

Spaceborne SAR Systems for Polarimetric, Interferometric and PolinSAR Applications

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Postfach 1116

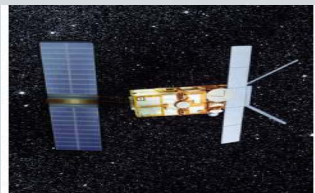
82230 Weßling, Germany

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SEASAT

NASA/JPL (USA)
L-Band, 1978



ERS-1

European Space Agency (ESA)
C-Band, 1991-2000



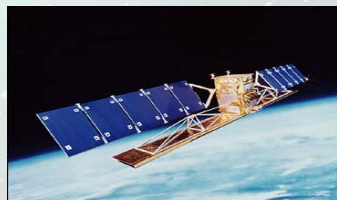
J-ERS-1

Japanese Space Agency (NASDA)
L-Band, 1992-1998



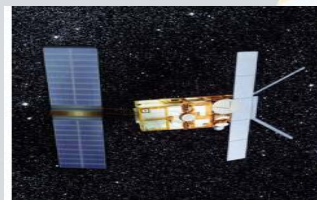
SIR-C/X-SAR

NASA/JPL, L- and C-Band (quad)
DLR / ASI, X-band
April and October 1994



RADARSAT-1

Canadian Space Agency (CSA)
C-Band, 1995-today



ERS-2

European Space Agency (ESA)
C-Band, 1995-today



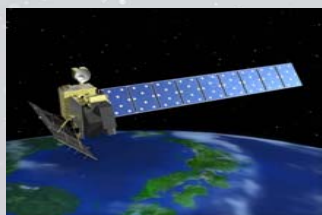
SRTM

NASA/JPL (C-Band), DLR (X-Band)
February 2000



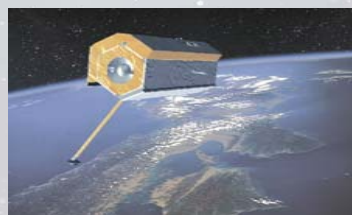
ENVISAT / ASAR

European Space Agency (ESA)
C-Band (dual), 2002-today



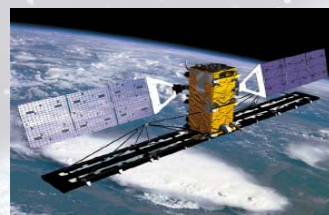
ALOS / PALSAR

Japanese Space Agency (JAXA)
L-Band (quad), 2005



TerraSAR-X

German Aerospace Center (DLR) / Astrium
X-Band (quad), 2006



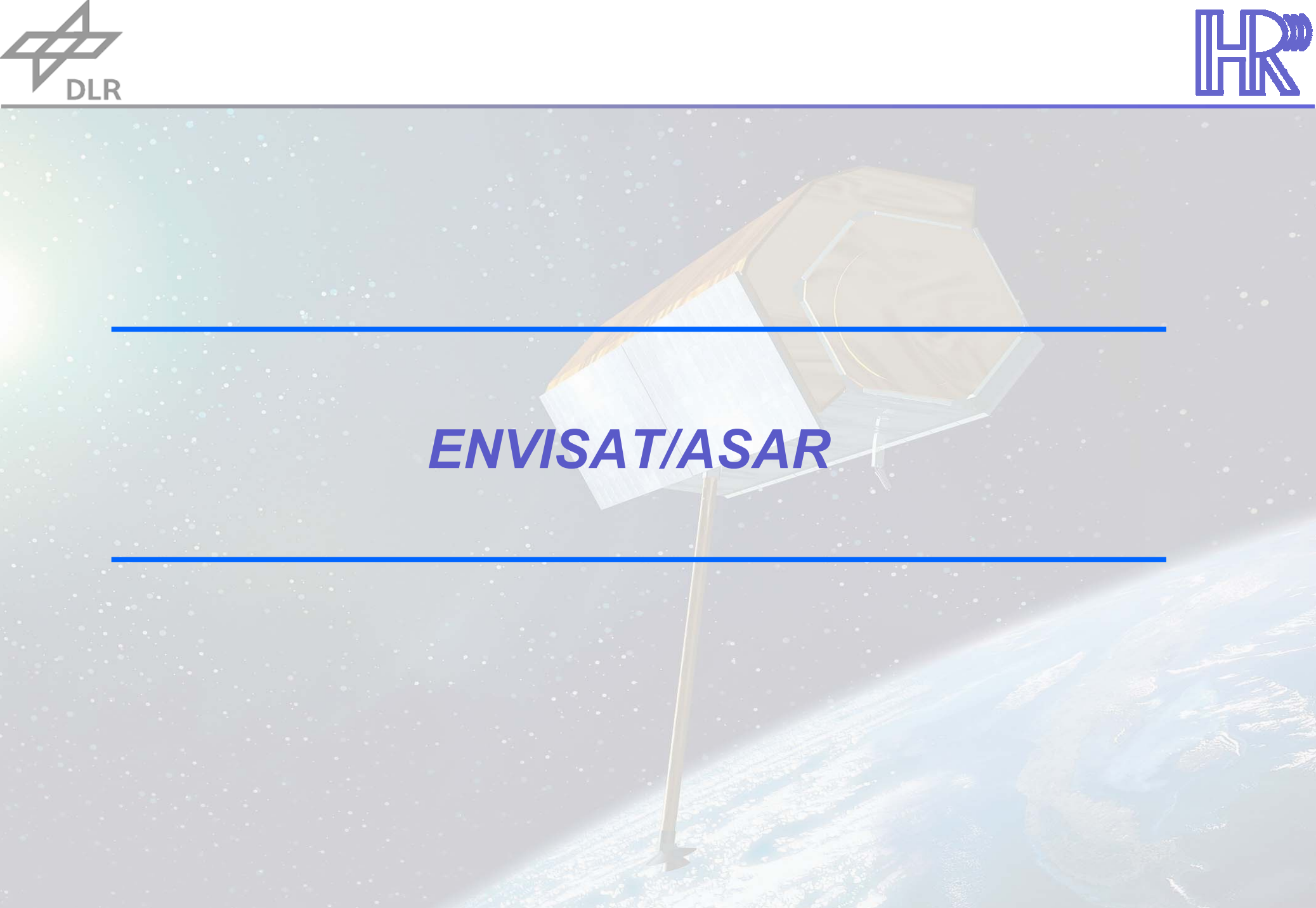
RADARSAT-II

Canadian Space Agency (CSA)
C-Band (quad), 2006



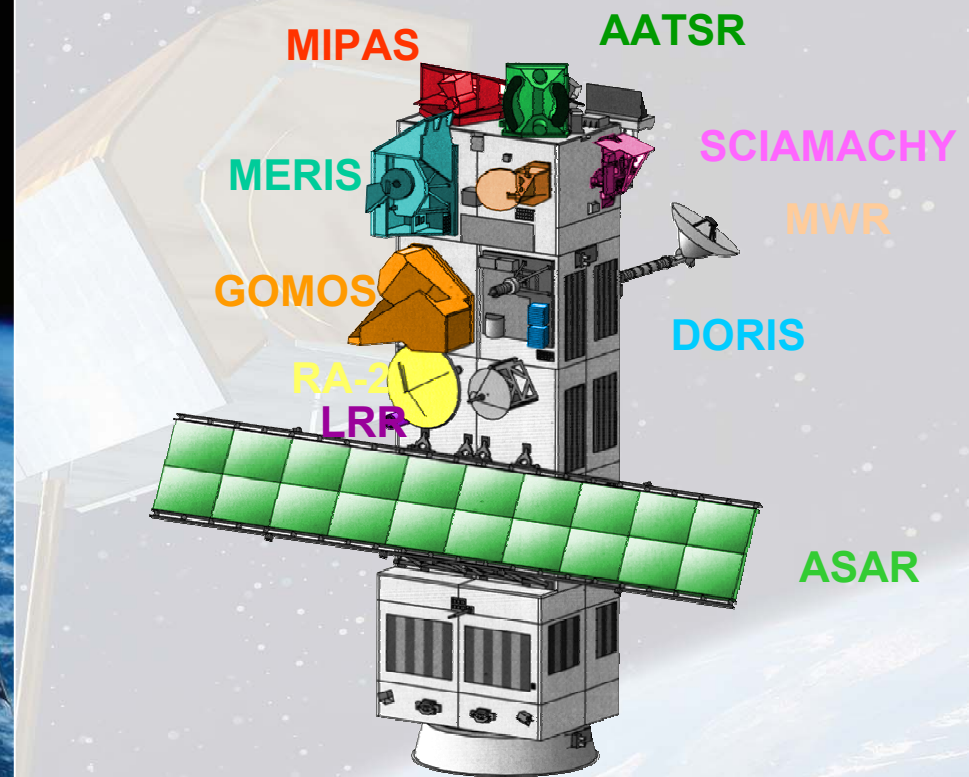
CosmoSkymed

ASI / Alenia
X-Band (dual), 2006

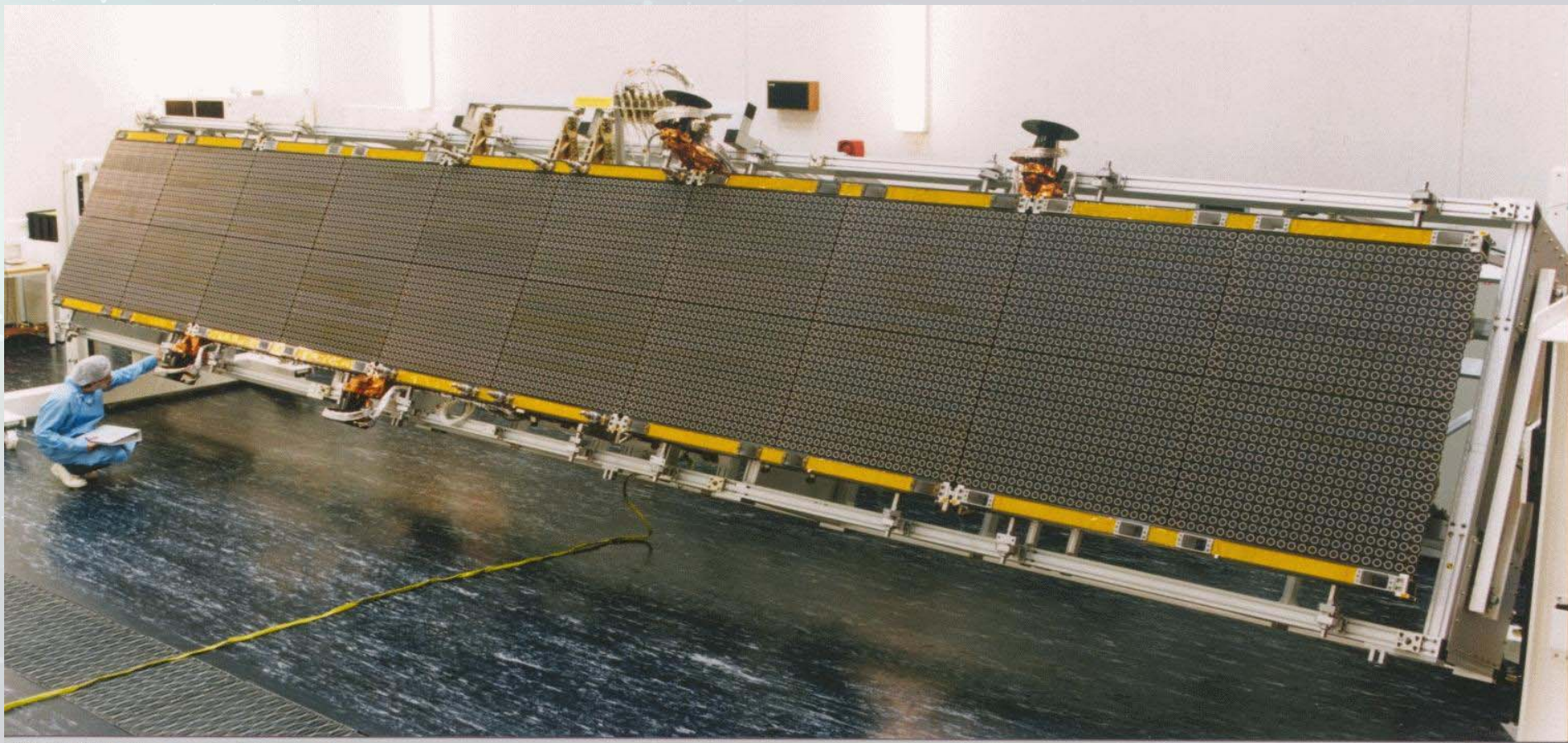


The background of the slide features a 3D rendering of the ENVISAT satellite in orbit. The satellite is a large, complex structure with a prominent white rectangular panel and a large, gold-colored, multi-faceted antenna. It is positioned against a backdrop of a starry space and the Earth's horizon, which shows blue oceans and brown landmasses. Two horizontal blue lines frame the central text.

ENVISAT/ASAR

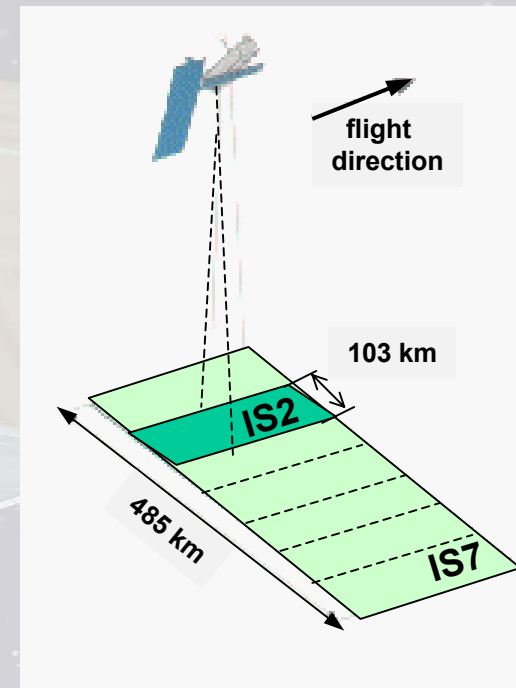
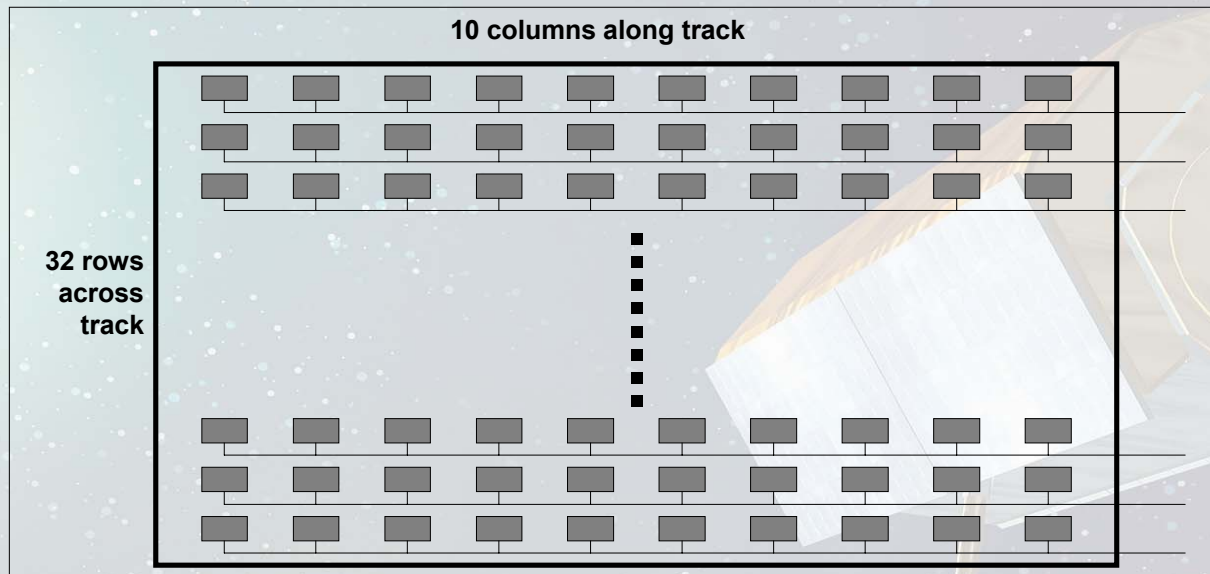


- Most ambitious Remote Sensing satellite for environmental monitoring
- Unique combination of 10 instruments



ASAR Antenna consisting of 320 Transmit/Receive Modules in 32 rows and 10 elements per row

ASAR-Antenna (ca. 10 m x 1.33 m and 320 T/R-Modules)



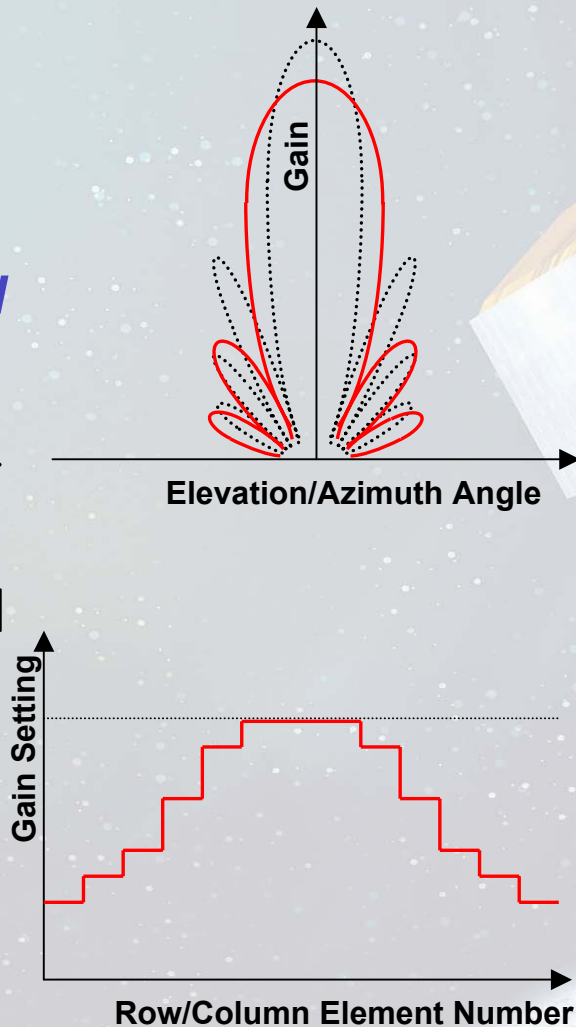
ASAR imaging modes

	IS 1	IS 2	IS 3	IS 4	IS 5	IS 6	IS 7
PRF [Hz]	1678	1645	2096	1680	2082	1698	2097
ASAR center look angle	16.6°	20.1°	25.2°	29.5°	32.8°	35.6°	38.0°
ASAR swath width in ground range [km]	103.7	103.3	82.3	88.6	64.5	71.0	56.7

