



A survey of in-flight radiometric calibration methods and their applicability to Nanosatellites

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Outlines :

Presentation of the companies

Characteristics of some EO VHR sensors (LargeSat & CubeSat)

The radiometric model

Characterization of :

- dark signal
- inter-detector gain
- instrument noise
- absolute radiometric gain

Some clues for Radiometric Calibration of CubeSats

VH RODA Workshop 2019, Frascati

Partnership: Image quality & Calibration



Agenium Space is part of Agenium Group, a French SME created in 2003



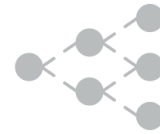
8.5 M Euros Turnover
FY2019

AGENIUM Space Core Business: **Expertise in EO domain**



Image Processing

- Geometry, radiometry, atmospheric corrections, straylight, image quality simulation, information extraction
- Ground Processing Development



Deep learning

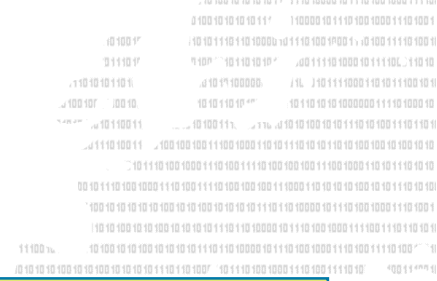
- General-purpose methodology applied to EO issues
- Ad-hoc comprehensive solutions tailored for specific user needs



Flight dynamics

- Extrapolation and orbit restitution, positioning and station-keeping, SCAO
- Expertise shared with AGENIUM group

Founded in 2014, Geo4i is a French SME specialized in the processing and operating of geospatial information in the fields of Defense, Security and Safety.



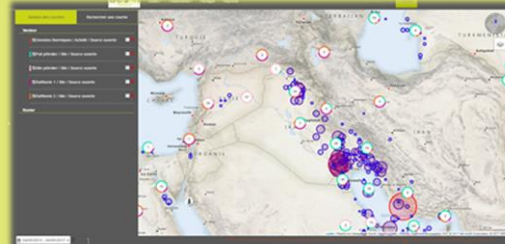
SPACE IMAGERY AND VIDEO PROVIDER

- Optical Multispectral / Radar
- Video
- 3D products
- Image Quality assessment



SOFTWARE DEVELOPMENT

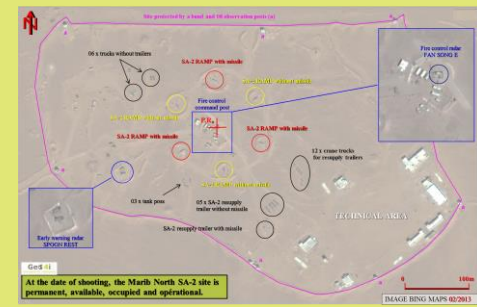
- Images exploitation and analysis
- Automated processing
- Target Identification
- Dedicated tools



GeoSpace Platform

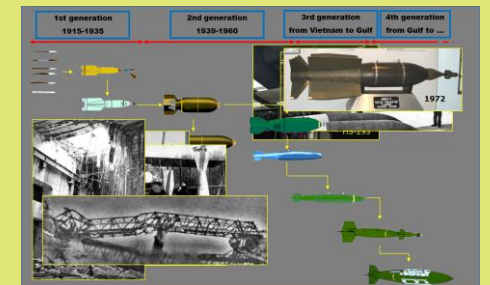
PRODUCTION

- Space Image analysis
- Multi-Source geolocated data analysis
- Value added product generation from radar and multispectral imagery



TRAINING & CONSULTING

- Data Management and open source
- Remote sensing knowledge for Geo-Intelligence
- IMINT Fundamentals
- Military equipment identification



AG Space & Geo4i: User Ground Segment Definition & Development

Image Quality Expertise

Long experience in EO mission

Involved in all the mission steps:

- Sensor and IQ requirements,
- User ground segment definition (ATBDs for L0, L1a,b,c)
- Preparation and participation to in flight commissioning (Calibration, Validation, optimization of algorithms)
- Mission performance control

Software Development & Optimization

- Experienced development engineers
- Operational image processing software development
- SW products for vectorization/portability

Deep Learning Use & Expertise

Software development for Image analysis and target identification

Expertise in:

- On board processing,
- Object detection and identification,
- Land cover/land use
- Change and anomalies detection,
- 3D reconstruction
- ...

