|  | miss ingFlag | Forward, Backward | OBS, cal | 2 | \| ines_500, bands_500 | H5T_STD_18LE | Wi issing line flag | $\begin{aligned} & \text { In Forward/Backword looking band file, status of missing line flag } \\ & \text { for band1-4/6-9 is stored. The status is shown below. } \\ & 0: \text { No missing line } \\ & 1: \text { Missing line (No pixel available in the line) } \\ & 2: \text { Invalid (observed in other observation mode) } \end{aligned}$ | - | - | - | There is no dataset if I ines_ 500 is 0 . |
|  | observationTime | Forward, Backward | OBS, CAL | 2 | \| ines_500, bands_500 | H5t_String | Line observationtime <br> tine center of <br> (texposure <br> (UTC)(UTime) | In Forward/Backword looking band file (band 1-4/6-9), the center of exposure time of each line (UTC) is stored If satTimeStatusFlag is abnormal, time is corrected by the time correction information <br> Time format:YYYY-MM-DDThh:mm:ss. ffffffz (28bytes, the last 1 byte is a null terminated string $\times$ lines_ $500 \times$ bands_500) | utc | - | - | There is no dataset if I ines_500 is 0 . |
|  | observationT ime_Cont inuoust ine | Forward, Backward | OBS, cal | 2 | \| ines_500, bands_500 | H5T_IEEE_F64LE | Line ebservation time tine eenter of (enposure tite) (seconds) | Total seconds of observationTime since 23:59:59 UTC, Dec 31, 2012. | sec | - | - | There is no dataset if I ines_500 is 0 . |

Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (3/7)

| Group Path/Dataset Name | Forward/Backward | Observation/Cal iration mode | Dimension | $\frac{\text { Data size }}{\text { Size }}$ | Data Type | Dataset Name | Explanation (Format) | Unit | ${ }_{\text {digit }}^{\text {Sigificant }}$ digit | $\begin{array}{\|c} \hline \text { Invalid } \\ \text { Value } \end{array}$ | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| sat Time | Forward, Backward | OBS, CAL | 1 | 1 ines_500 | H5T_STD_132LE | Line satel I ite time |  | sec | - | - | There is no dataset if I ines_500 is 0 . |
| satTimeStatusf lag | Forward, Backward | OBS, CAL | 1 | 1 ines_500 | H5T_ST_I8LE | Satellite time system status flag | Status flag of satellite time system is stored. o Normal (The correction of time error is in ot required.) 1 : Abnormal (The correction of time error is required.) | - | - | - | There is no dataset if lines_500 is 0 . |
| observationcounter | Forward, Backward | OBS, CAL | 1 | 1 ines_500 | H5T_STD_I32LE | $\begin{aligned} & \text { Precise observation } \\ & \text { clock (counter) } \end{aligned}$ | Count value of internal clock counter is stored. 1 count= 128 microseconds | - | - | - | There is no dataset if lines_500 is 0 . |
| integrationNum | Forward, Backward | OBS, CAL | 2 | \| ines_500, bands_500 | H5T_STD_132LE | Number of <br> integration index | In Forward(band1-4)/Backword(band 6-9) looking band file, number of integration of each line is stored. | - | - | - | There is no dataset if I ines_ 500 is 0 . |
| integrationtime | Forward, Backward | OBS, cal | 2 | \| ines_500, bands_500 | H5T_IEEE_F64LE | Integration time |  exposure time | sec | 10 | - | There is no dataset if I ines_500 is 0 . |
| LineAttri ibute_1km (Information for each line of band5, 10) |  |  |  |  |  |  |  |  |  |  | $\begin{array}{\|l\|} \text { There is no dataset if } \\ \text { lines_ } 1 \mathrm{~km} \text { is } 0 . \end{array}$ |
| $\mathrm{m}_{\text {issi ingFlag }}$ | Forward, Backward | OBS, CAL | 2 | \| ines_1kn, bands_1km | H5T_STD_18LE | $\left\lvert\, \begin{aligned} & \text { Wis sing line status } \\ & \text { flag } \end{aligned}\right.$ | In Forward/Backword looking band file, status of missing line flag for band1-4/6-9 is stored. The status is shown below. <br> 0 : No missing line <br> 1: Missing line (No pixel available in the line) <br> 2: Invalid (observed in other observation mode) | - | - | - | $\begin{aligned} & \text { There is no dataset if } \\ & 1 \text { ines } 1 \mathrm{~km} \text { is } 0 \text {. } \end{aligned}$ |
| observationtime | Forward, Backward | OBS, CAL | 2 | \| ines_1kn, bands_1km | ht__String | Line observation time time eenter of (texposure time) (UTC) |  | UTC | - | - | $\left\lvert\, \begin{aligned} & \text { There is no dataset if } \\ & \text { lines } 1 \mathrm{~km} \text { is } 0 . \end{aligned}\right.$ |
| observationT ime_Cont inuoust ime | Forward, Backward | OBS, CAL | 2 | \| ines_1kn, bands_1km | H5T_IEEE_F64LE | Line observationtime <br> tine e eenter of <br> (texposuret et ime) <br> (seconds) | Total seconds of observationTi ime since 23:59:59 UTC, Dec 31, 2012. | sec | - | - | There is no dataset if lines 1 km is 0 |
| satTime | Forward, Backward | OBS, CAL | 1 | ${ }^{\prime}$ ines_1km | H5T_STD_I32LE | Line satell ite time |  | sec | - | - | There is no dataset if I ines_lkm is 0 . |
| satTi mestatusFl lag | Forward, Backward | OBS, CAL | 1 | ${ }^{\prime}$ ines_1km | H5__STD_8LE | $\begin{aligned} & \text { Satellite time } \\ & \text { status flag } \end{aligned}$ | Status flag of satellite time status is stored. o: Normal (The correction of time error is not required.) 1: Abnormal (The correction of time error is required.) | - | - | - | There is no dataset if I ines_1km is 0 . |

Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (4/7)


Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (5/7)


Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (6/7)

| Group Path/Dataset Name |  | Forward/Backward | Observation/Cal iration mode | Dimension | $\frac{\text { Data size }}{\text { Size }}$ | Data Type | Dataset Name | Explanation (Format) | Unit | $\begin{gathered} \underset{\text { dignificant }}{\text { digit }} \\ \hline \end{gathered}$ | (Invalid <br> Value | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
|  | satVel_ECI | Forward, Backward | OBS, CAL | 2 | subsetNumLines, 3 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Satellite velocity } \\ & \left(\begin{array}{l} \text { ECI }(\text { (TOO }) \end{array}\right. \end{aligned}$ | Satell ite velocity in ECC( (ToD) for each sample and line exposure time of standard band image is stored. | km/s | 10 | - | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumLines is } 0 \text {. } \end{aligned}$ |
|  | satArgLat | Forward, Backward | OBS, CAL | 1 | subsetNumLines | H5T_IEEE_F64LE | $\begin{aligned} & \text { Argument of } \\ & \text { latitude } \end{aligned}$ | Argument of latitude of observation time is stored $0 \leqq$ satArgLat<360 | deg | 10 | - | There is no dataset if subsetNumLines is 0 . |
|  | satOrbitPrecision | Forward, Backward | OBS, CAL | 1 | subsetNumLines | hti_string | $\left\lvert\, \begin{aligned} & \text { Precision of } \\ & \text { satellite orbit }\end{aligned}\right.$ |  | - | - | - | There is no dataset if subsetNumLines is 0 . |
|  | satAtt | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 4 | H5T_IEEE_F64LE |  |  | - | 10 | - | There is no dataset if subsetNumL ines is 0. |
|  | satRPY | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE | Satellite Attitude Roll/Pitch/Yaw | ```Satellite attitude for each sample and line exposure time is stored as roll, pitch, and yaw angles atRPY[subsetNumLines] [b] \(b=0\) : roll \(b=2\) : yaw``` | deg | 10 | - | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumLines is } 0 \text {. } \end{aligned}$ |
|  | yauSteer ingFlag | Forward, Backward | OBS, CAL | 1 | subsetNumLines | H5T_STD_18LE | Yaw steer ing flag | Yaw steer ing flag indicates the operation of yaw steer ing. yawSteer ingF lag[subsetNumLines] o: Not execute (OFF) 1: Execute (ONN 2: Inval lid due to data loss and so on | - | - | 2 | $\left.\begin{array}{\|l\|} \text { There is no dataset if } \\ \text { subsentumbines is } 0 . \end{array} \right\rvert\,$ |
|  | satAttInterpol at ionMethodF lag | Forward, Backward | OBS, CAL | 1 | subsetNumL ines | H5T_STD_18LE | $\begin{aligned} & \text { Satel lite attitude } \\ & \text { interoolation } \\ & \text { method flag } \end{aligned}$ | Interpolation method for calculating satellite attitude is stored. <br> : Interpolation <br> 1: Extrapolation | - | - | - | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumLines is } 0 \text {. } \end{aligned}$ |
|  | satAttInterpol ationoual i tyF lag | Forward, Backward | OBS, CAL | 1 | subsetNumL ines | H5T_STD_18LE | $\begin{aligned} & \text { Satell lite atti itude } \\ & \text { Interpolation } \\ & \text { aual ity flag } \end{aligned}$ |  | - | - | - | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumLines is } 0 \text {. } \end{aligned}$ |
|  | satToECR_Matr ix | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 9 | H5T_IEEE_F64LE | vCoordinate transformation matrix from satelinte-fixed to ECR (WGS84) | Coordinate transformation matrix convert from satellite-fixed oordinate to ECR(WGS84) are stored in the following order: <br> (0, 1, 2) <br> $(3,4,5)$ <br> $(6,7,8)$ | - | 10 | - | There is no dataset if <br> subsetNumL ines is 0. |
|  |  |  |  |  |  |  |  |  |  |  |  |  |
|  | solarPos_ECR | Forward, Backward | OBS, CAL | 2 | subsetNumLines, 3 | H5T_IEEE_F64LE | $\left\lvert\, \begin{aligned} & \text { Apparent solar } \\ & \text { position at the } \\ & \text { line exposure point } \\ & \text { lin }\end{aligned}\right.$ <br> (ECR (WGS84)) | Apparent solar position in ECR(IGS884) for each sample line is stored. | km | 10 | (0, 0, 0) | There is no dataset if subsetNumLines is 0. |
|  | solarvel_ECR | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Apparent solar } \\ & \text { Aelocity at tre } \\ & \text { line exposur } \\ & \text { point (CCR (IGS844)) } \end{aligned}$ | Apparent solar velocity in ECR(MGS844) for each sample line is stored. | km/s | 10 | (0, 0, 0) | There is no dataset if subsetNumLines is 0 . |
|  | solarPos_ECI | Forward, Backward | OBS, CAL | 2 | subsetNumLines, 3 | H5T_IEEE_F64LE | Apparent solar position at the line exposure point (ECI (TOD) | Apparent solar position in ECI (TOD) for each sample line is stored. | km | 10 | (0, 0, 0) | There is no dataset if <br> subsetNumLines is 0 . |
|  | solarvel I_EI | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE | $\begin{aligned} & \text { Apparent solar } \\ & \text { velocitity at the } \\ & \text { line exposure point } \\ & \text { (ECI (TOD)) } \end{aligned}$ | Apparent solar velocity in ECI (TOO) for each sample line is stored. | kn/s | 10 | (0, 0, 0) | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumLines is } 0 \text {. } \end{aligned}$ |