**Widened
Main Lobe**

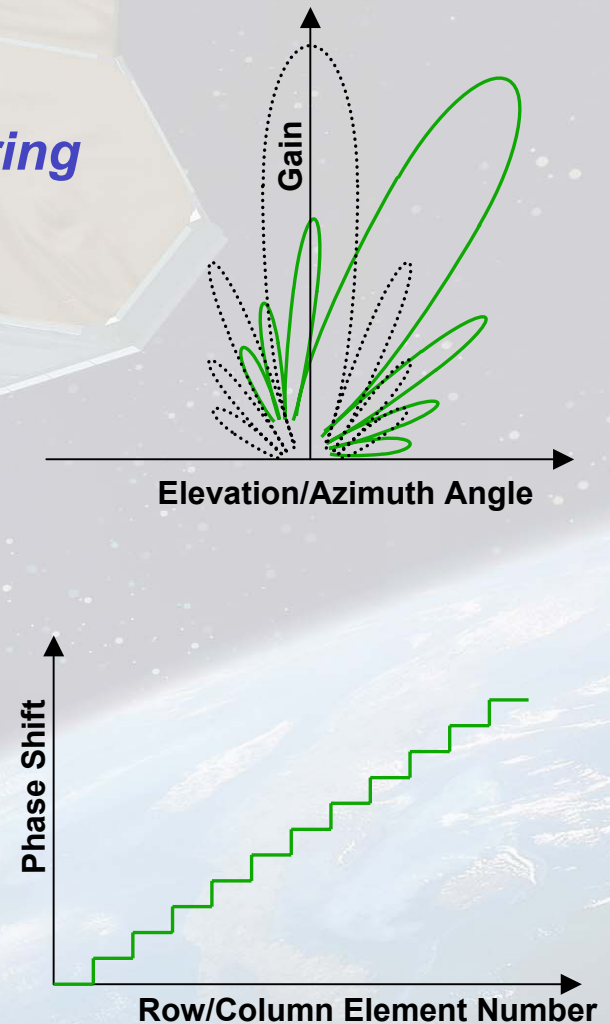
**Suppressed
Sidelobes**

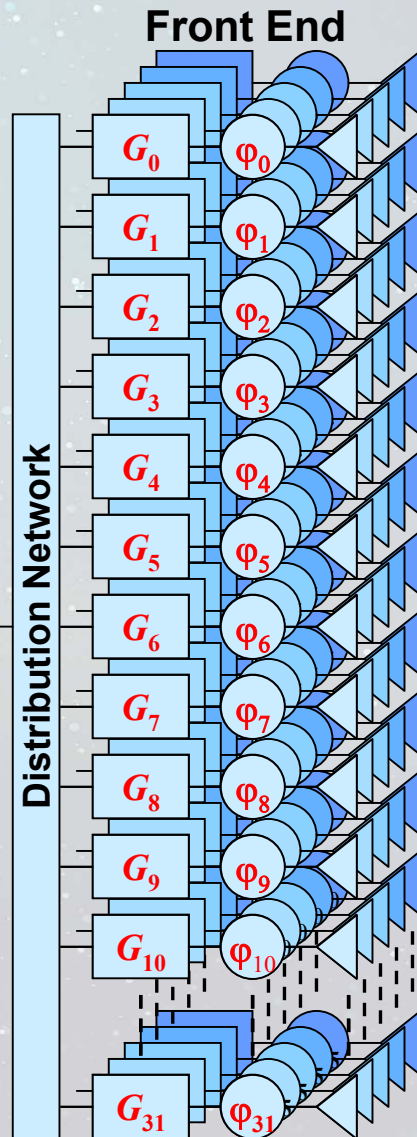
**Amplitude
Tapering**



Beam Steering

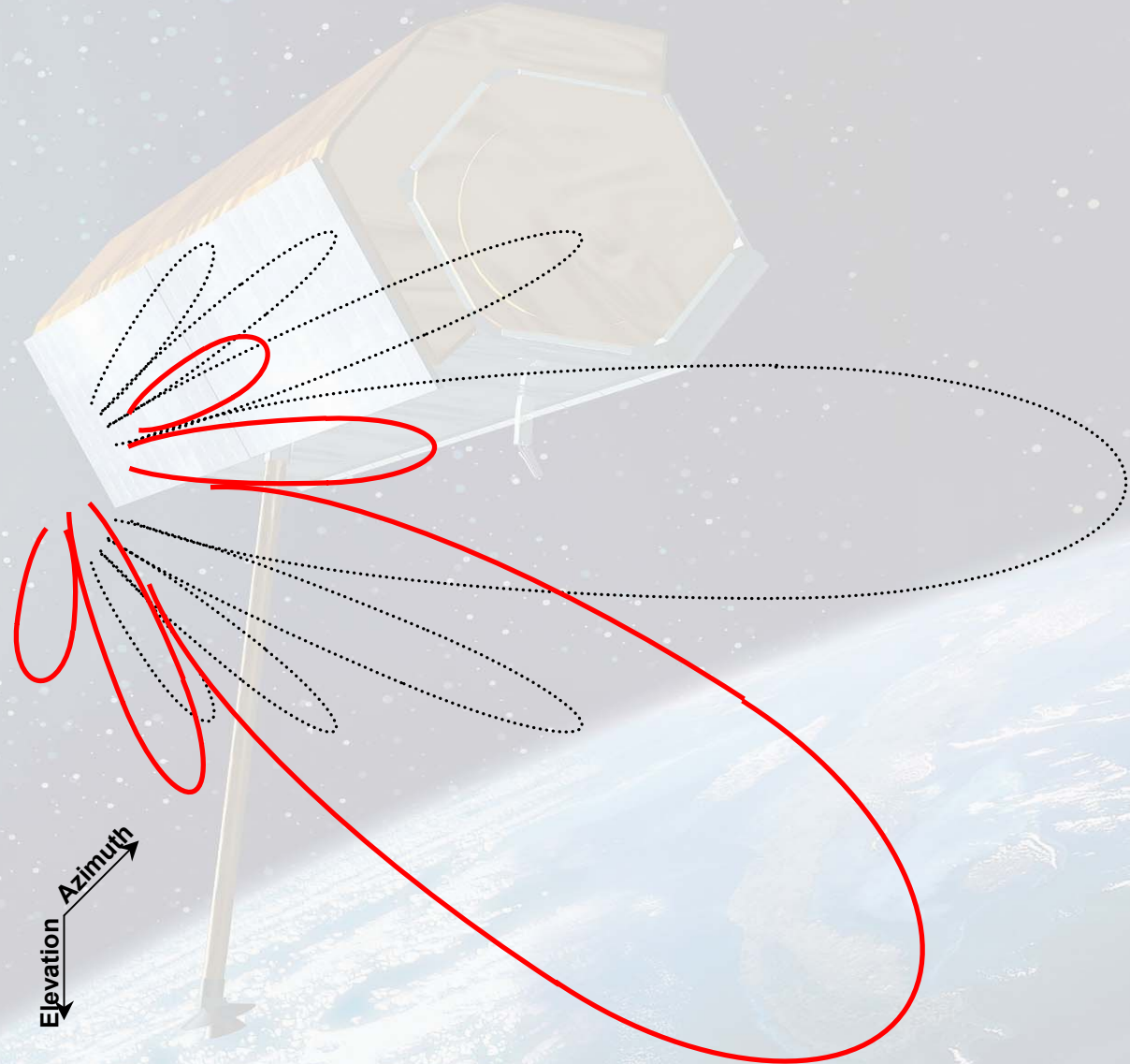
**Linear Phase
Tapering**

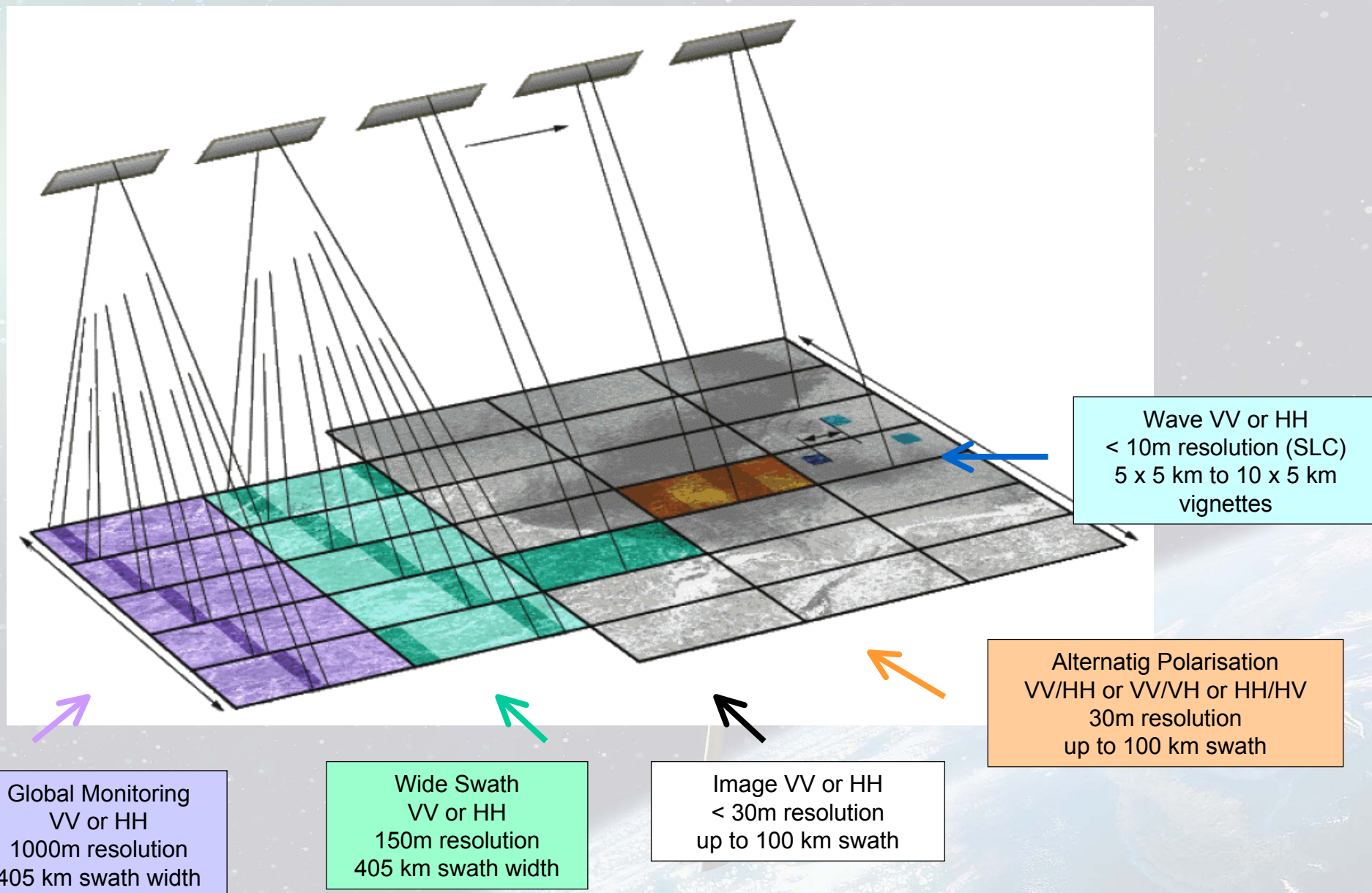




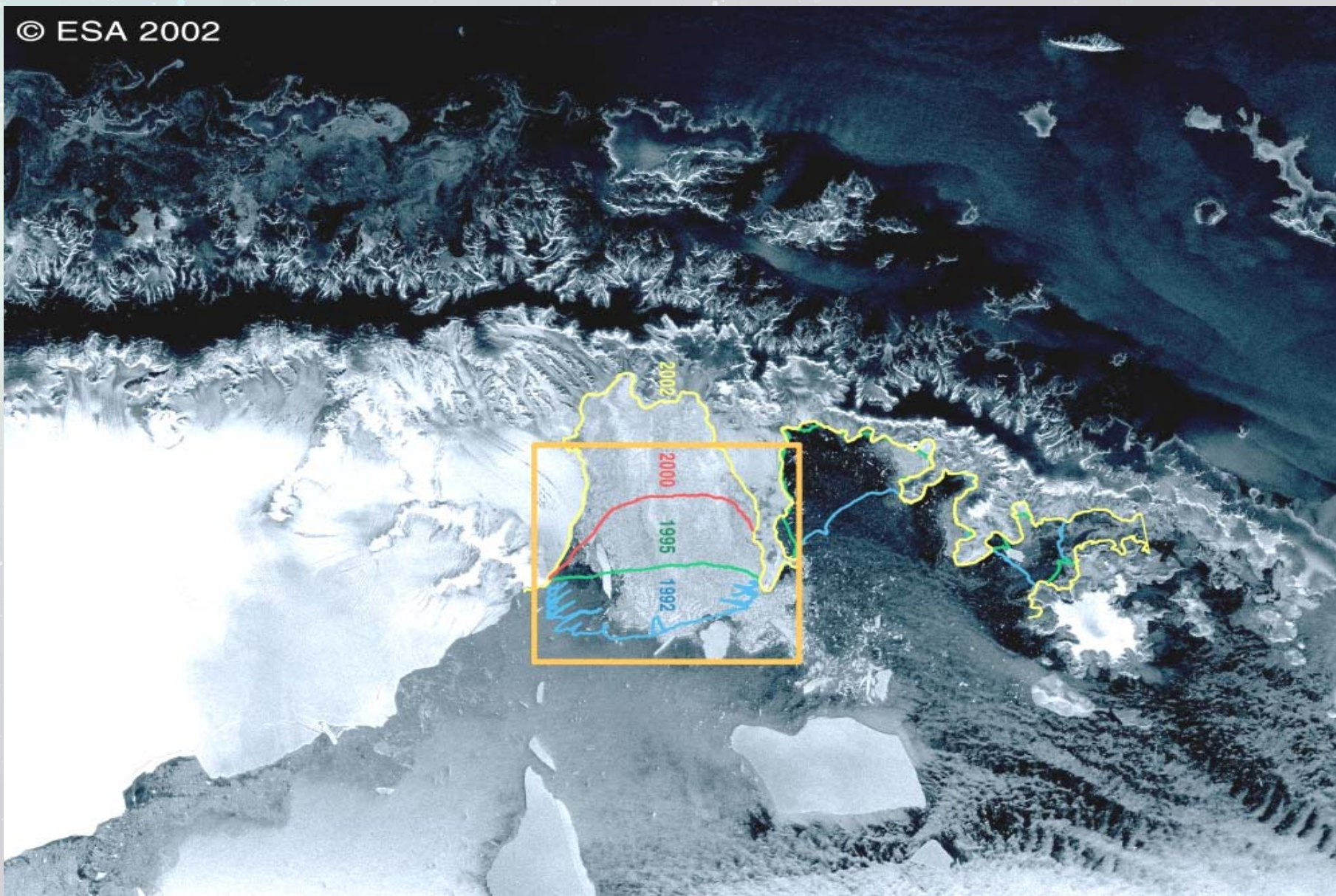
RF-Unit

Pulse Generator,
Up/Down-Conv.,
D/A-Conv.,
Data Storage, etc.



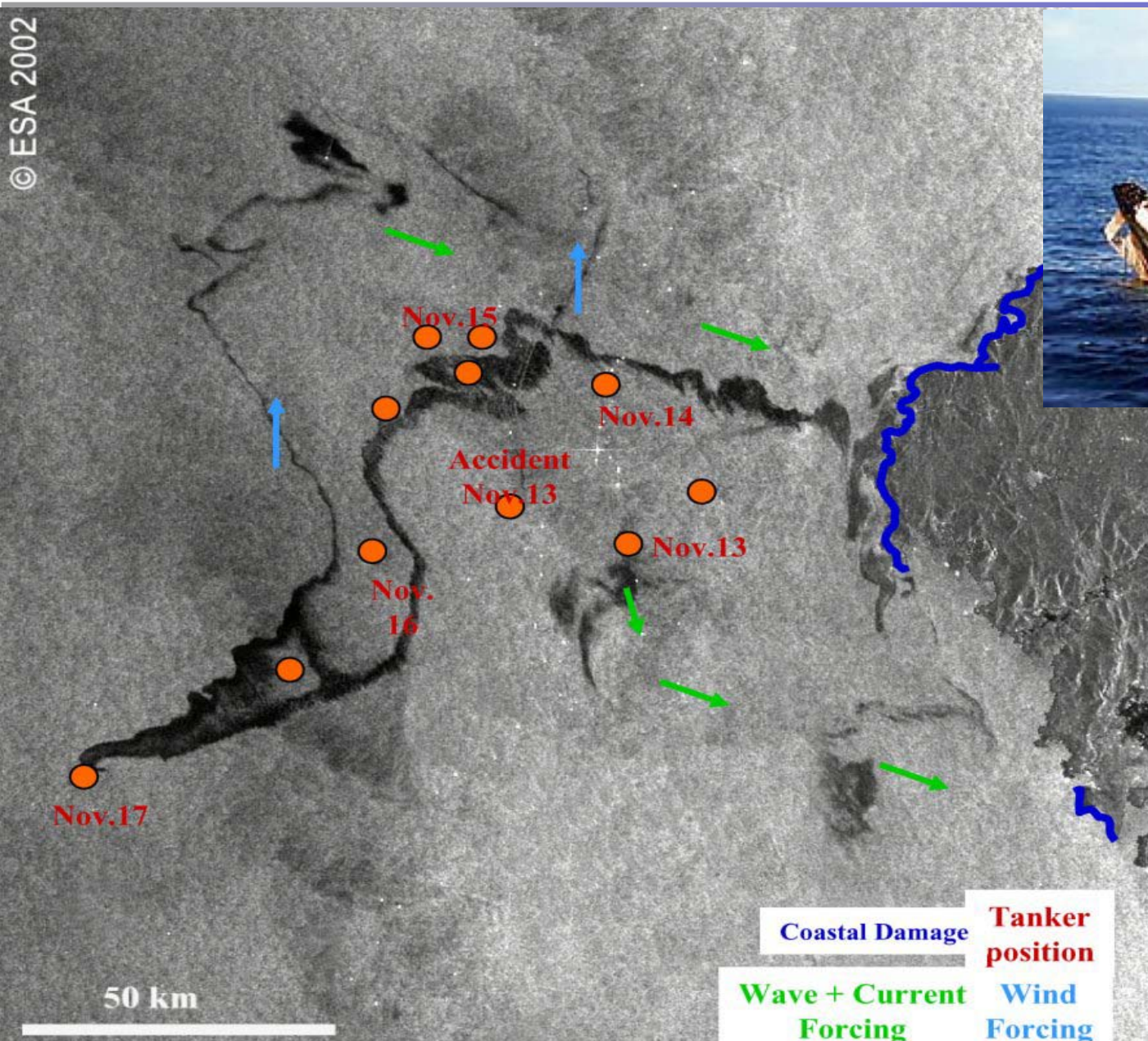


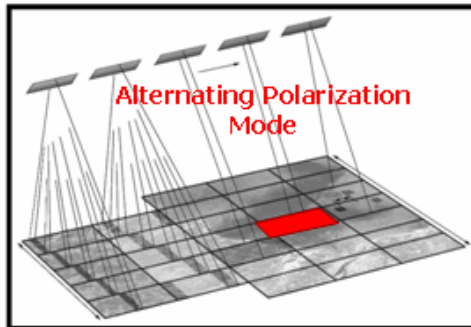
© ESA 2002



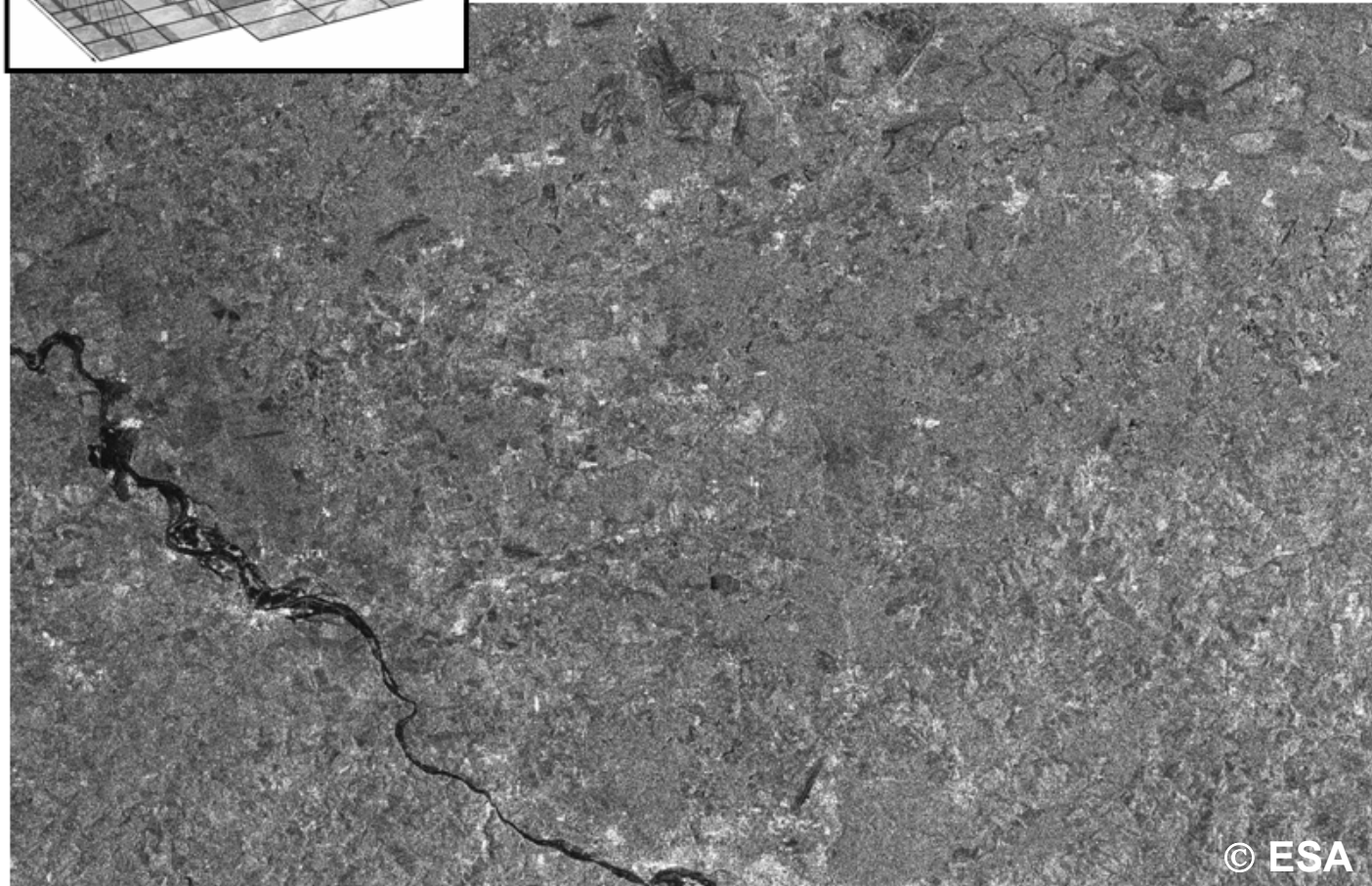
Oil Detection with ASAR

© ESA 2002





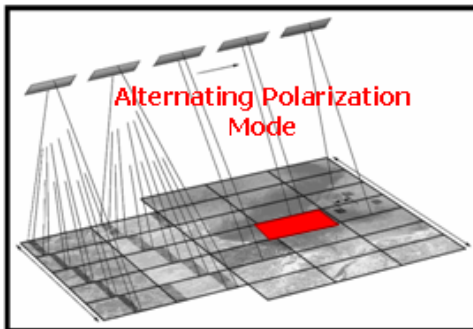
German flood event (Summer 2002)
ASAR Alternating Polarization mode