Team experience in Image Quality & Ground Segment

Activities	IQ Expertise	Software Architecture	Software Development	Project Management
Sentinel 2 Pre-Studies and ATBDs authoring (CNES)	X		X ²	
S2 GPP and IPF (ESTEC and ESRIN)	X	X	X	X
S2 In Flight Commissioning (incl. preparation) (ESTEC and CNES)	X		X ³	X ¹
Venüs Pre-Studies and ATBDs authoring (CNES)	X		X ²	
Venüs In Flight Commissioning (incl. preparation) (CNES)	X		X ³	X ¹
MicroCarb Pre-Studies and ATBDs authoring (L1) (CNES)	X		X ²	
3MI GPP and IDS (ESTEC and LEONARDO)	X	X	X	X
3MI Ground Calibration (ESTEC and CSL)	X		X	
FLEX L2 Studies (ESTEC)	X	X		X

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1 : Management of the team supporting Customer in the flight commissioning

2 : Development of prototypes useful for the studies

3 : Development of IQ and calibration tools useful for commissioning phase

Other Missions: Spot5 / SCARAB (Meghatropiques) / IIR (Calipso) / Taranis / Merlin / ...

End customers: CNES, ESA, EUMETSAT

Small satellites

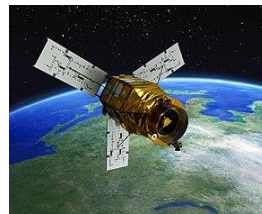
Natural evolution of our core activities

- Image processing : Missions & Ground segments
 - Support to mission specification, image quality & performance assessment
 - Ground segments specifications
 - Ad-hoc Ground Segment development
 - Training for civil and military missions

- Deep learning: On-board processing
 - Objectives:
 - Data reduction -> bottleneck for small satellites
 - Data analysis for intelligent payloads: agility inter-missions, downlink selection, rapid information transmission ...
 - Collisions detection/prediction
 - Studies on advanced DNN simplification methods (CNES, ESA)
 - Commercial products and services

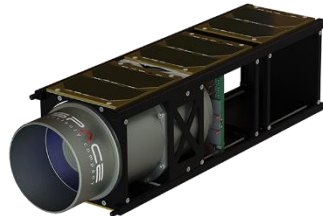
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VHR on LargeSats and CubeSats



LargeSats

Mission	Venus	Pleiades HR	KompSat 3	WorldView 3
Resolution	5.3 m	0.7m (PAN)	0,7m (PAN)	0.31m (PAN)
Swath	27.5 km	20 km	15 km	13.1 km
Spectral bands	12 MS	PAN+4MS	PAN+4MS	PAN+8VNIR+8SWIR
Mass	260kg	940kg	980kg	2800kg



CubeSats

Mission/sensor	DOVE (Planet)	Chameleon (SAC)	KubeKam (SATlantis)
Resolution	3.7m (@ 475km)	6m (@500km)	2m (@500km)
Swath / Frame	24.6*16.4 km ²	49km	13,2 km
Spectral bands	RGB + NIR	PAN+4MS	PAN+4MS
Platform size / status	3U / 120+ on orbit	3U/ On the shelves	12U / On development

Very High Resolution missions are multi-spectral (MS) missions often based on **line-scan technology, with TDI**. Panchromatic (PAN) band is often more resolved than MS bands.

The Radiometric model

Optical model : From radiance outside the instrument to photon flux observed by one-pixel detector

$$\phi_{(photons/m^2s)} = Sa_{(sr)} \cdot \int_{\lambda} SRF(\lambda) \cdot K^{-1}_{(J/photon)}(\lambda) \cdot R_{(W/m^2.sr.\mu m)}(\lambda) \cdot d\lambda + Noise$$

SRF : Spectral Response Function

Sa : Solid angle of the detector

K : Energy of one Photon at a given wavelength

R : Radiance as input of instrument

Detector model : from photon flux to output digital count.

$$X_{(DC)} = Int [((\phi_{(photons/m^2.s)} \cdot A_{(m^2)} \cdot Qe_{(e-/photon)} + Dark_{(e-/s)}) \cdot Ti_{(s)} \cdot G_{(V/e-)} + Offset_{(V)}) \cdot Qf_{(DC/V)} + Noise]$$