Table 6-2 Dataset definition of CAI-2 L1A forward/backward looking band file (HDF5) (7/7)

| Group Path/Dataset Name |  | Forward/Backward | Observation/Cal i ration mode | ensi | $\frac{\text { Data size }}{\text { Size }}$ | Data Type | Dataset Name | Explanation (Format) | Unit | $\xlongequal[\substack{\text { Signif icant } \\ \text { digit }}]{ }$ | $\begin{array}{\|l\|l\|} \hline \text { Invalid } \\ \text { Value } \end{array}$ | Notes |
| :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: | :---: |
| $\square$ Lunargeometry |  |  |  |  |  |  |  |  |  |  |  |  |
|  | IunarPos_ECR | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE |  | True lunar position in ECR(VGS84) for each sample line is stored. | km | 10 | (0, 0, 0) | There is no dataset if subsetNumLines is 0. |
|  | Iunarvel_ECR | Forvard, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE | True lunar velocity at the line exposure point (ECR (WGS84)) | True lunar velocity in ECR(VGS84) for each sample line is stored. | km/s | 10 | (0, 0, 0) | $\begin{aligned} & \text { There is no dataset if } \\ & \text { subsetNumL ines is } 0 \text {. } \end{aligned}$ |
|  | IunarPos_ECI | Forward, Backward | OBS, CAL | 2 | subsetNumL ines, 3 | H5T_IEEE_F64LE |  | True lunar position in ECI (TOD) for each sample line is stored. | km | 10 | (0, 0, 0) | There is no dataset if subsetNumLines is 0. |
|  | Iunarvel_ECI | Forward, Backward | OBS, CAL | 2 | subsetNumLines, 3 | H5T_IEEE_F64LE | True lunar velocity <br> at the line <br> atposure <br> point (EC ( $T 00$ ) $)$ | True lunar velocity in ECI (TOO) for each sample line is stored. | km/s | 10 | (0, 0, 0) | $\begin{array}{\|l\|} \hline \text { There is no datasest if } \\ \text { subsetNumLines is } 0 \text {. } \end{array}$ |