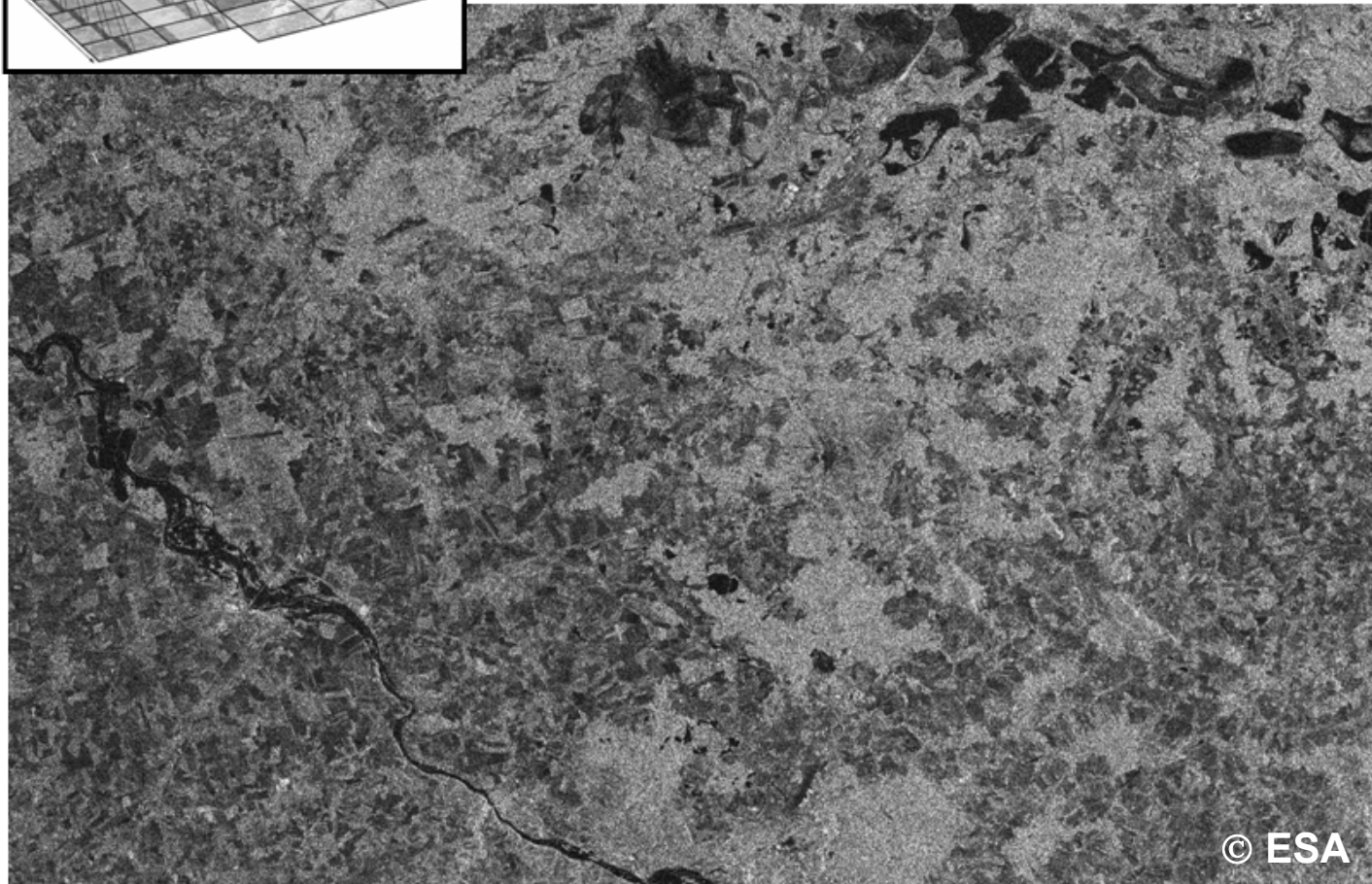
HH

swath IS2
19 August 02

© ESA



German flood event (Summer 2002) ASAR Alternating Polarization mode

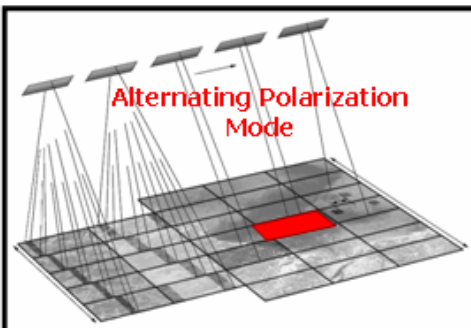


HV

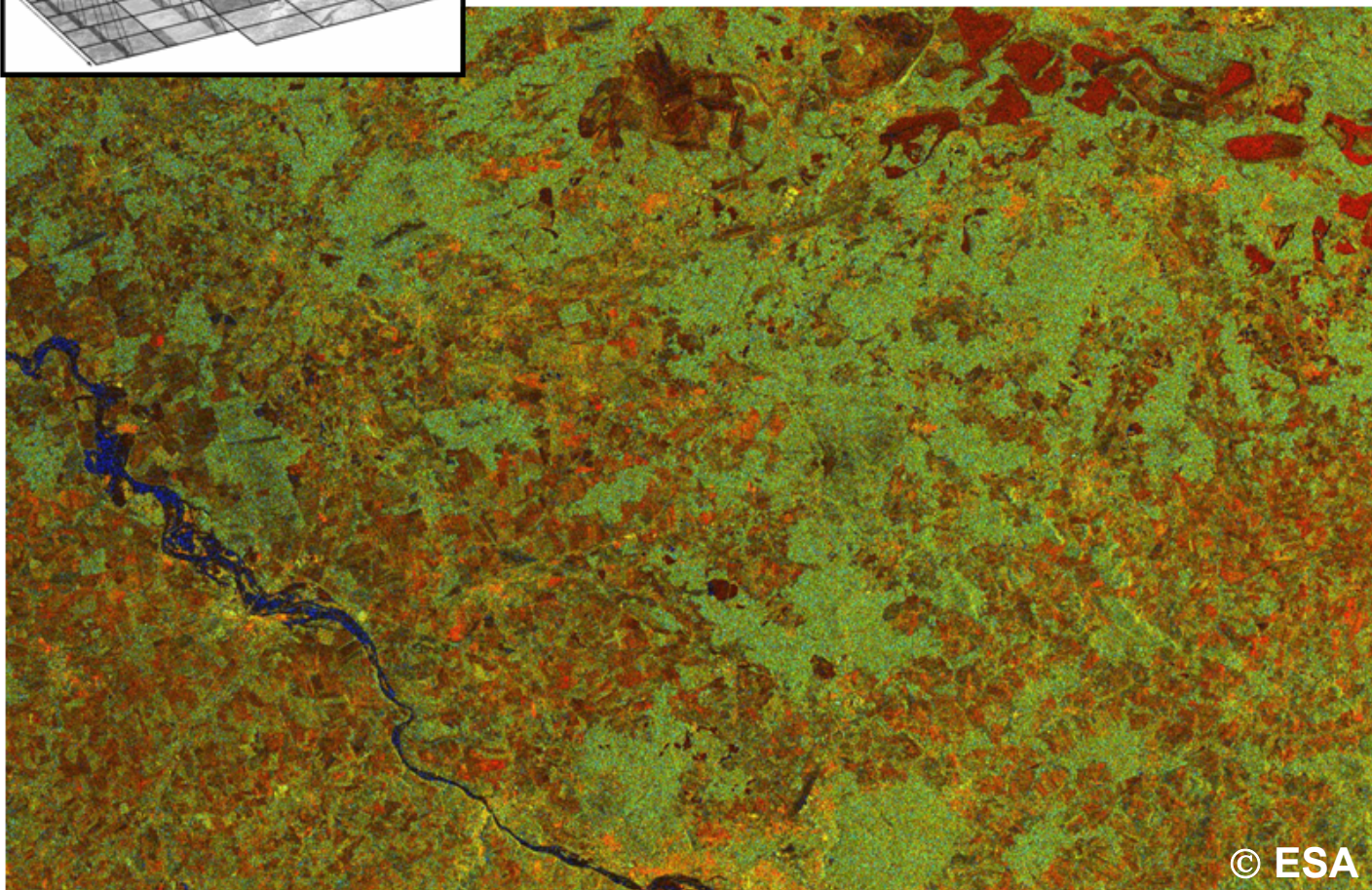
swath I52
19 August 02

© ESA

Single-Pol versus Polarimetric ASAR Image

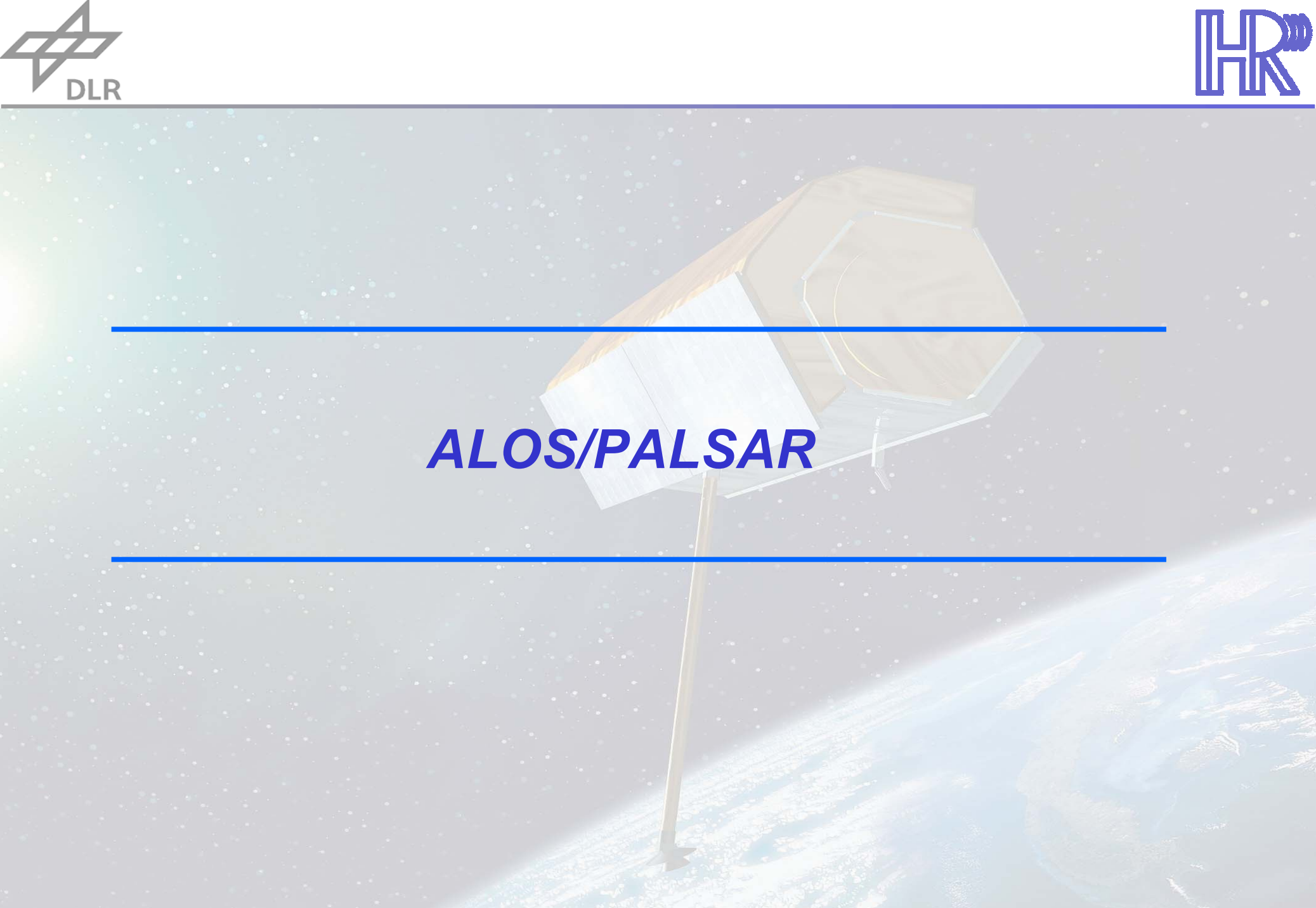


German flood event (Summer 2002)
ASAR Alternating Polarization mode



HH
HV
HH-HV

swath IS2
19 August 02

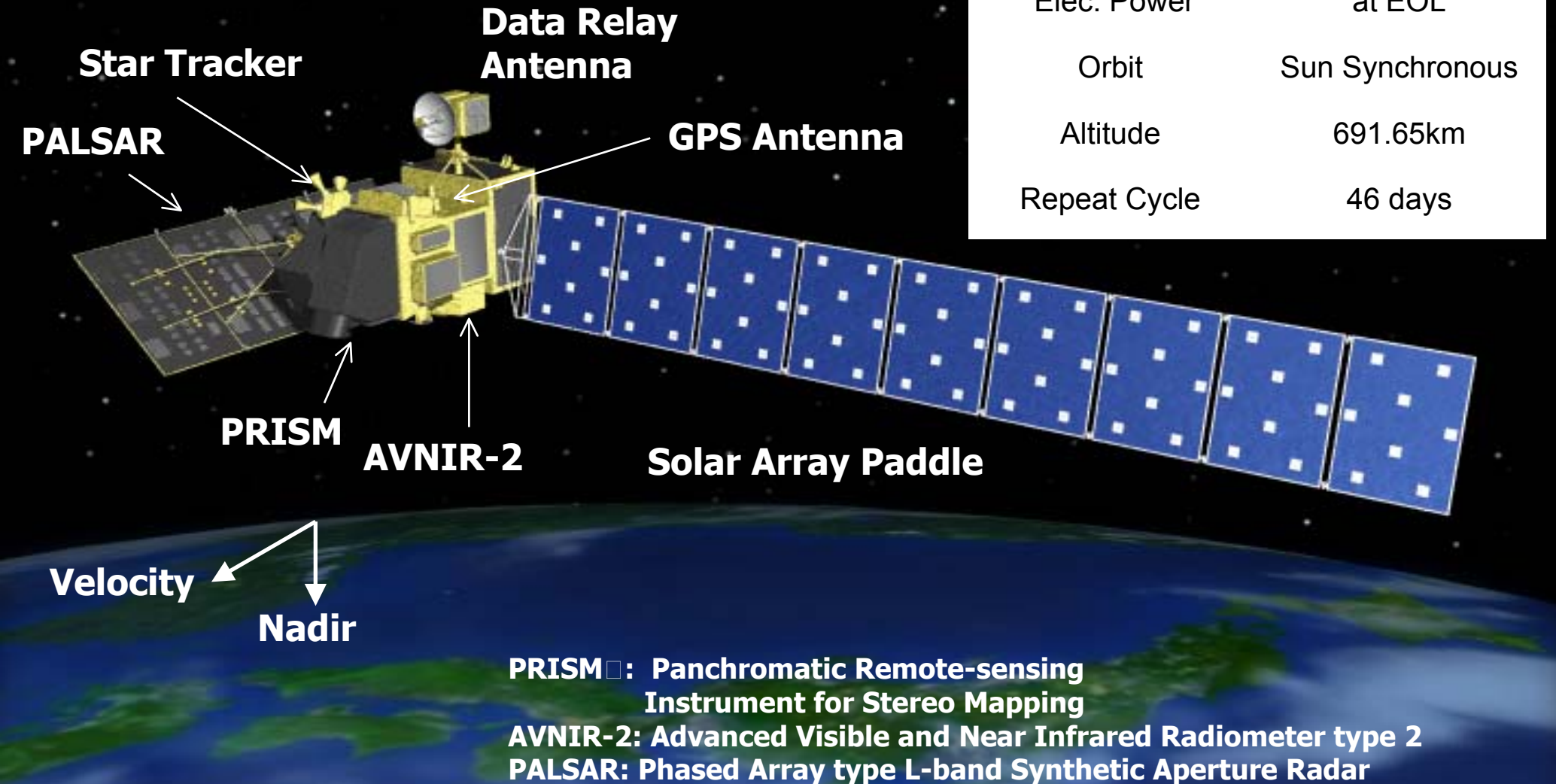


The background of the slide features a 3D rendering of the ALOS/PALSAR satellite in orbit. The satellite is a large, complex structure with a prominent rectangular antenna array. It is positioned against a backdrop of a starry space and the curved horizon of the Earth, which shows blue oceans and brown landmasses. Two horizontal blue lines frame the satellite and the title text.

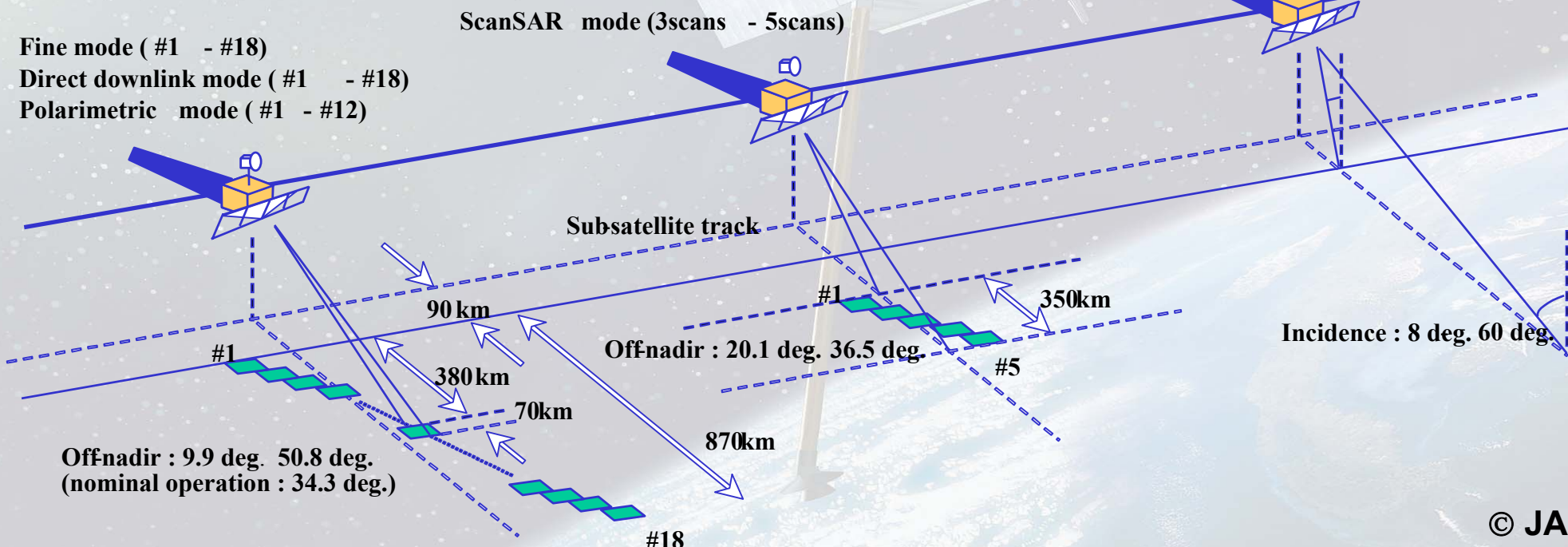
ALOS/PALSAR

ALOS Satellite System

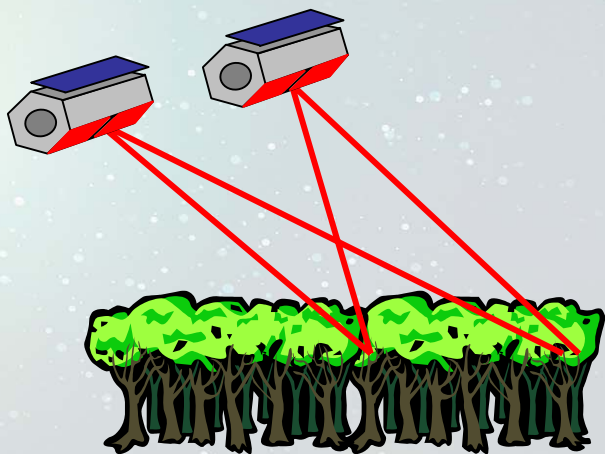
Launch Date	September 2005
Launch Vehicle	H-IIA
Spacecraft Mass	about 4,000kg
Generated Elec. Power	about 7kW at EOL
Orbit	Sun Synchronous
Altitude	691.65km
Repeat Cycle	46 days



Frequency	1270 MHz
Band Width	28/14MHz
Modes	FBS,FBD,POL, SCAN
Antenna beams	18 (STRIP)+5 SCANS
Image Width	70Km
Incidence	7.7 - 60 degrees
Tr. power	2Kw
Resolution	9m(2L)
Data rate	240Mbps