A : Area of the detector

G : electronic Gain

Qe : Quantum efficiency

Offset : electronic Offset

Dark : Dark current

Qf : Quantization factor

Ti : integration Time

Simplified model for linescan multispectral sensor :

$$X_{i,j} = (A_{k,tdi} \cdot g_{j,tdi} \cdot R_{i,j} + Dark_{j,tdi})$$

average_j (g_j)=1 for R_{ref} g_j : linear, polynomial, piece-wise linear

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Estimation of Dark signal

Long acquisition (to reduce instrument noise) :

- over Ocean by night (without Moonlight).
- with Telescope shutter closed

Dark dependence on temperature :

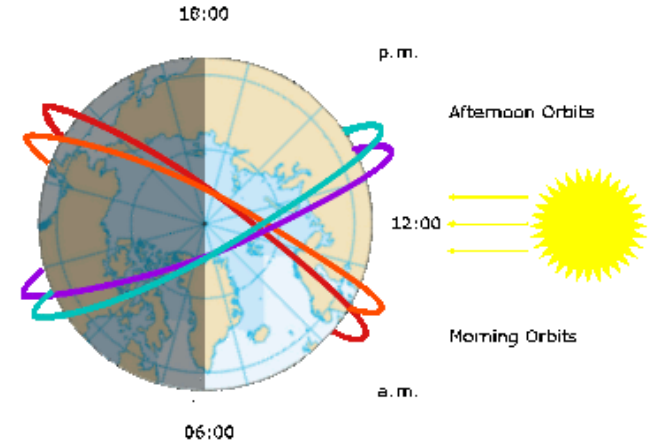
- Not an issue when focal plane is thermally regulated.
- Else, need to know dark currents function of temperature (\Rightarrow cubesats)

$\Rightarrow [T_{min}, T_{max}]_{science} \subset [T_{min}, T_{max}]_{calibration} ?$

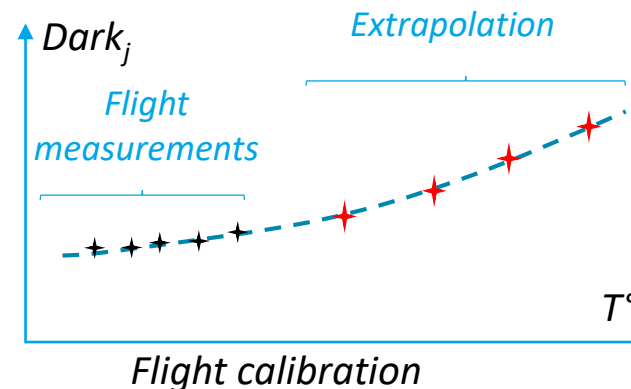
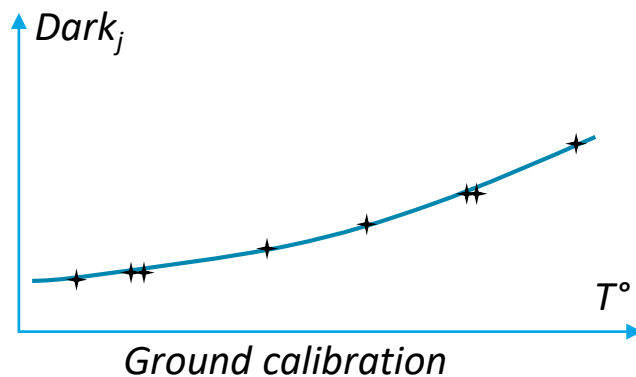
Not obvious when calibration acquired on eclipse side and science acquired on sun side of the orbit

One solution :

- On ground calibration : define the shape of thermal laws.
- In flight calibration : using the ground law to extrapolate new values at higher temperatures.