Pol-InSAR



Observables

**3 Interferometric
Coherences**

**3 Interferometric
Phases**

*for one baseline and three
polarisation states*

**I Step:
Model and Validation
Development**

**Coherent
Scattering
Model**

Natural Scatterers



**Forest
Parameters**



**Surface
Parameters**

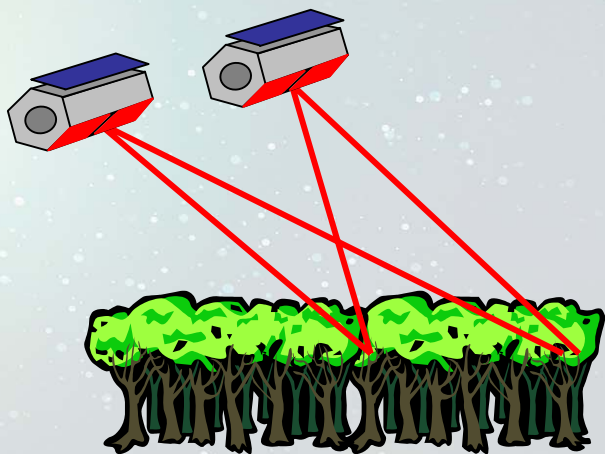


**Agricultural
Parameters**



**Land Ice
Parameters**

Pol-InSAR



Observables

**3 Interferometric
Coherences**

**3 Interferometric
Phases**

*for one baseline and three
polarisation states*

**II Step: Model
and Validation
Inversion**

**Coherent
Scattering
Model**

Natural Scatterers



**Forest
Parameters**



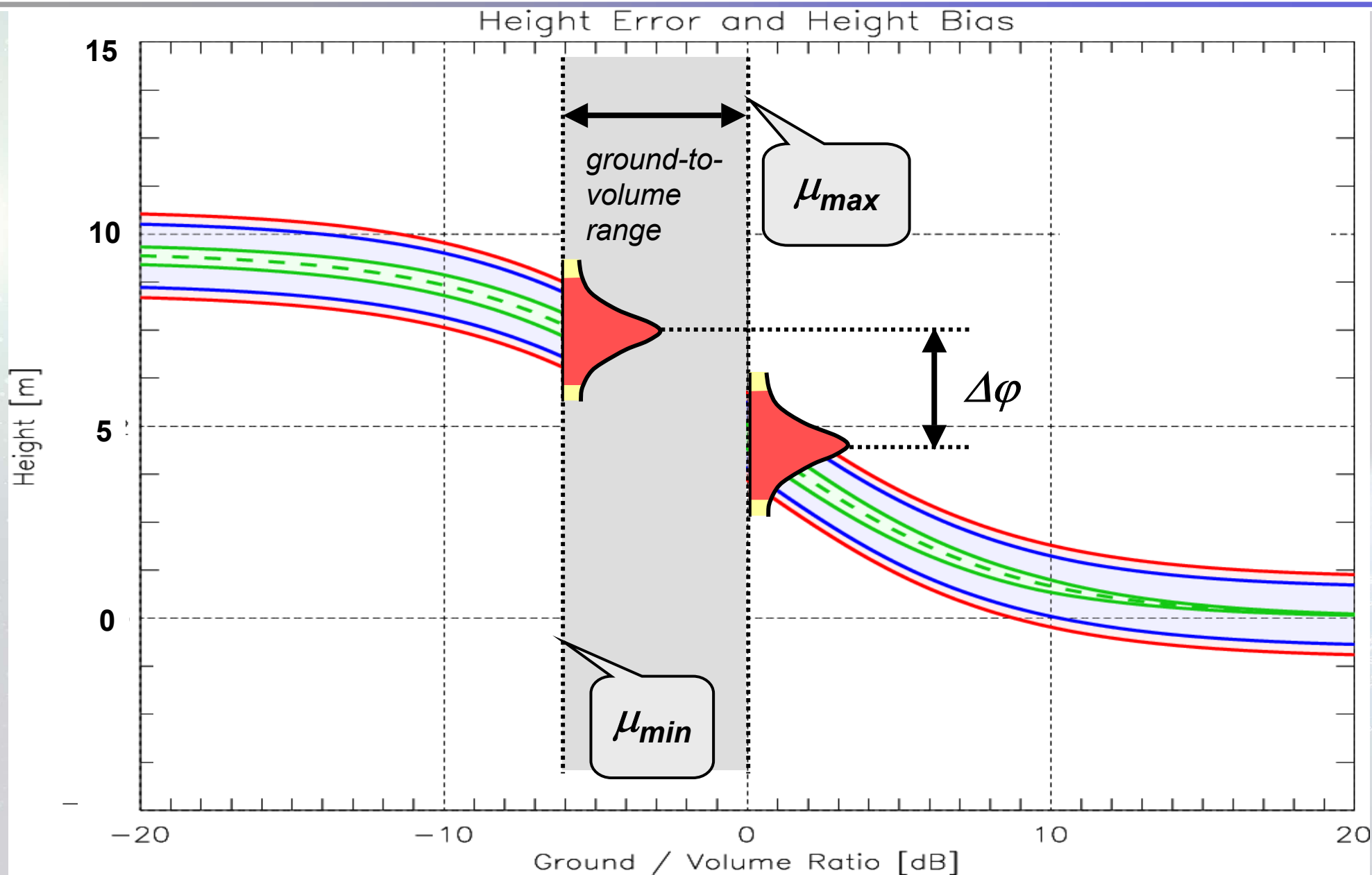
**Surface
Parameters**

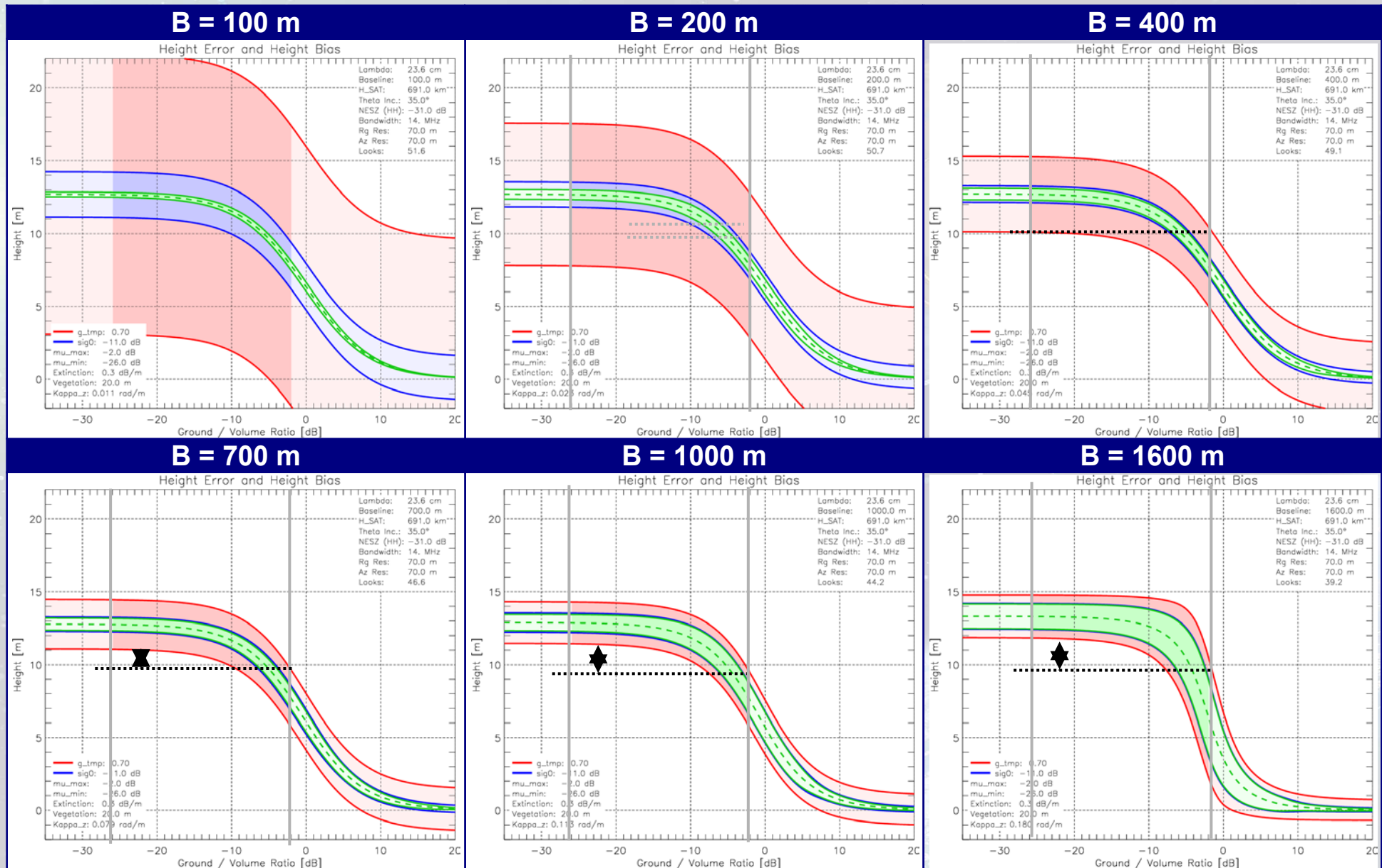


**Agricultural
Parameters**

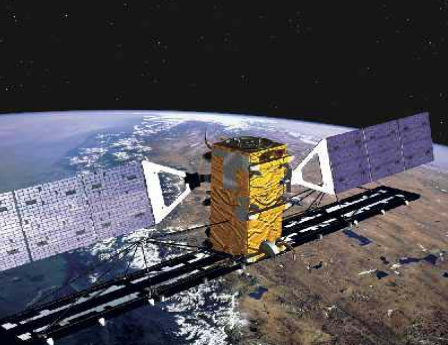


**Land Ice
Parameters**





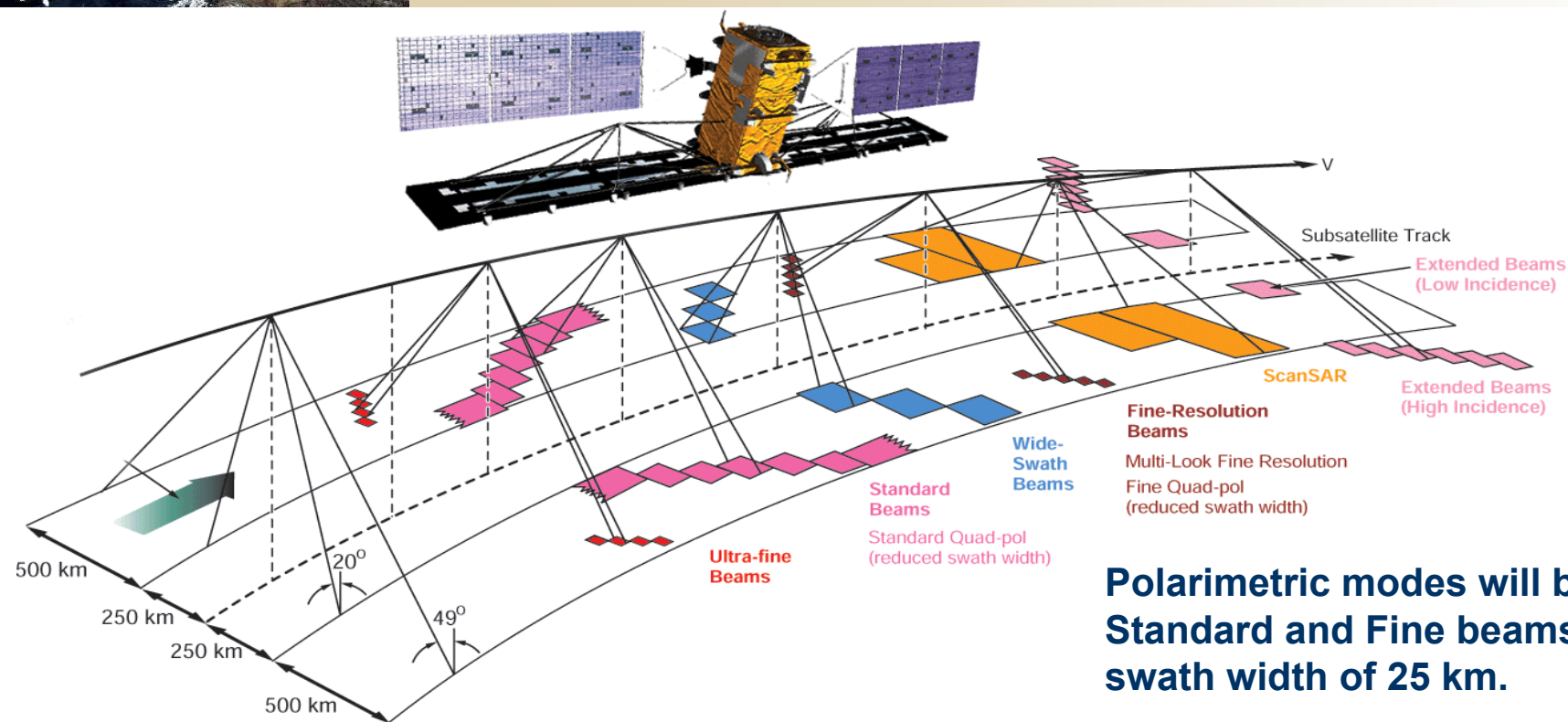
RADARSAT-II



RADARSAT-2



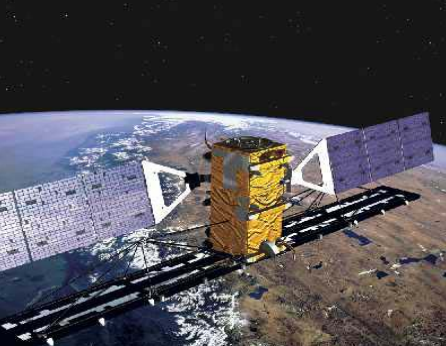
RADARSAT-2 Imaging Modes



Polarimetric modes will be available on Standard and Fine beams with a limited swath width of 25 km.

All modes will be available in selective single or dual polarization.

All modes will be available on either side of the satellite.

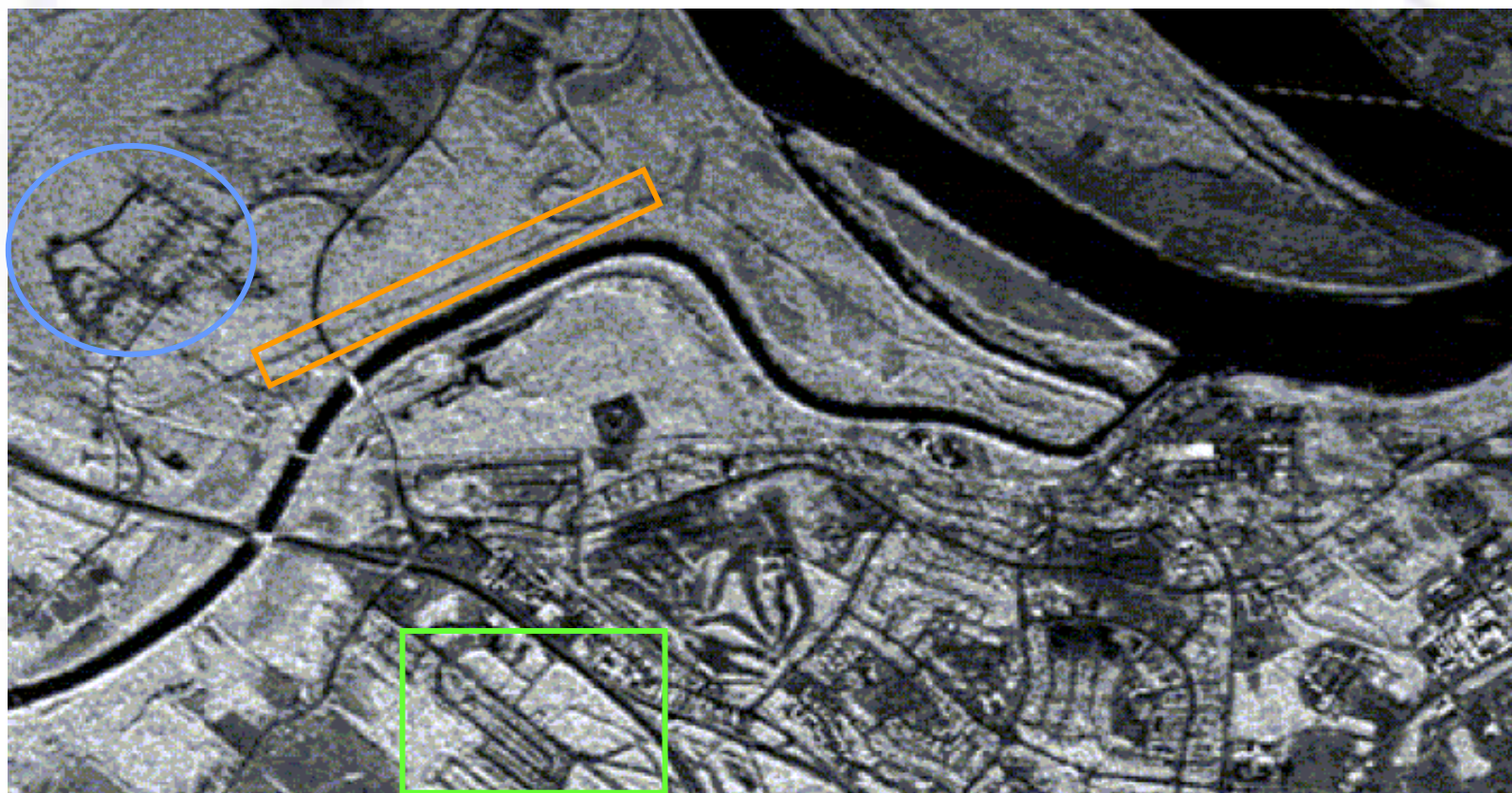


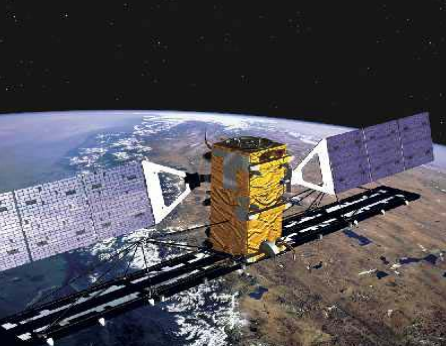
RADARSAT-2



Ultra-fine Imaging

Comparison
between
RADARSAT-1 Fine
beam (10 m
resolution) and
simulated
RADARSAT-2
Ultra-fine beam (3
m resolution)

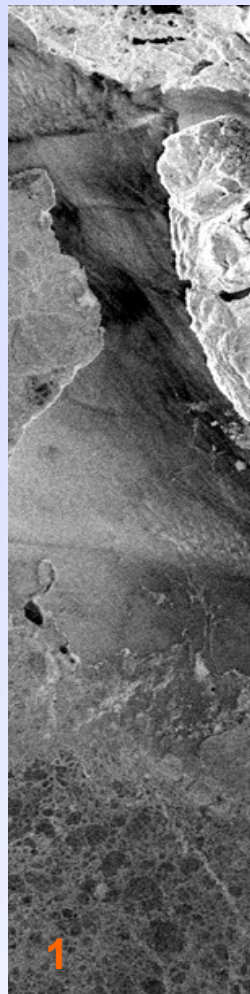




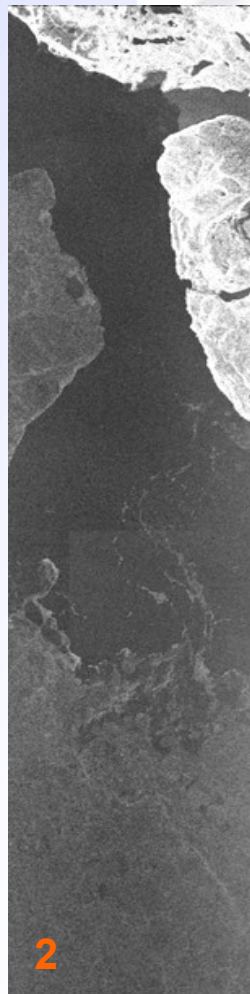
RADARSAT-2



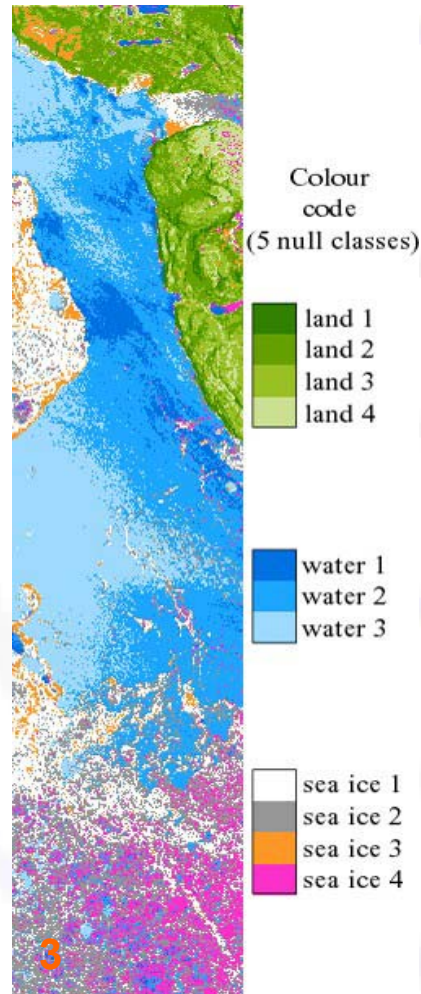
Polarimetric Application: Sea Ice



1



2



1- HH polarization image showing enhanced ice-type information.

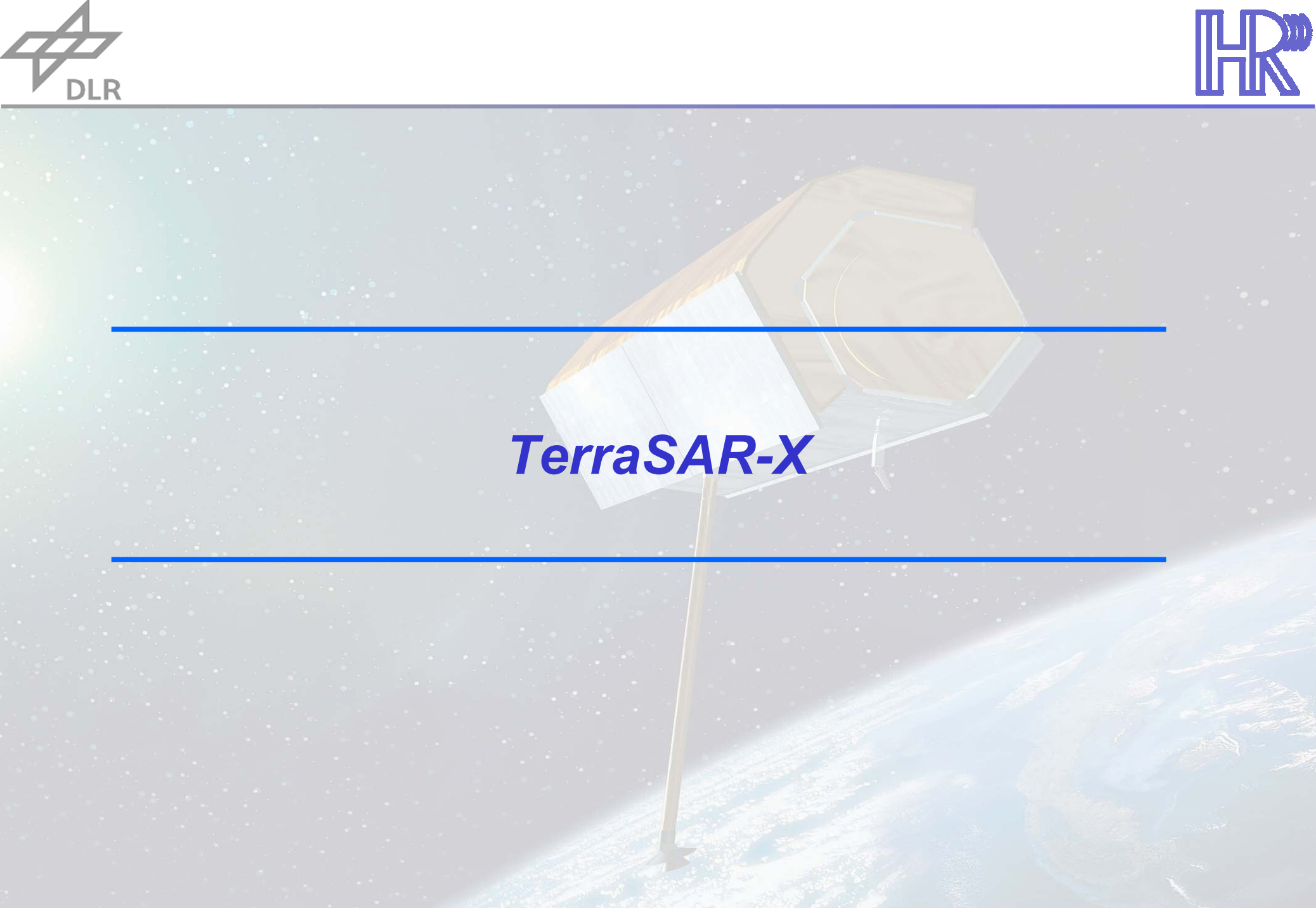
2- HV polarization image showing enhanced ice-edge information.

3- H/A/ α Maximum Likelihood classification results after five iterations.

Data acquired from C-band Shuttle Imaging Radar (SIR) of the west coast of Newfoundland, April, 1994.

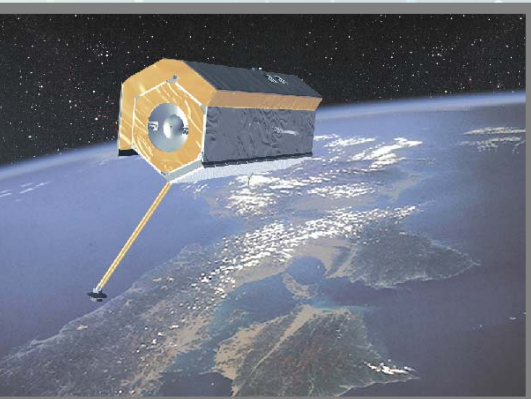
Study performed by MDA through the Earth Observation Application Development Program (EOADP) of the the Canadian Space Agency.



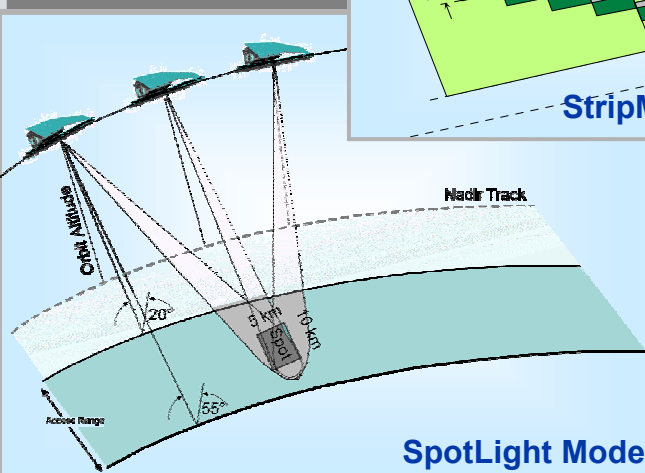
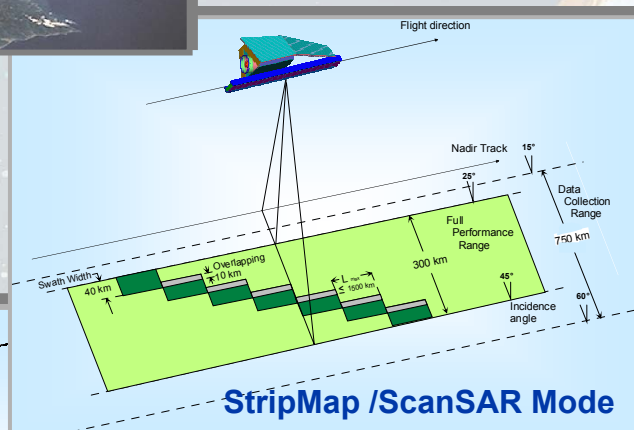


The background of the slide features a 3D rendering of the TerraSAR-X satellite in orbit above the Earth. The satellite is shown from a perspective that highlights its large, rectangular radar antenna array, which is composed of many smaller elements. The satellite's body is a complex, multi-faceted structure. The Earth's surface is visible in the lower right, showing a mix of blue oceans and brown/green landmasses. The sky is a deep blue with numerous white stars. Two horizontal blue lines are drawn across the middle of the image, framing the satellite's name.