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Estimation of g_j (flat-fied)

Acquisitions with uniform targets covering the whole swath:

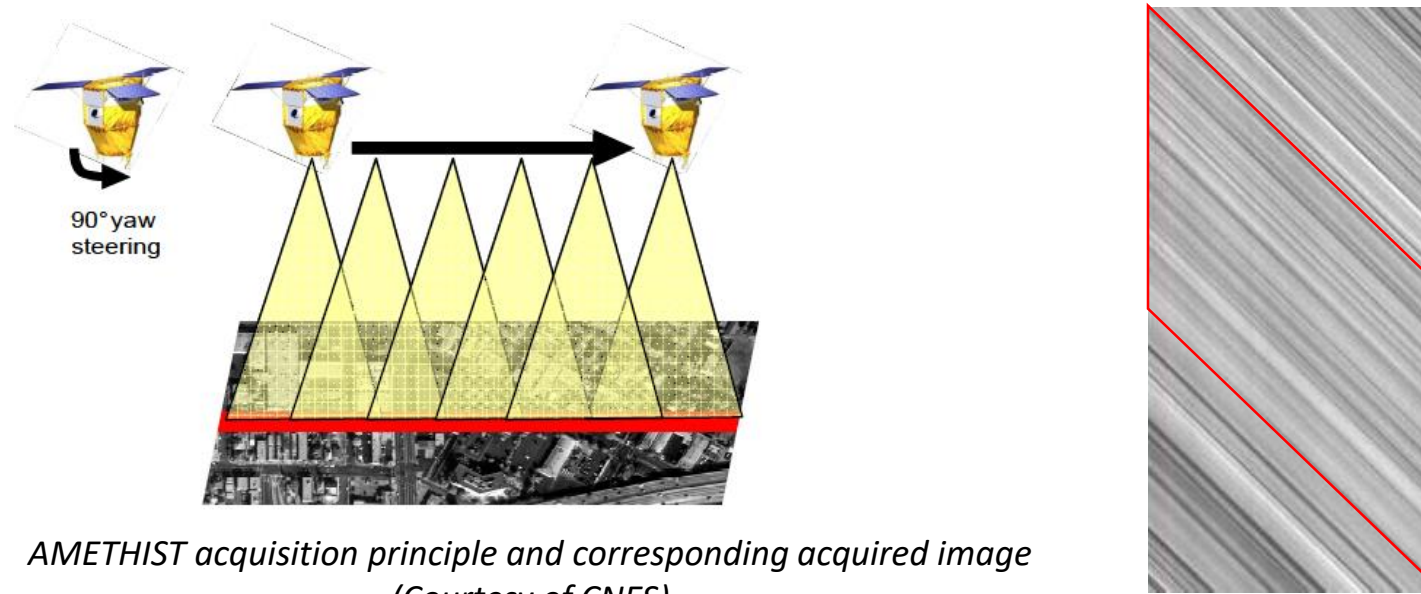
1/ Deployable sun diffuser (Sentinel 2) => Complex to develop on a cubeSat.

2/ AMETHIST : On any landscape with 90° yaw steering (Pleiades HR)

⇒ Need an accurate steering : scan-line parallel to the ground velocity => Complex for a CubeSat

⇒ Each pixel of the scan-line is acquiring the same landscape.

⇒ A histogram matching method allows to reach the inter-detector gains (including non-linearity)



*AMETHIST acquisition principle and corresponding acquired image
(Courtesy of CNES)*

Estimation of g_j (flat field)

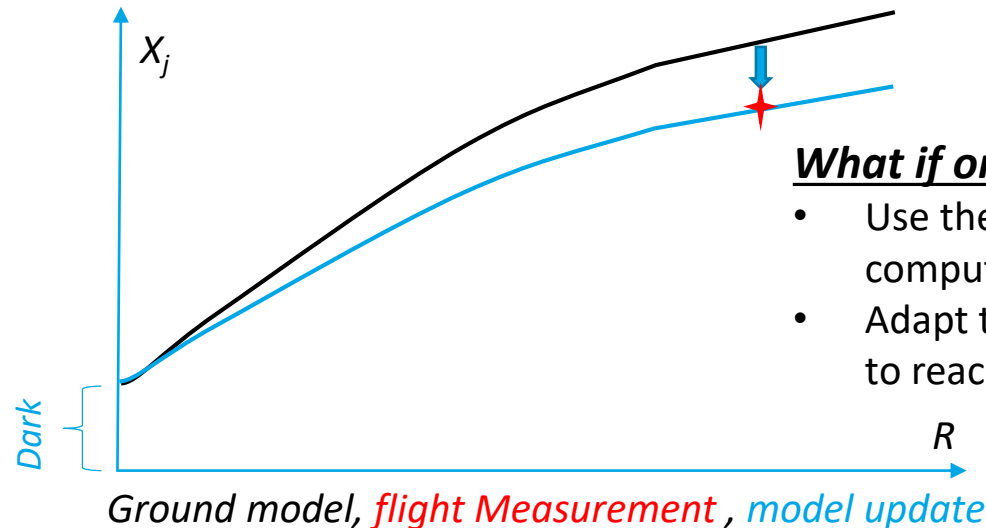
3/ Classical method : Uniform natural landscapes (0° yaw steering)

(Spot5, Venµs)

- ⇒ Identify the areas where landscape is uniform.
- ⇒ Correct from Dark and sun angles.
- ⇒ Compute mean-line to reduce impact of instrument and landscape noises
- ⇒ **Method usable for CubeSats : compatible with small swath and bad agility**

4/ Uniform natural landscapes +90° yaw steering

- ⇒ Limits the impact of landscape non uniformity across track.
- ⇒ Compared to AMETHIST :
 - ⇒ Allows less accuracy in yaw steering (landscape is quasi uniform)
 - ⇒ no direct access to non-linearity.

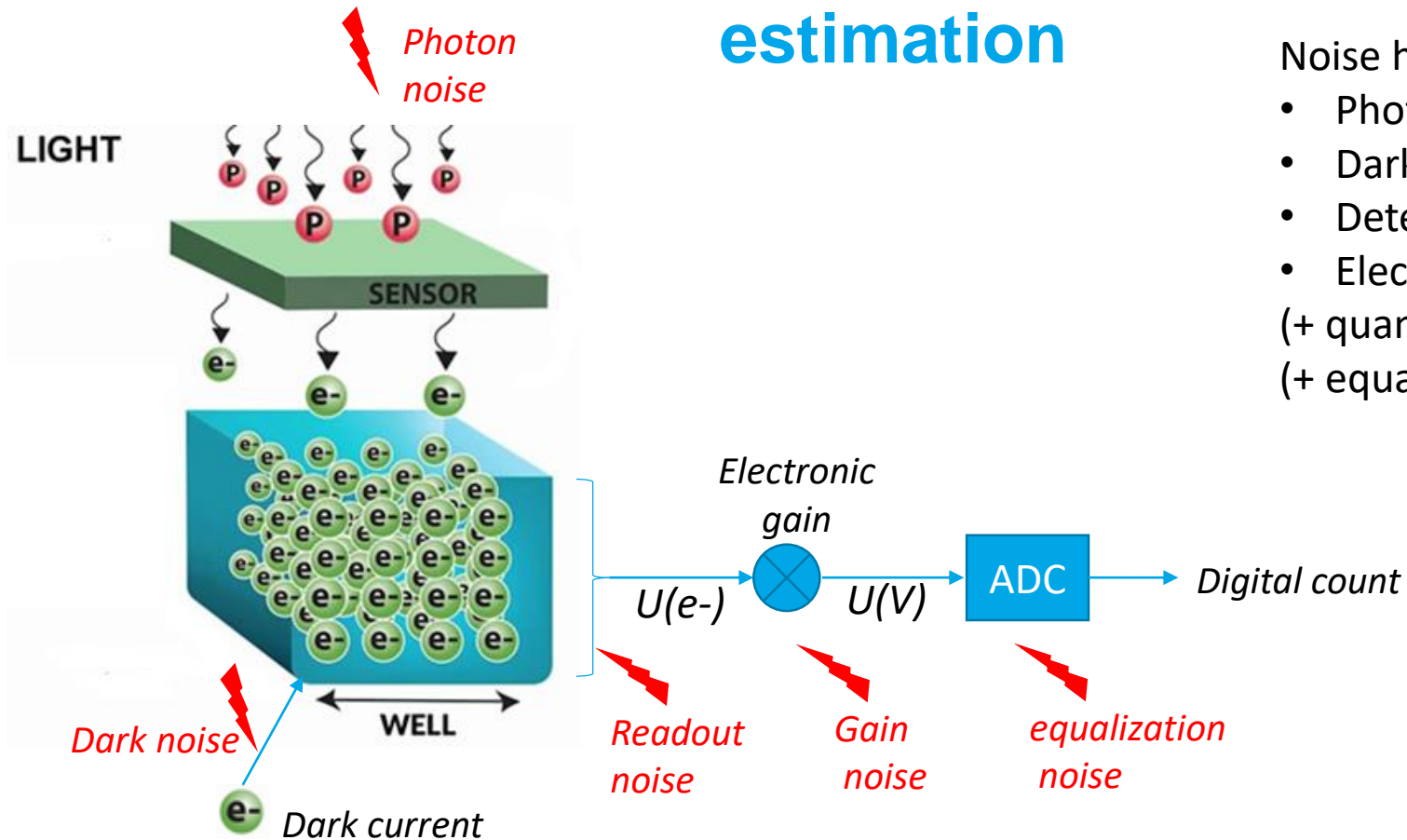


What if only one radiance level available ?