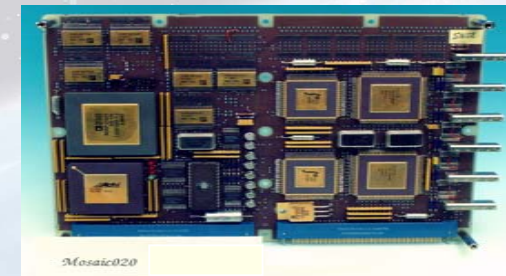
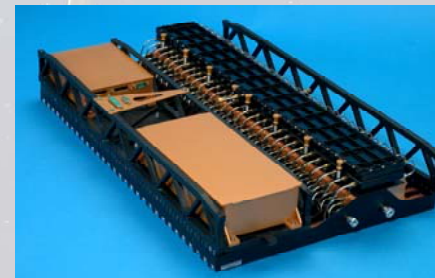
TerraSAR-X



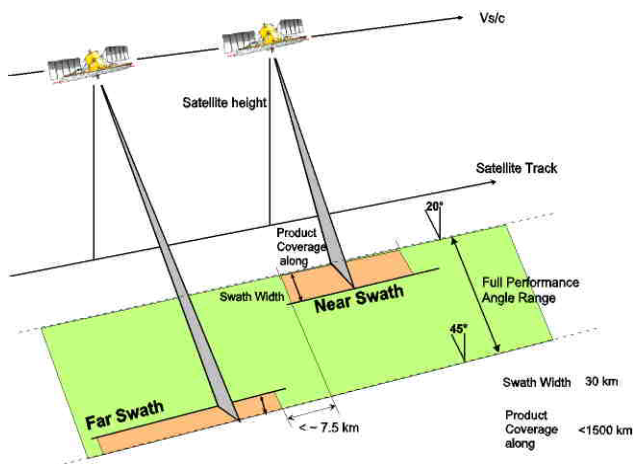
Start: 2006



- Private Public Partnership between DLR and ASTRIUM
- Goal is to establish a sustainable Global EO information service
- Technology demonstrators:
 - DESA: X-Band Front-End Demonstrator
 - TOPAS: on-board SAR Processing and Storage Demonstrator (1999-2002)

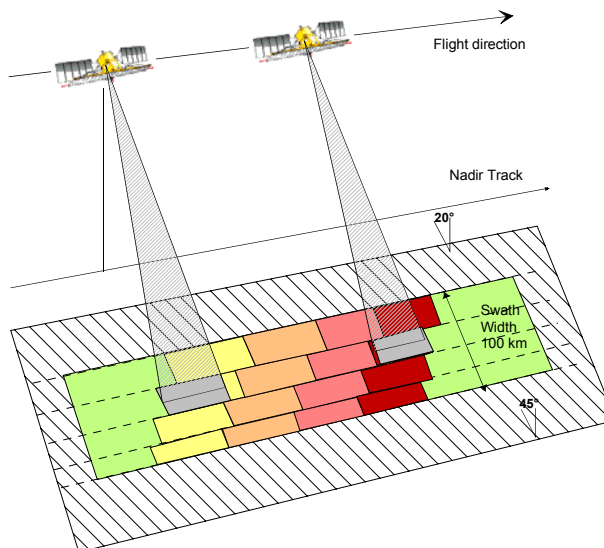


- Continuation of a successful X-Band technology program in Germany



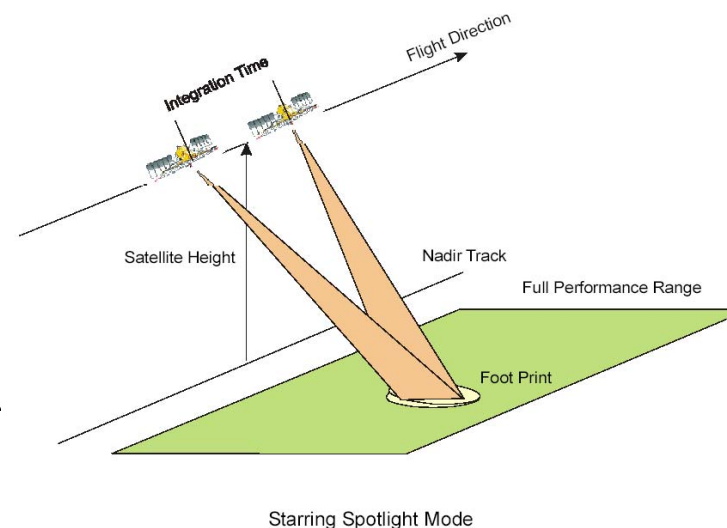
Strip Map Mode

Resolution: 3 m × 1,5 m ... 3,5 m
Scene Size: 100 km × 30 km



Scan SAR Mode

Resolution: 3 m × 1,5 m ... 3,5 m
Scene Size: 100 km × 100 km

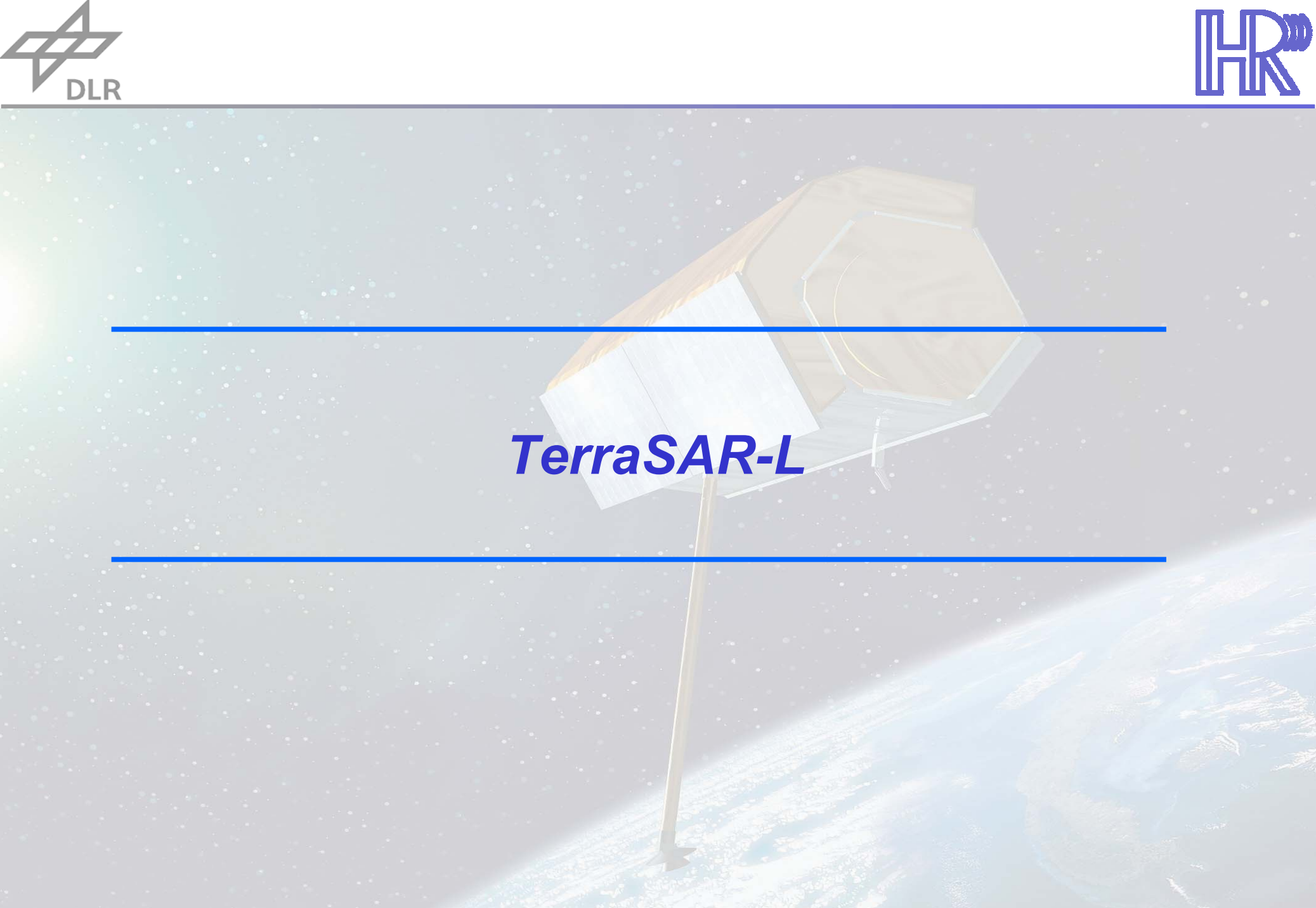


Spotlight Mode

Resolution: 1 m × 1,5 m ... 3,5 m
Scene Size: 5 km...10 km × 10 km

- The data collection incidence angle range is from 15° to 60°
- Dual polarisation data acquisition is possible in all operational modes
- 300 MHz experimental mode with 0.5 m slant range resolution
- Further experimental modes are feasible due to the Dual Receiving

Antenna

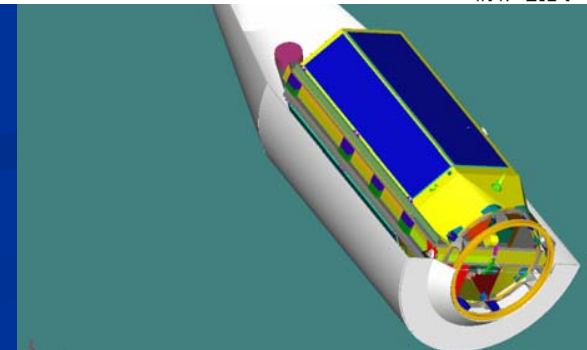
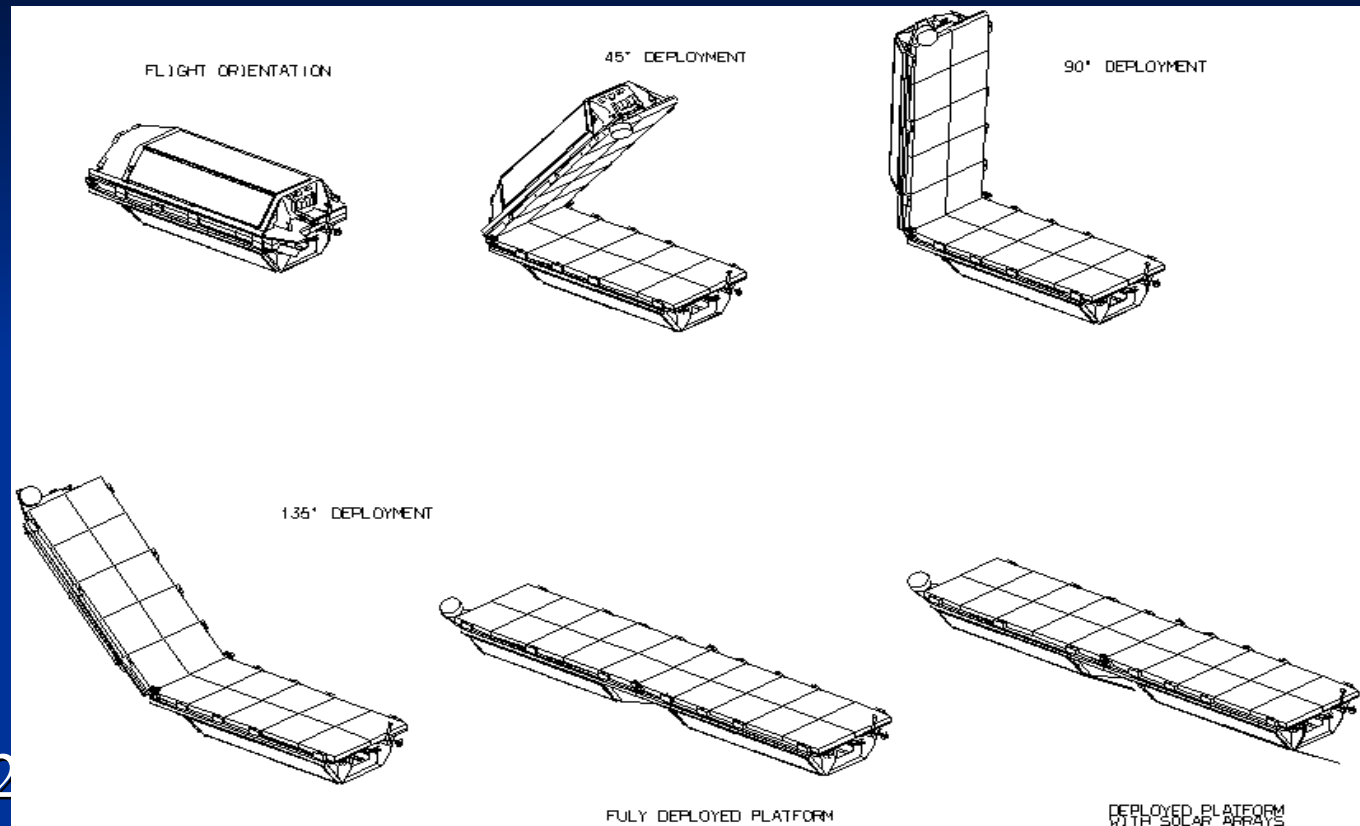


The background of the slide features a 3D rendering of the TerraSAR-L satellite in orbit. The satellite is a large, complex structure with a prominent white rectangular panel and a large, gold-colored, multi-faceted antenna. It is positioned against a backdrop of a starry space and the Earth's horizon, which shows blue oceans and brown landmasses. Two horizontal blue lines frame the satellite and the text.

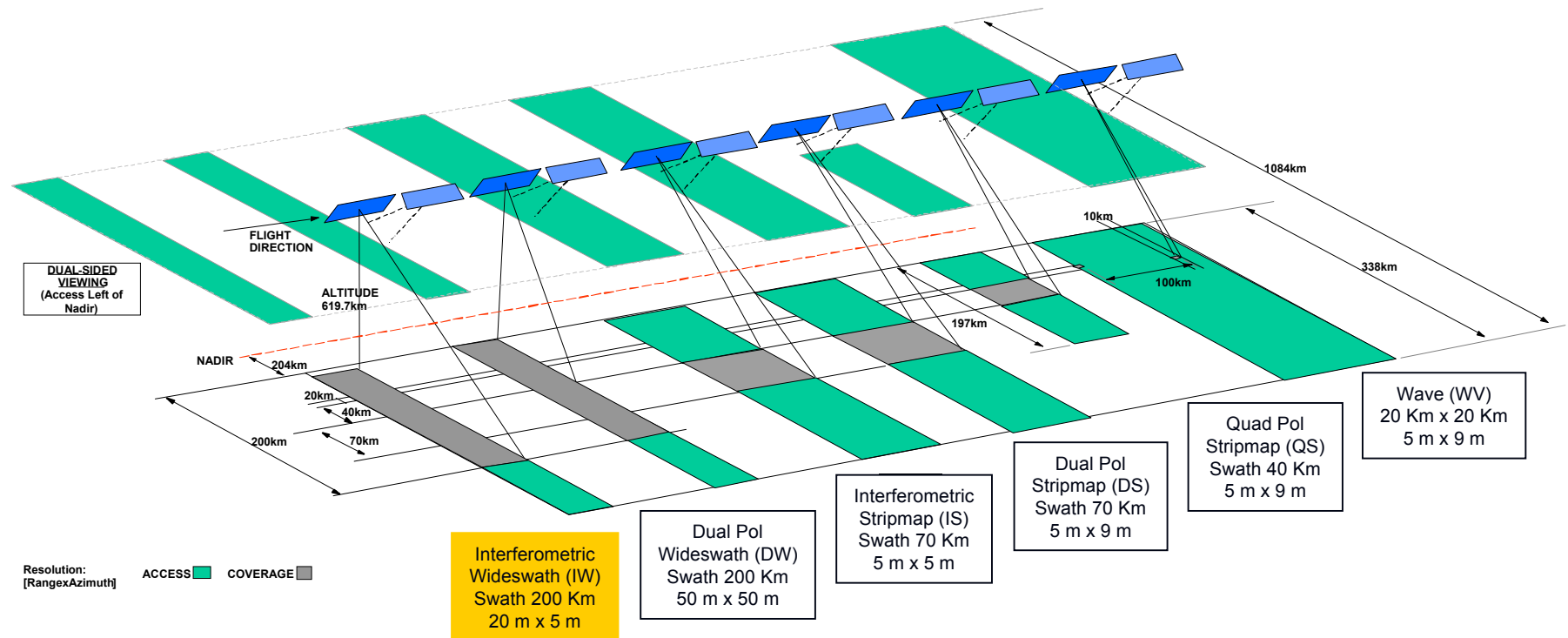
TerraSAR-L

The Spacecraft

- Snapdragon
- Mass: 2700 kg
 - 1800 kg p/l
 - 900 kg L-SAR
- Antenna: 32 m²
- Power: 5 kW
 - 4.5 kW L-SAR
- On-board Memory: 622
- Data downlink: 300 MHz



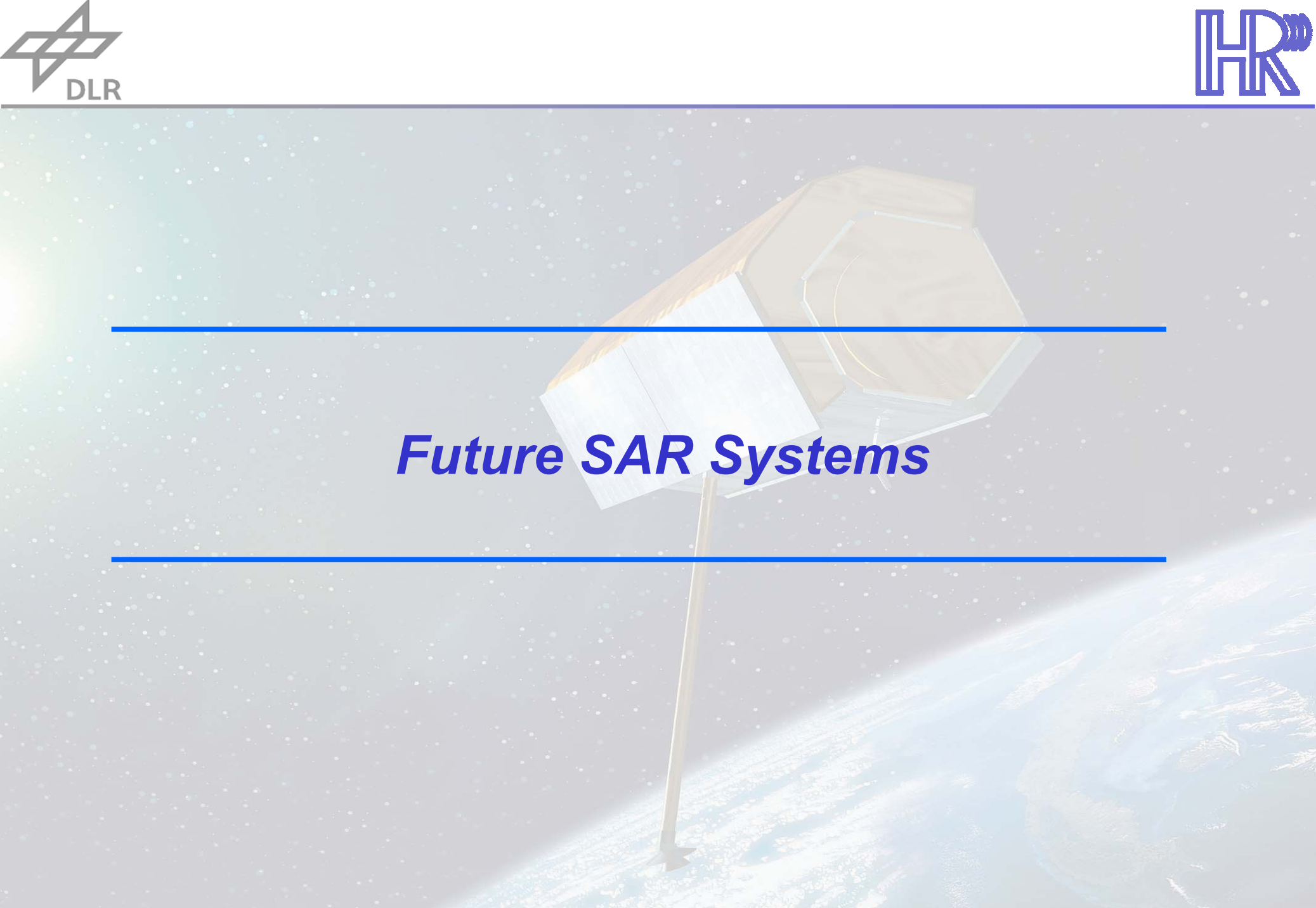
L-SAR Modes



- QS, DS, DW pol. + high radiometric performance
- IW, IS for interferometry – **IW as main mode**
- WV for wave spectra

Future SAR Systems



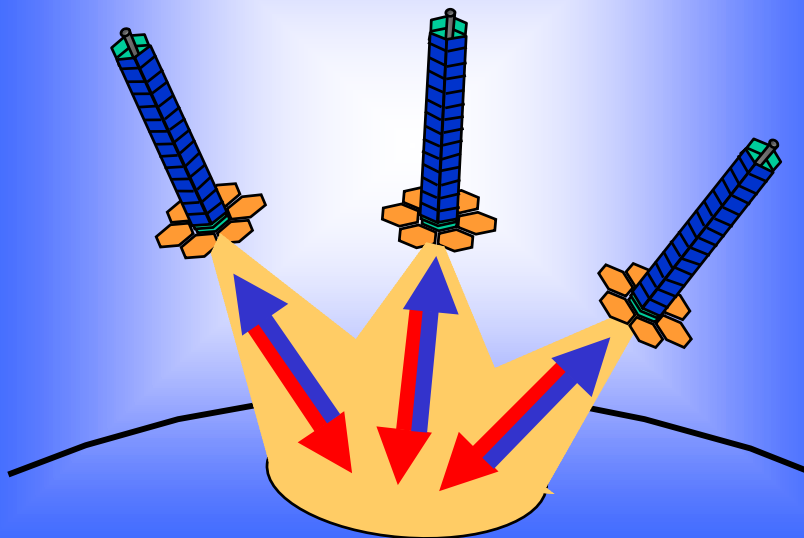


Future SAR Systems

- **Bistatic Systems:** spatial separation between transmitter and receiver
- **Multistatic Systems:** multiple transmitters/receivers

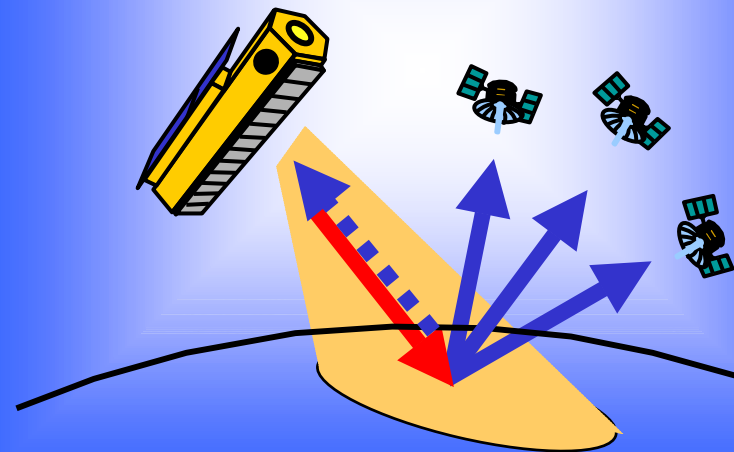
Fully active system:

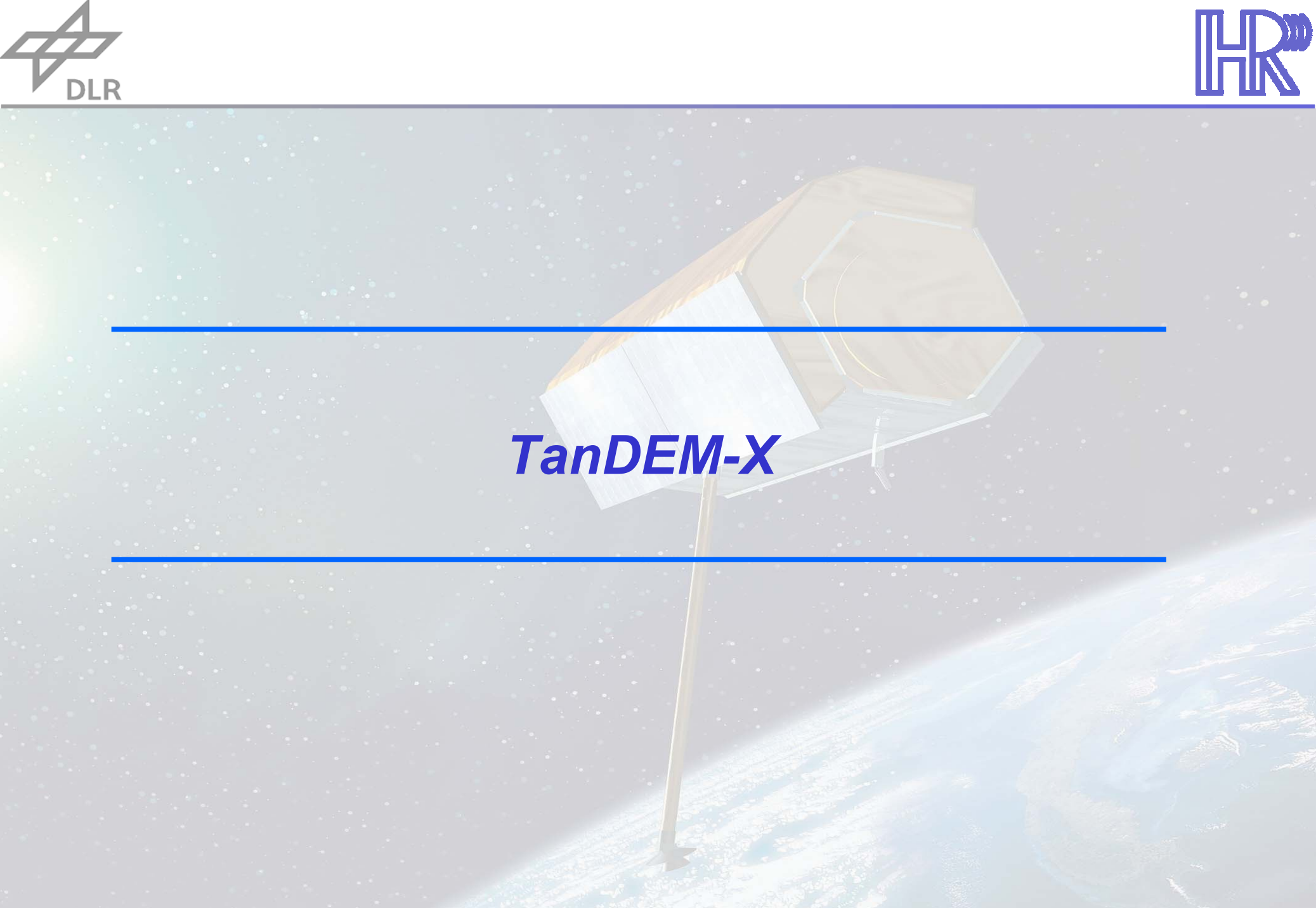
- TechSAT21, TanDEM-X, RADARSAT-2/3



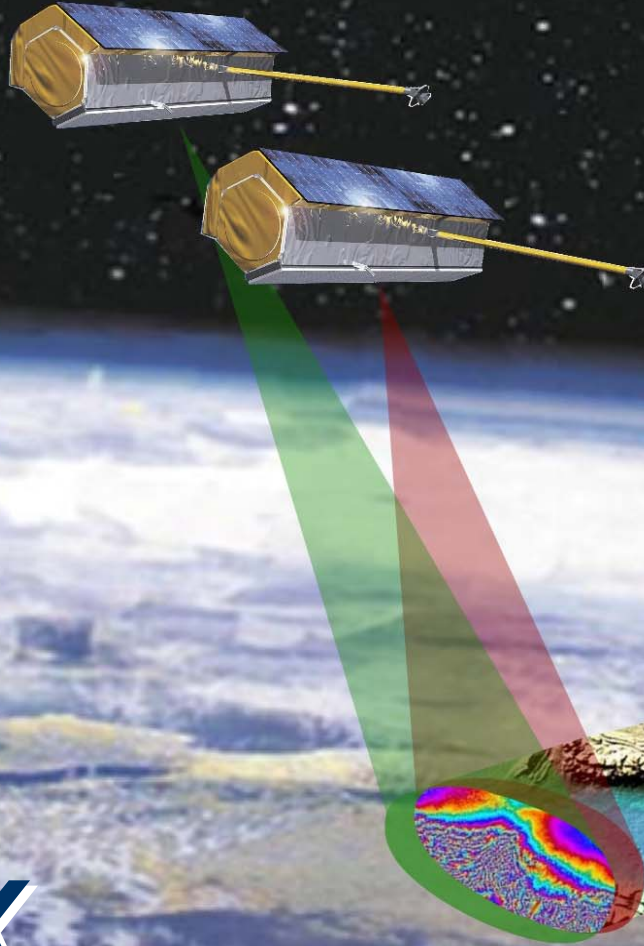
Partially active system:

- ARGUS/FROMAGE, BISSAT, Cartwheel, Pendulum





TanDEM-X



TanDEM-X

**Response to DLR Call
for Proposals for the next
National Earth Observation Mission**

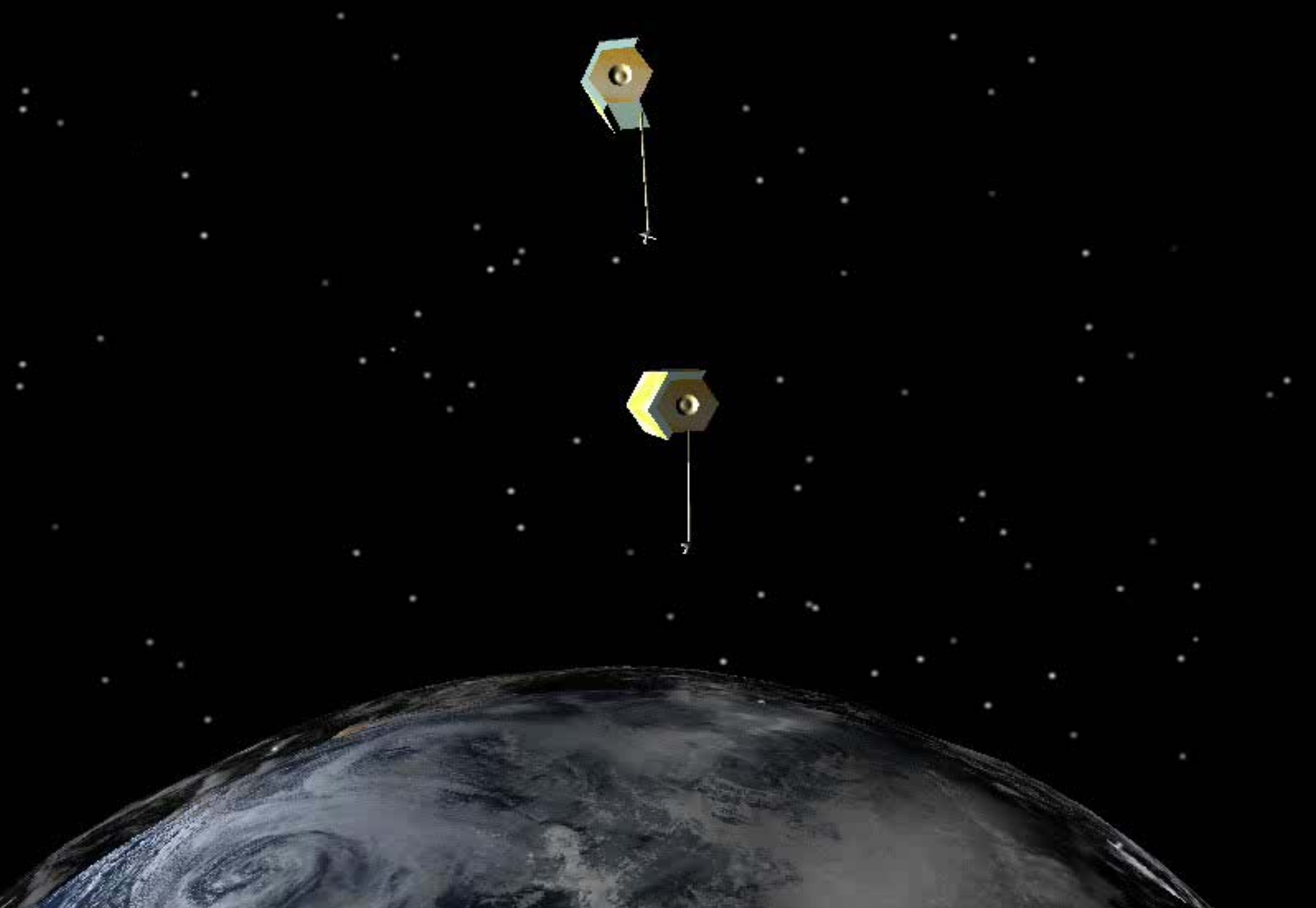


SRTM / X-SAR

DTED-2

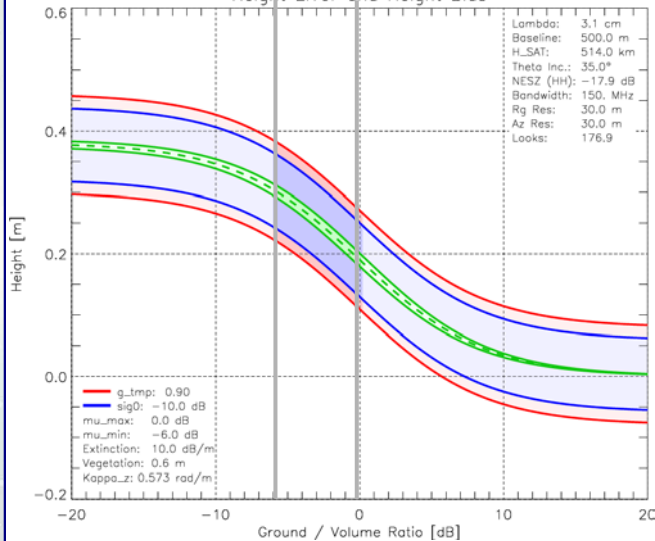
E-SAR (Simulation TanDEM-X)

DTED-3



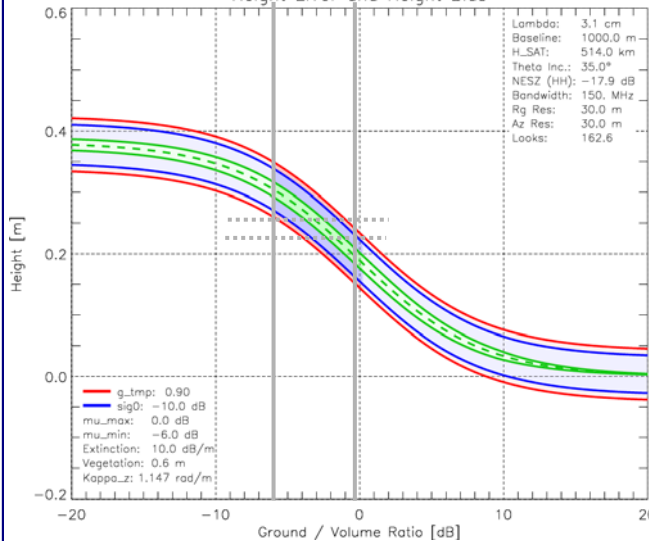
B = 500 m

Height Error and Height Bias



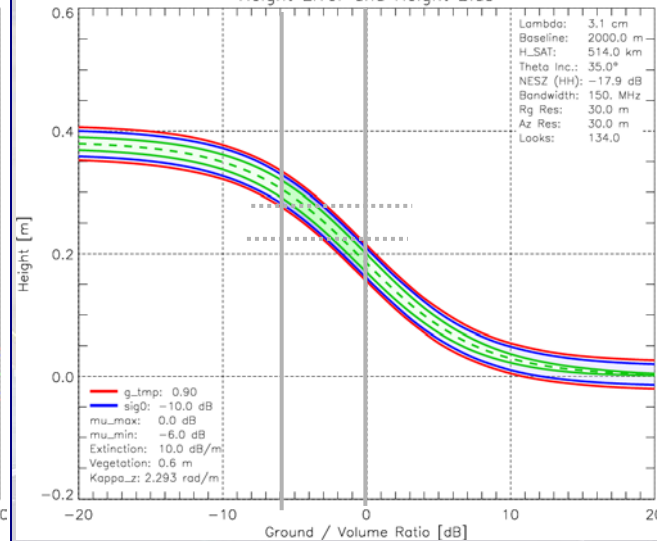
B = 1000 m

Height Error and Height Bias



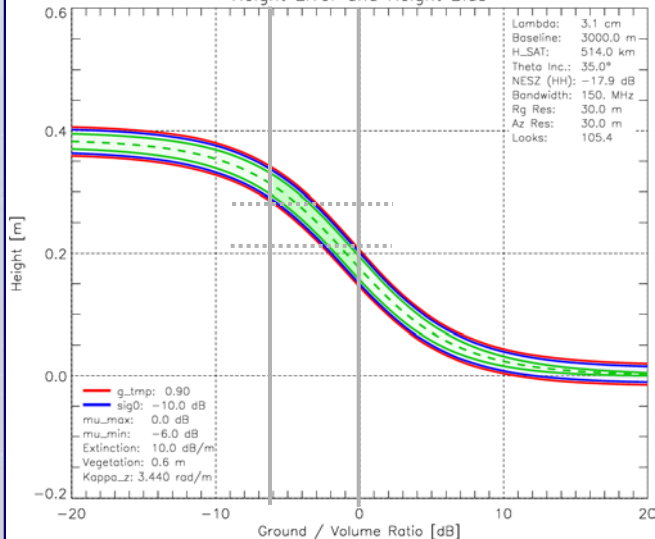
B = 2000 m

Height Error and Height Bias



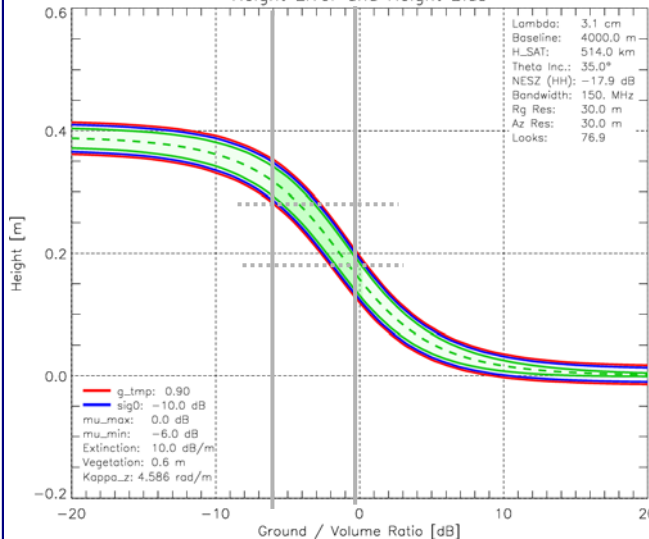
B = 3000 m

Height Error and Height Bias



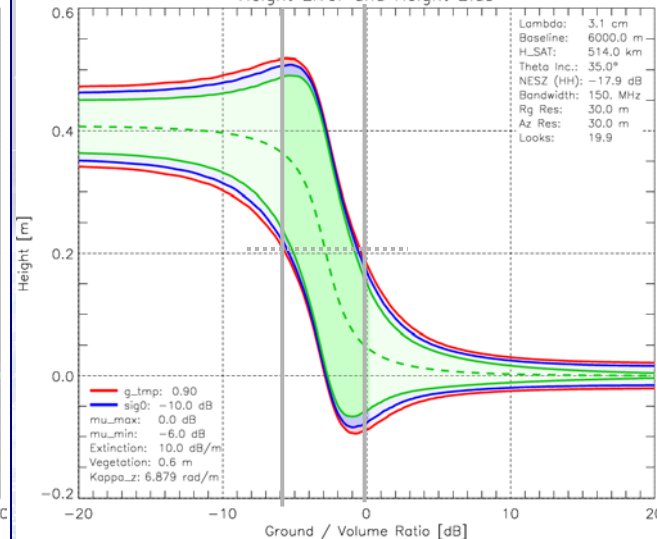
B = 4000 m

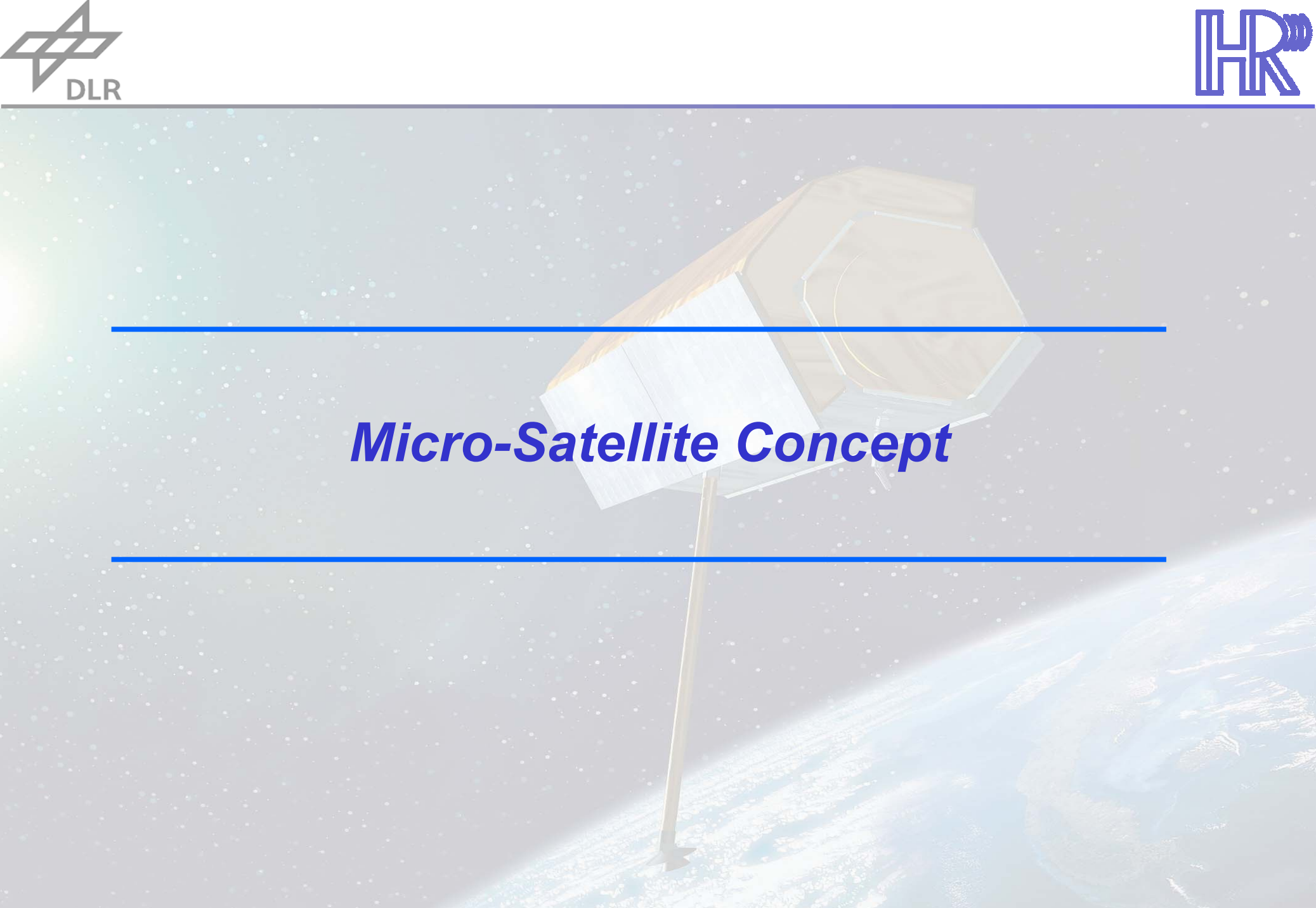
Height Error and Height Bias



B = 6000 m

Height Error and Height Bias

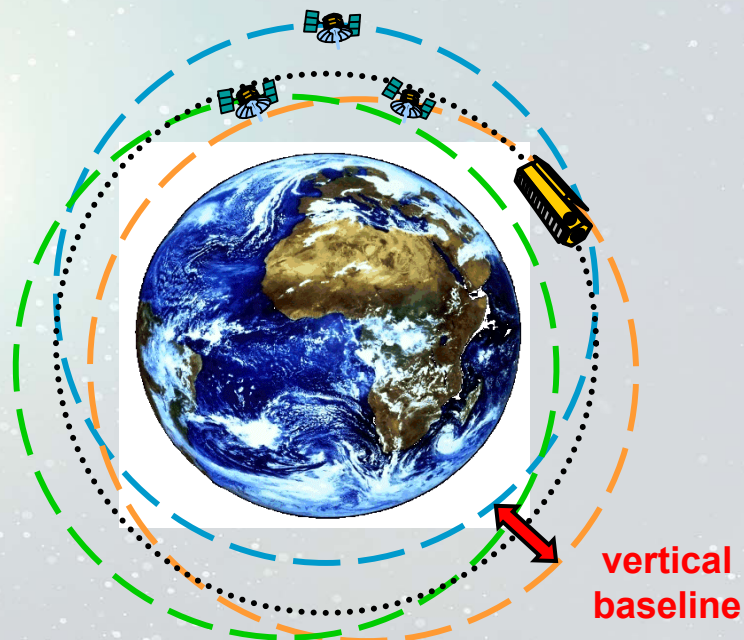




The background of the slide features a 3D rendering of a micro-satellite in orbit. The satellite has a white rectangular body and a large, gold-colored, multi-faceted solar panel array. It is positioned against a backdrop of a starry space and the curved horizon of the Earth, which shows blue oceans and white clouds. Two horizontal blue lines frame the title text.