- Use the non-linearity shape computed during ground calibration.
- Adapt the shape with a **scale factor** to reach right value for the measurement.

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Noise estimation



Noise has different contributors:

- Photon noise
 - Dark noise
 - Detector readout noise
 - Electronic Gain noise
- (+ quantization noise)
(+ equalization residual noise)

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Some contributors are depending only on incoming flux, some are depending on digital count.

Classical formula : $\sigma^2 = A + B.R$ (R : Observed radiance)

Noise estimation

$$\sigma^2 = A + B.R$$
$$SNR = R / \sqrt{A + B.R}$$

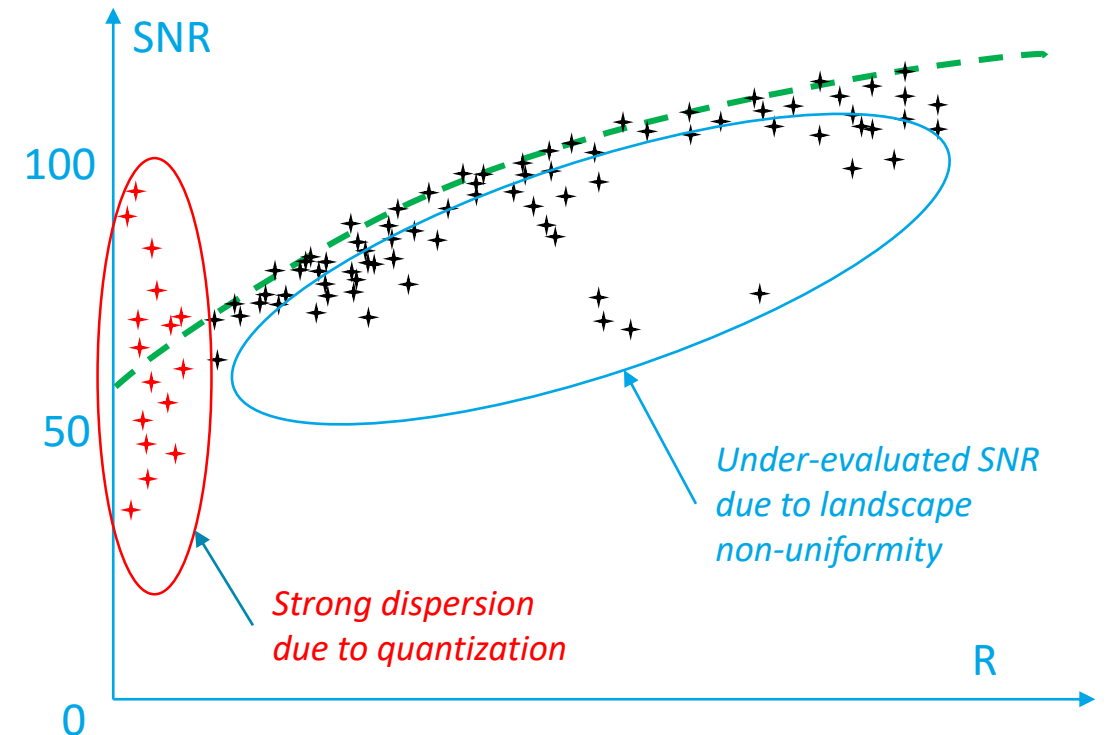
Characterization of A and B : use of uniform areas at different radiances.