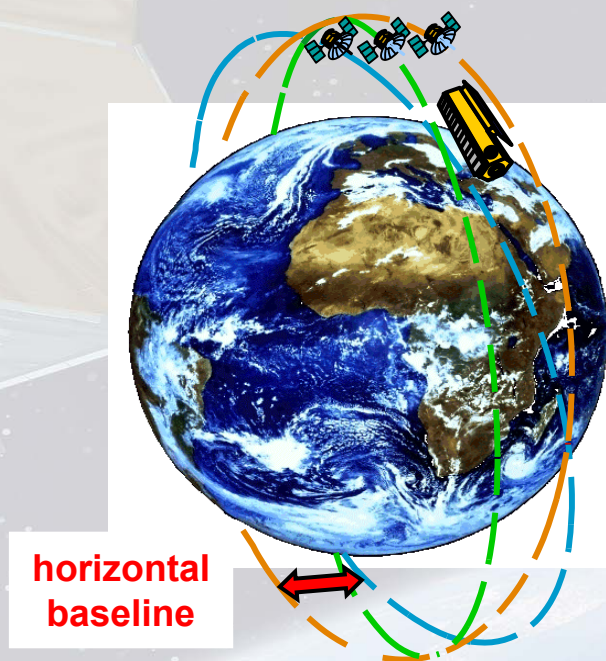
Micro-Satellite Concept

Interferometric Cartwheel



Digital Elevation Model (DEM)
from SRTM Mission

Cross-Track Pendulum



Single-Pass Cross-Track Interferometry with Multistatic Satellite Configuration:

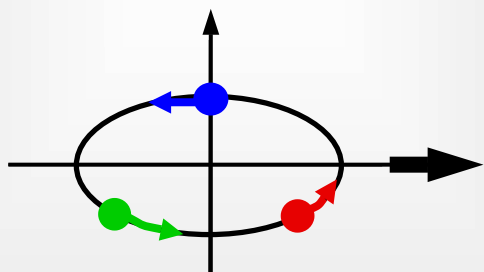
- *no temporal decorrelation (as opposed to repeat-pass interferometry)*
- *no atmospheric distortions (as opposed to repeat-pass interferometry)*
- *large interferometric baselines (as opposed to e.g. SRTM)*

Master Satellite:

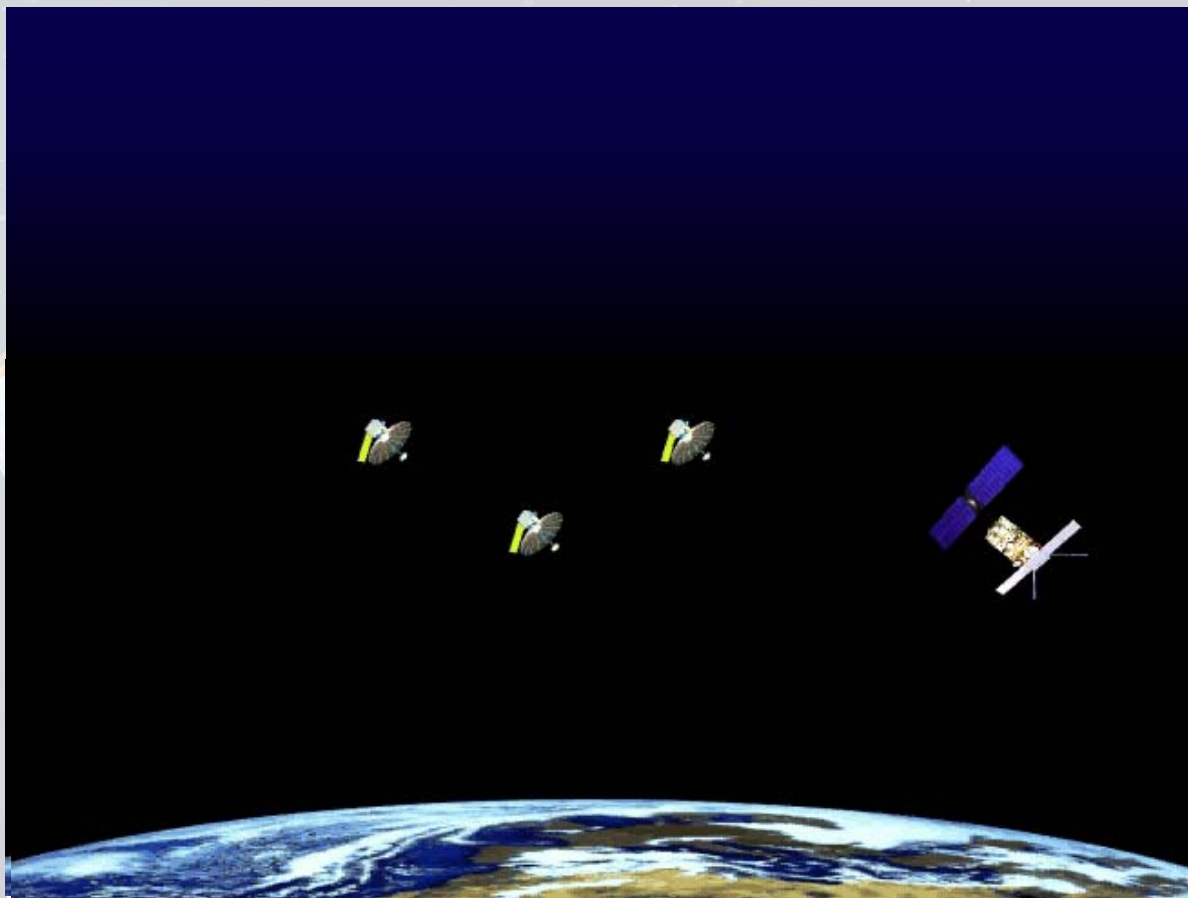
- e.g. circular orbit

Receiver Satellites:

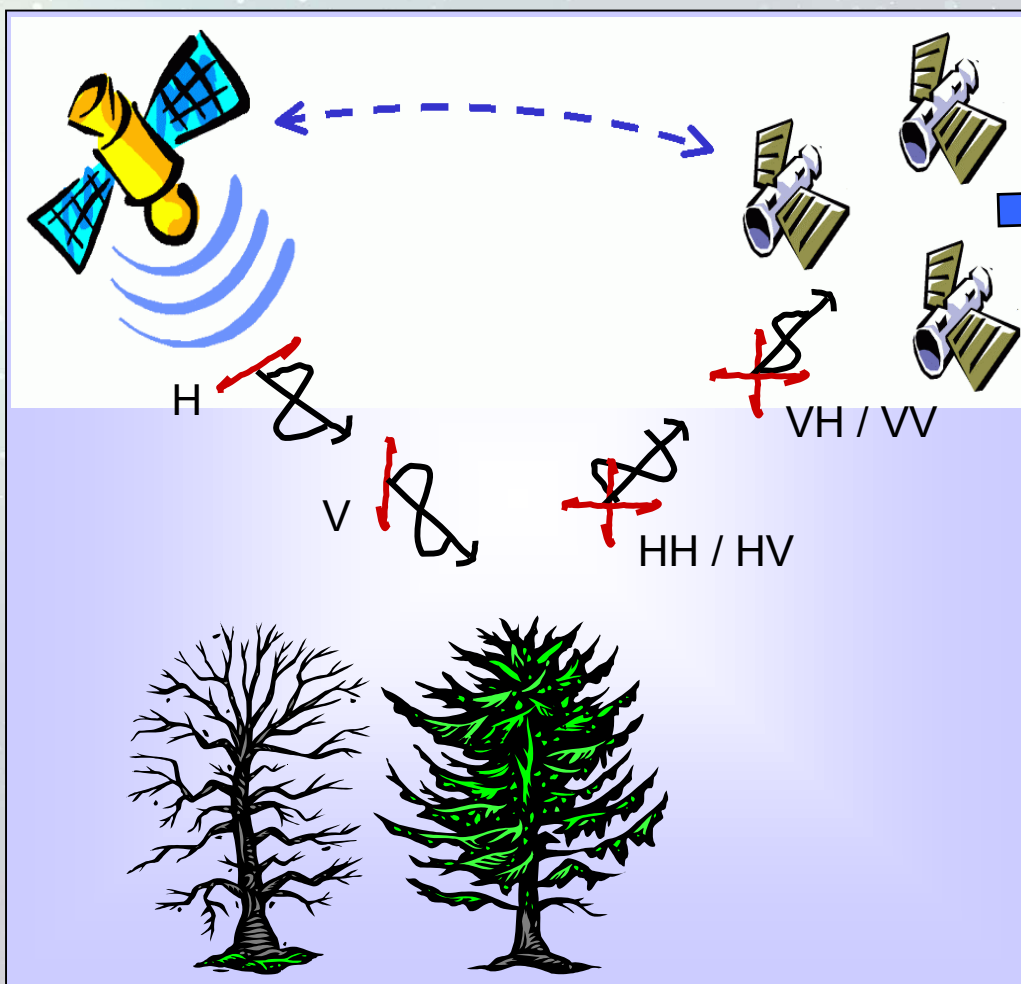
- elliptical orbits with equal eccentricities
- arguments of apogee differ by 120° for a Cartwheel with 3 satellites
- very stable maximum vertical baseline for all orbital positions
- relative movement of receivers may be approximated by CW-ellipse:



- coupling of along-track and across-track baselines

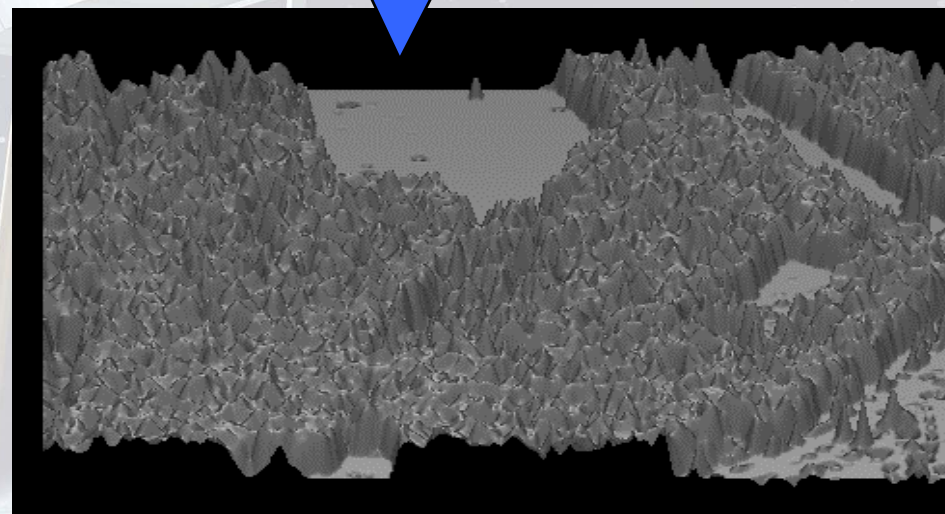


Polarimetric Interferometry with Passive Micro-Satellite Configuration



Inversion of Coherent Scattering Model
(e.g. Random Volume + Ground)

$h_V, \sigma, \phi_0, m_{1-3}$

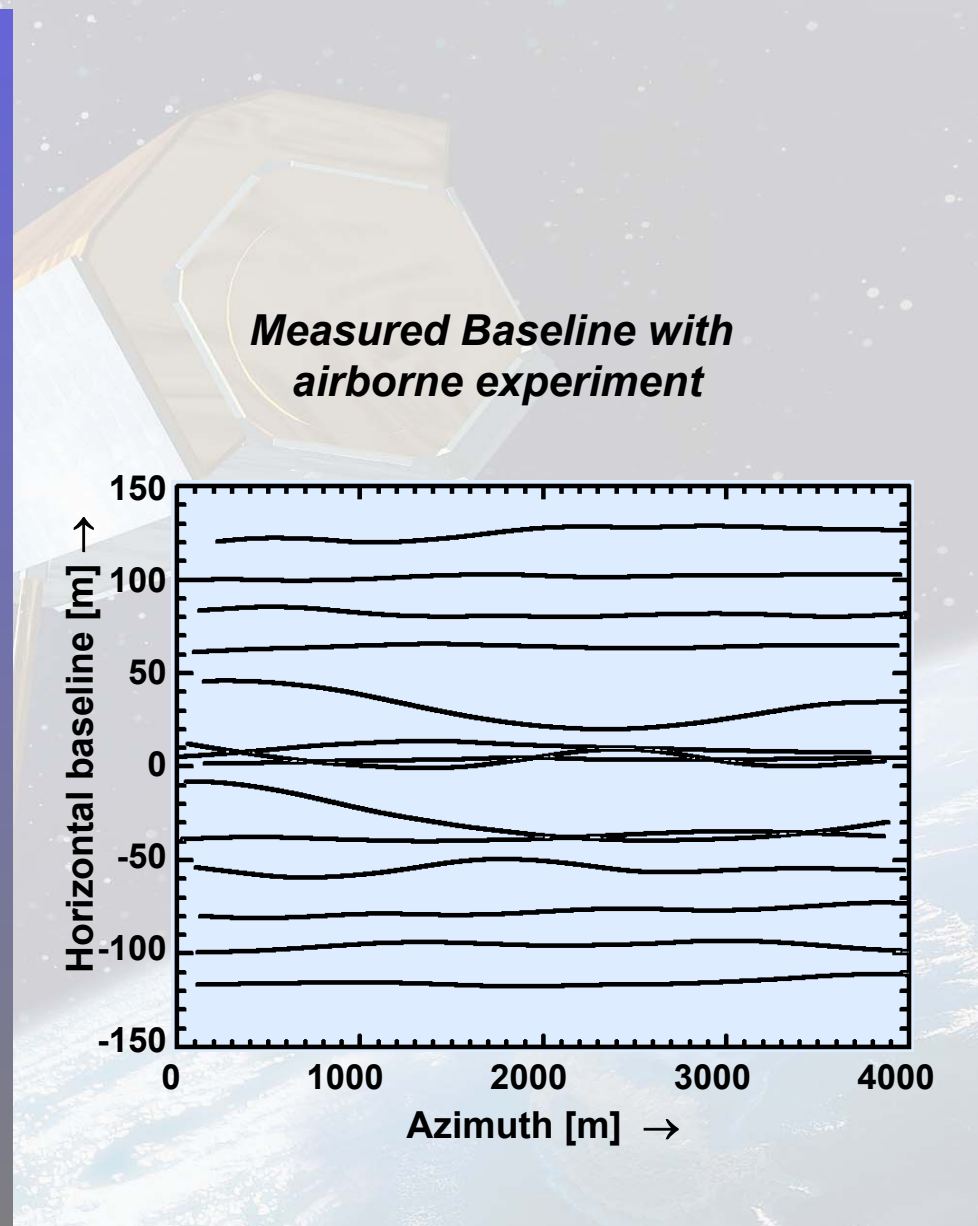
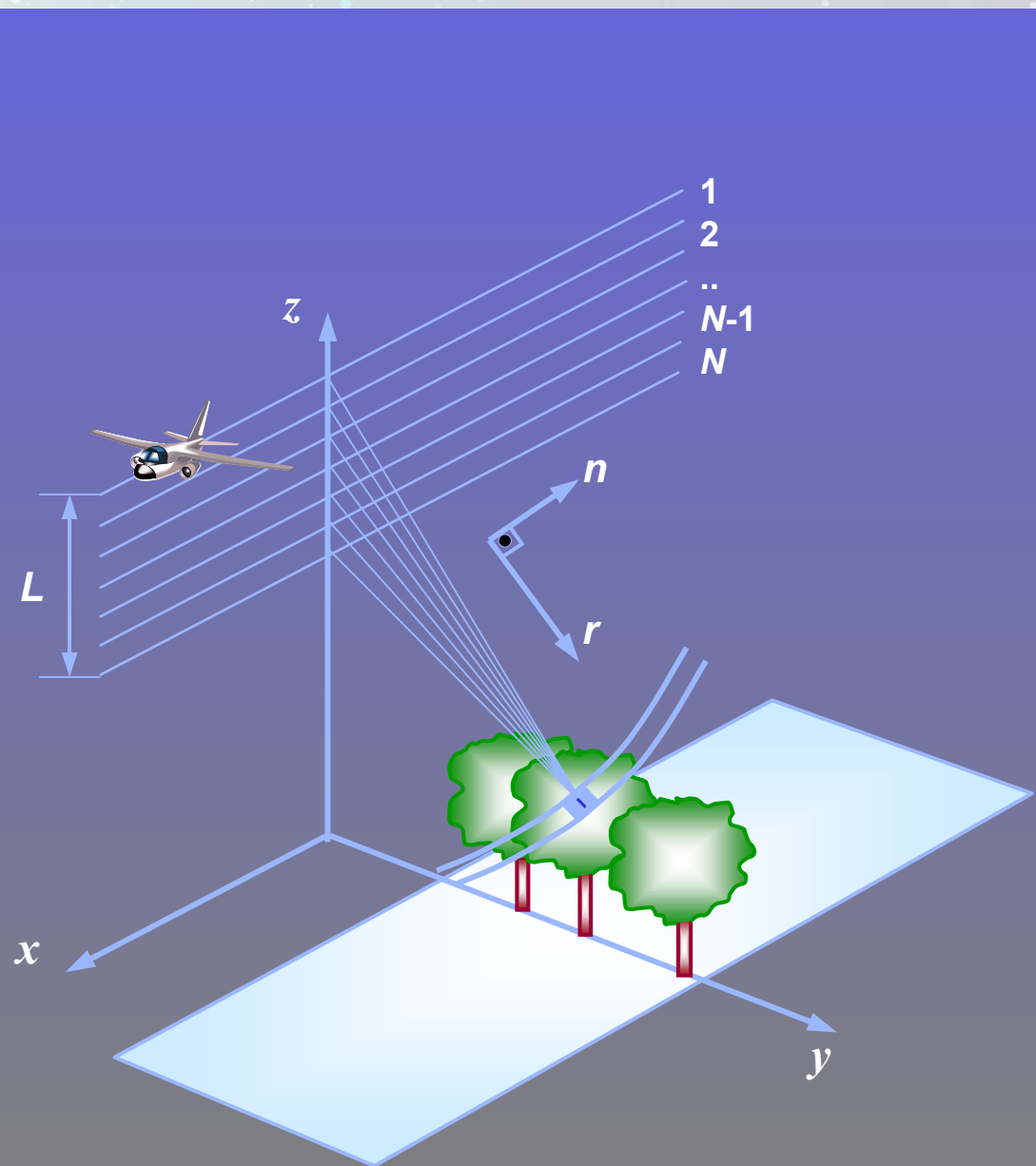


Estimated forest height

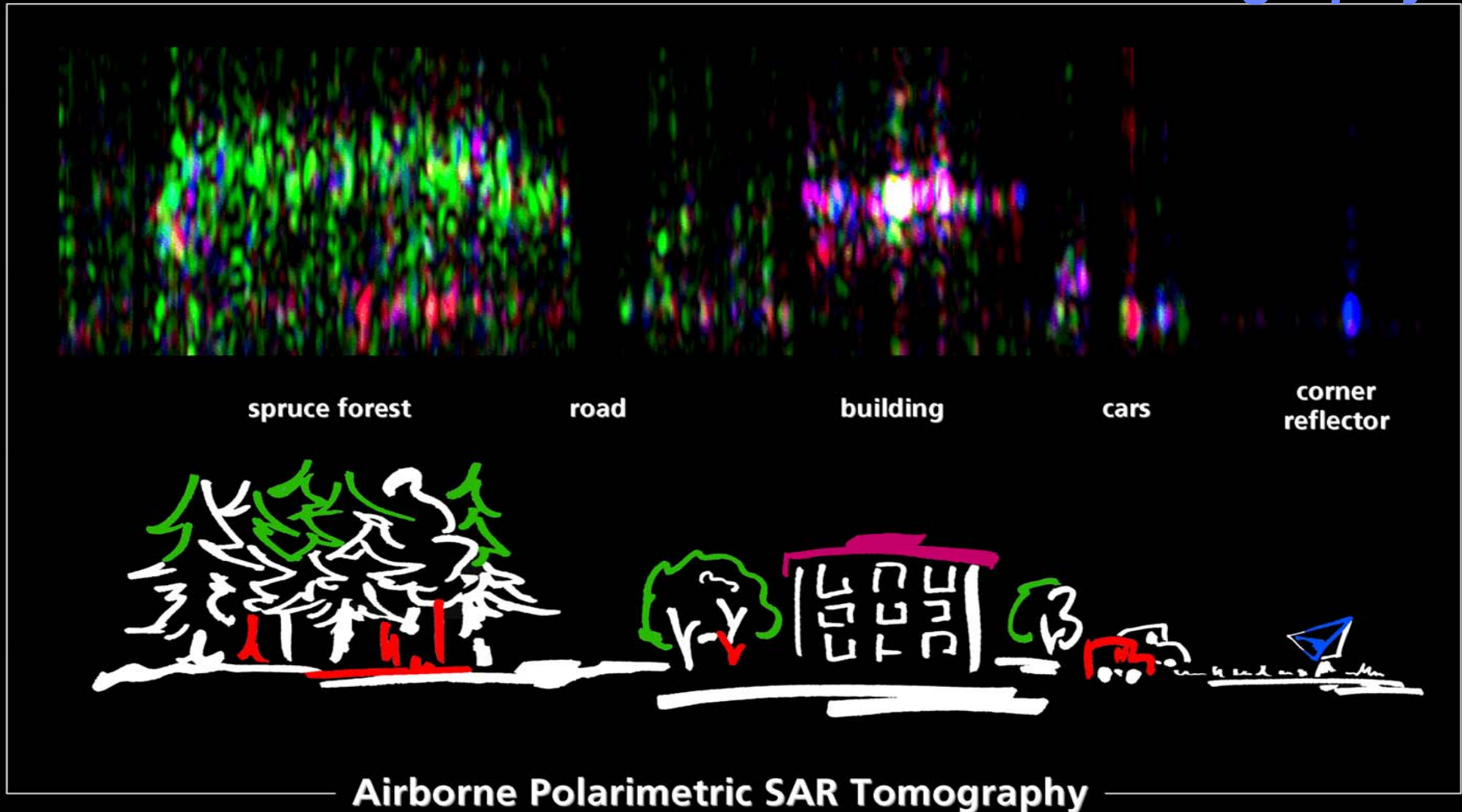


A 3D rendering of a satellite in space. The satellite has a white rectangular body and a large, gold-colored, hexagonal solar panel array. It is positioned above the Earth's horizon, which shows blue oceans and white clouds. The background is a dark space filled with numerous small white stars. Two horizontal blue lines frame the title text.