- Uniform areas from diffuser acquisitions, or from natural landscapes
- Compute statistics **only along the columns** (to avoid perturbations due to equalization residues)
- Compute **HF standard deviation** (to avoid perturbations due to LF landscape variations)

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*Selection of quasi-uniform areas
on Snow landscape acquisition over Antarctica*



SNR measurements, and SNR law function of radiance

Estimation of Ak (absolute radiometric calibration)

Acquisitions on reference targets:

1. Sun diffuser or other internal references
 - Possible ageing issues
 - **Complex for CubeSats**
2. PICS (Sahara and Arabia sites)
 - BRDF + Atmosphere information available
 - Need to process its own radiative transfer
 - Targets are large scale sites (100*100km²)
3. Using RadCalNet
 - Easy to use, good accuracy
 - Only quasi Nadir observations on small targets.
4. On Moon
 - No atmosphere, nor ageing issues
 - ROLO/POLO models
 - **Need strong de-pointing capabilities (CubeSats ?)**
5. Simultaneous Nadir Overpasses (SNO) with other sensors :
 - Comparison between sensors
 - Useful for a constellation of same satellites (different orbits)
 - Use of LargeSat images as “Gold Reference”



Sentinel2 sun diffuser (ESA, Sener, CSL)

Other methods :

- Rayleigh over ocean (blue bands)
- Deep Convective Clouds, Sunglint (interband calibration)

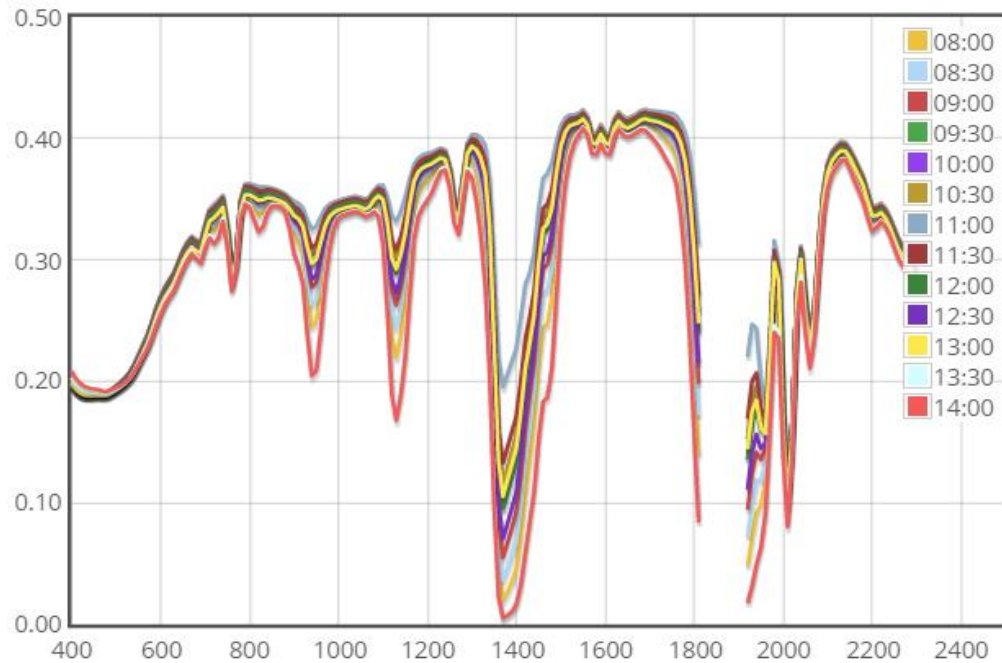


Moon Seen by Venus (Courtesy CNES)

Estimation of Ak : RadCalNet

RadCalNet provides :

- TOA reflectance spectra of 4 sites + associated uncertainties
- Corresponding to nadir observations
- Every 30min (9H-15H local time) when not cloudy
- Representative of ground surface of min 50*50m²



Gobabeb TOA reflectance spectra on 30th of September 2019



Ready to use :

- Integrate RADCALNET spectrum on Instrument SRF
- Compare with instrument observations

Constraint :

- Works for quasi Nadir observations
=> Constraints on orbits

Some clues for CubeSat radiometric calibration

Cost of solutions is a strong argument :

For sensor, platform, launch...

... even for calibration, and User Ground Segment (costs can be factorized in case of large constellation)

Calibration methods :

- A_k : RadCalNet if reachable by the mission orbit.
- g_j : classical method, Uniform areas reachable due to the small swath of the sensor.

Some specific issues to be tackled :

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- Low SNR (even if compensation using TDI) : Importance to assess noise level :
Noise computed on uniform landscape => method to discriminate landscape vs instrument noise.
- Stability of the sensor vs temperature : to take into account in the radiometric model
Importance to have a good ground calibration on a large set of temperatures.



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THANK YOU FOR YOUR ATTENTION