Multi-Baseline Approach



First Demonstration of Airborne SAR-Tomography



Upper image: Polarimetric color composite (L-band) of a tomographic slice in the height/azimuth-direction

■ HH+VV, ■ HH-VV, ■ 2*HV

Lower image: Schematic view of the imaged area

The background of the slide is a 3D rendering of a satellite in orbit. The satellite is a complex structure with a central body and several large, rectangular solar panel arrays extending outwards. It is positioned in the upper center of the frame, with its long boom pointing towards the bottom left. Below the satellite, the curved horizon of the Earth is visible, showing a mix of blue oceans and brown/green landmasses. The sky is a deep blue with numerous small white stars scattered throughout. Two horizontal blue lines are drawn across the slide, one above and one below the title text.

High-Resolution Wide-Swath

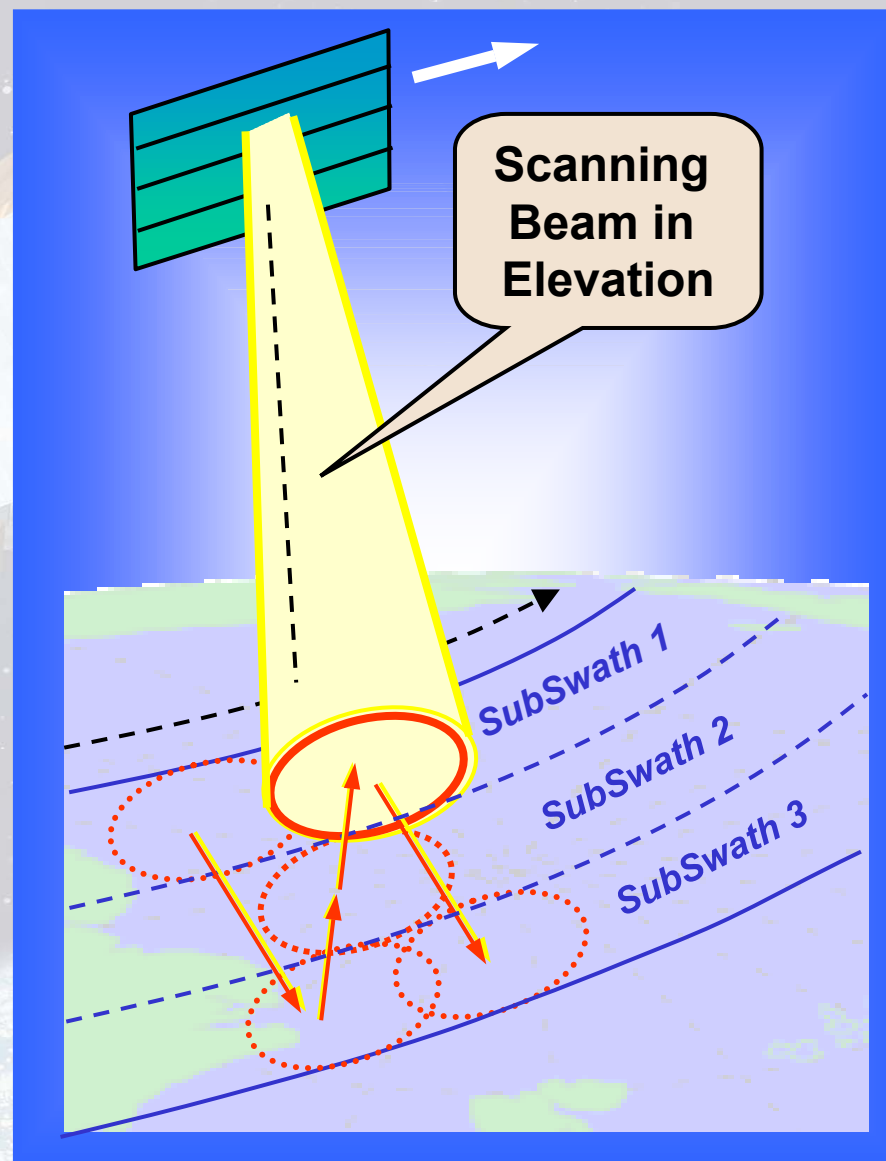
- **Standard approach for wide swath imaging**

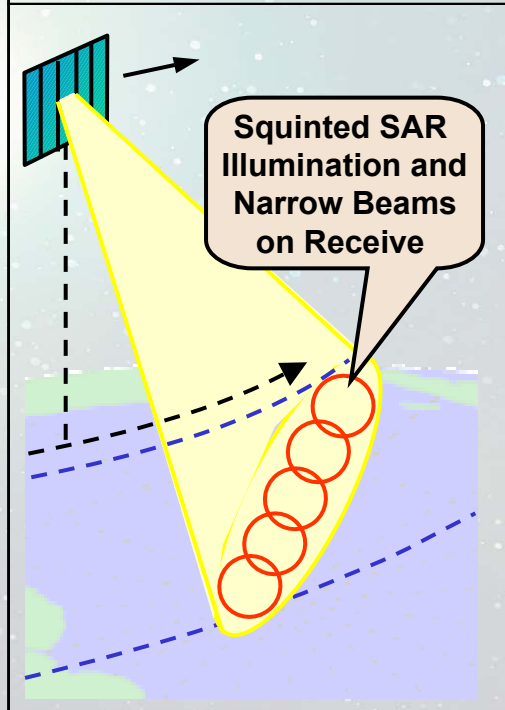
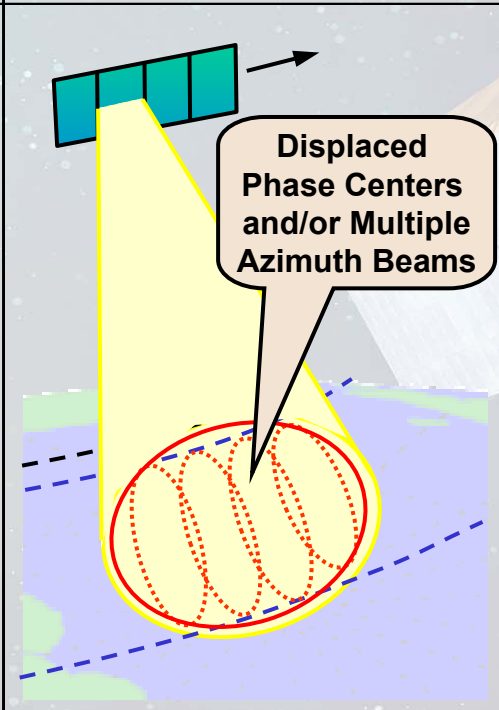
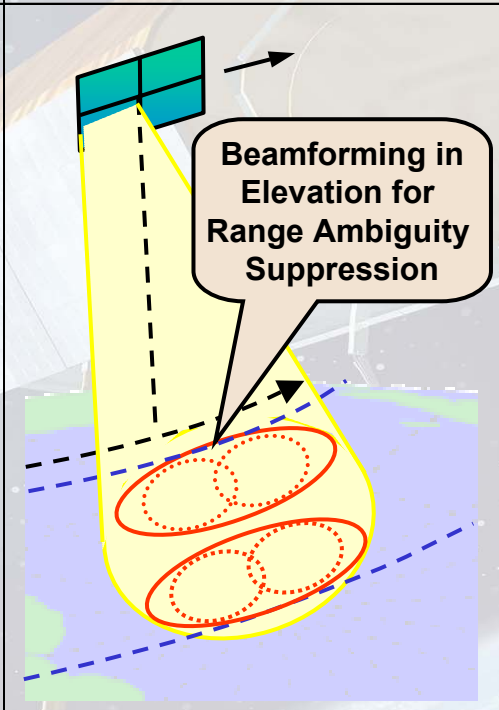
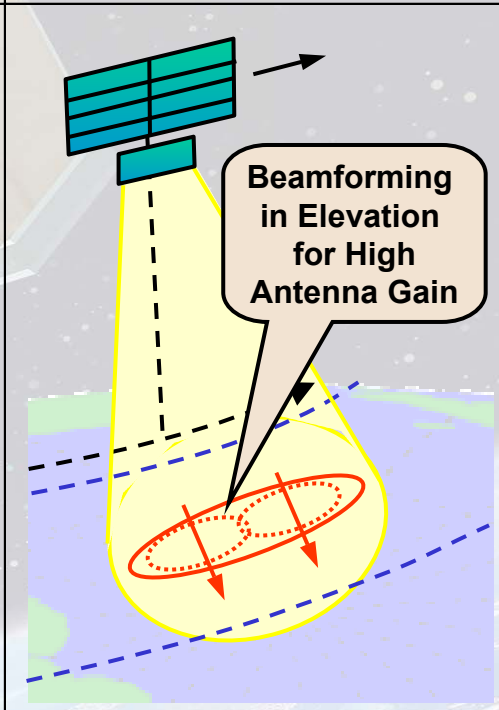
- **Basic idea:**

- Divide wide swath in multiple subswaths
- Scan subswaths in sequential order by appropriate beam steering in elevation (e.g. by using T/R modules)

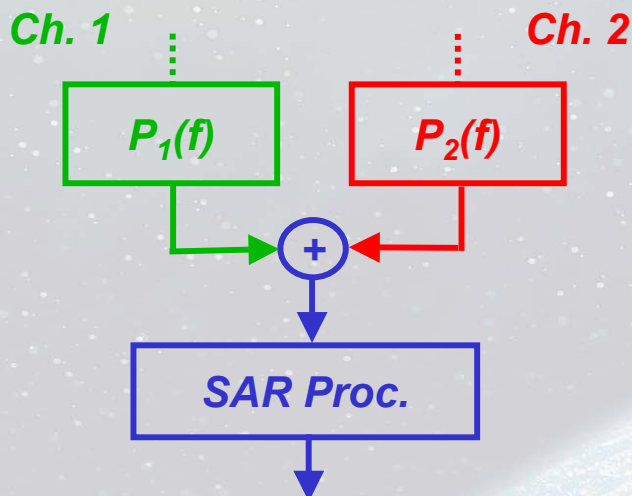
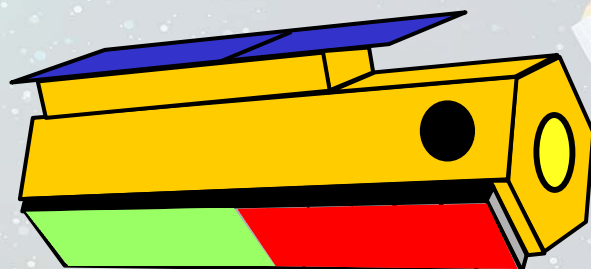
- **Shares full aperture synthesis time between multiple swaths:**

⇒ **Reduced azimuth resolution**



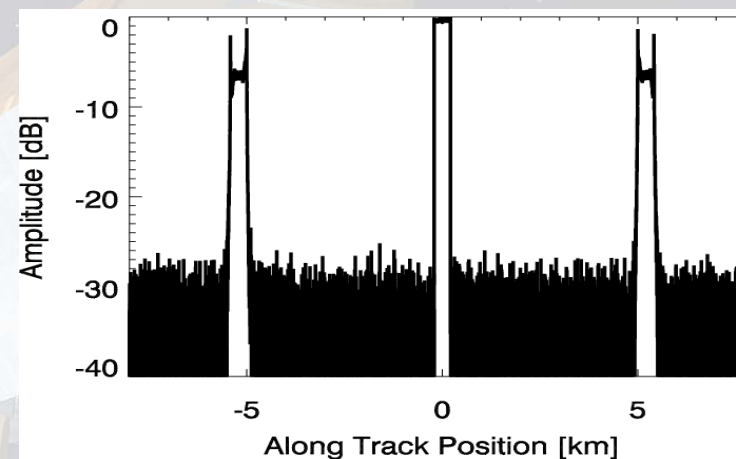
MultiBeam	DPC	Quad Array	HRWS
Claassen & Eckerman (1983) Jean & Rouse (1983)	Currie & Brown (1992)	Callaghan & Longstaff (1999)	Süss & Wiesbeck (2001) Younis & Wiesbeck (2002)
			
• almost constant incident angle across whole swath • wide swath imaging requires high squint	• good azimuth resolution • wide image swath • long receiving antenna • stringent PRF requirement	• good azimuth resolution • compact antenna • range gap in the middle of the swath	• good azimuth resolution • high receive gain by real-time beamforming • long receiving antenna

Wavelength	3.1 cm
Antenna Length (Tx)	2.4 m (4.8 m / 2)
Sub-Aperture Length (Rx)	2.4 m (4.8 m / 2)
PRF	3600 Hz
Velocity	7600 m/s
Channel SNR	20 dB

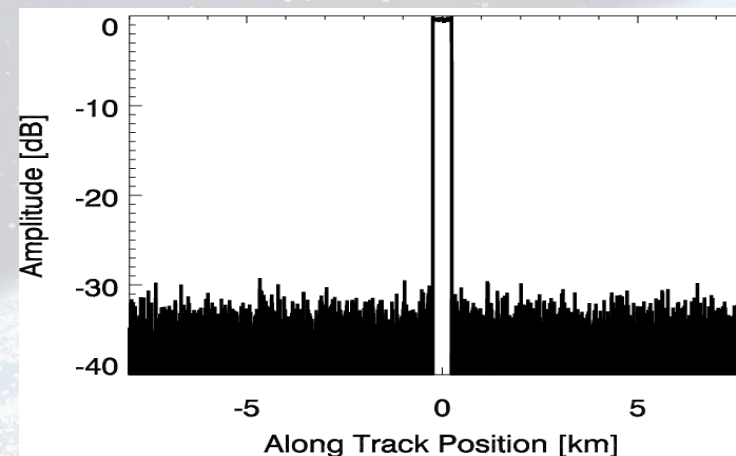


Response to Extended Scatterer

without reconstruction



with reconstruction





A 3D rendering of a satellite in space. The satellite has a central body with a large, flat, rectangular solar panel extended from one side. It is positioned above the Earth's horizon, which shows blue oceans and white clouds. The background is a dark space filled with numerous small white stars. Two horizontal blue lines frame the title text.

Satellite Constellations

Basic Idea:

- constant illumination by geostationary transmitter
- signal reception by multiple low-cost receivers

Receivers:

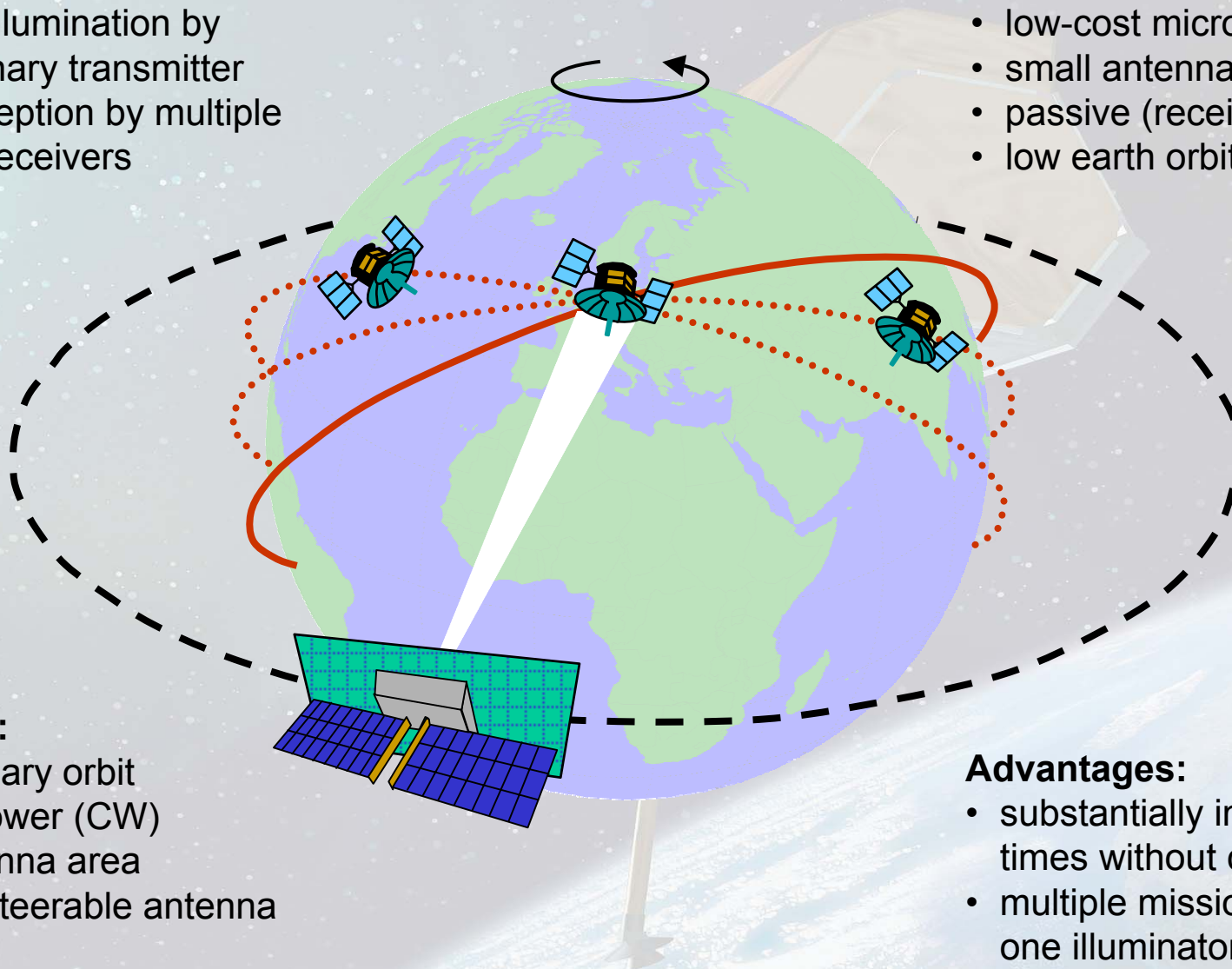
- low-cost micro-satellites
- small antennas
- passive (receive only)
- low earth orbit

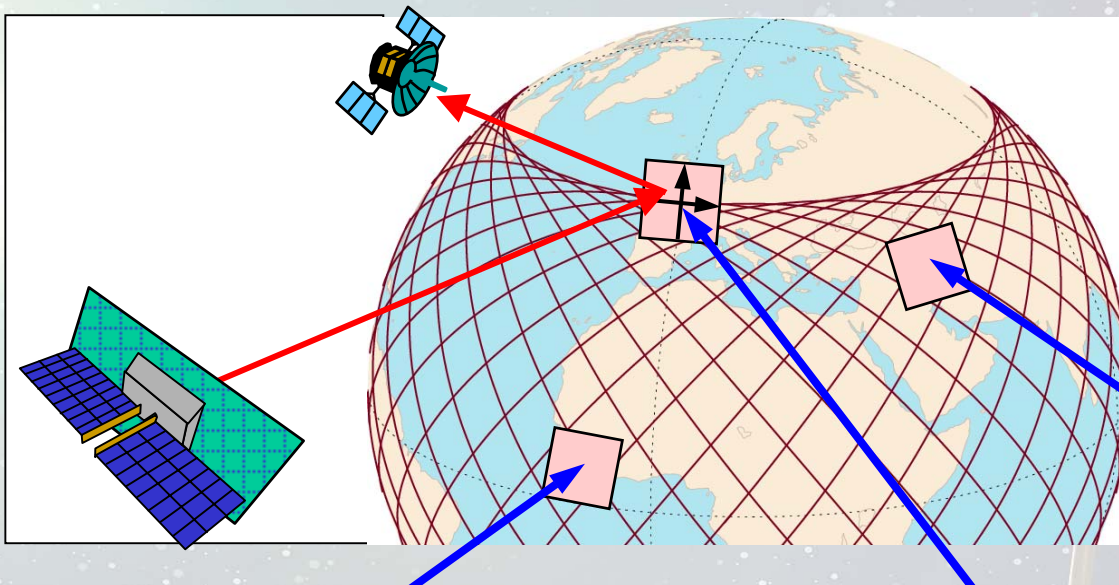
Illuminator:

- geostationary orbit
- high Tx power (CW)
- large antenna area
- optional: steerable antenna

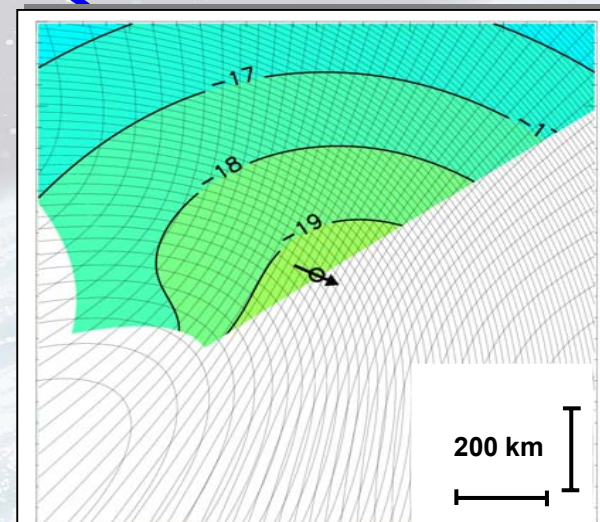
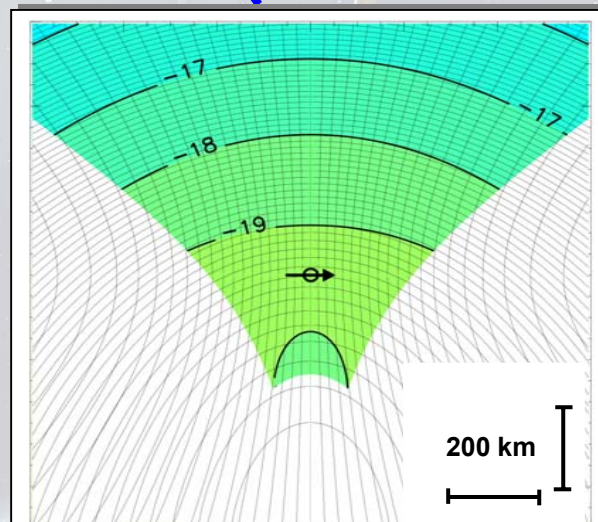
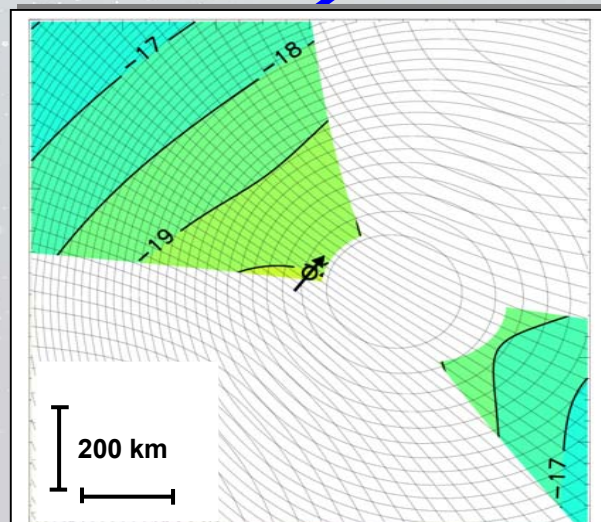
Advantages:

- substantially improved revisit times without cost explosion
- multiple missions may share one illuminator

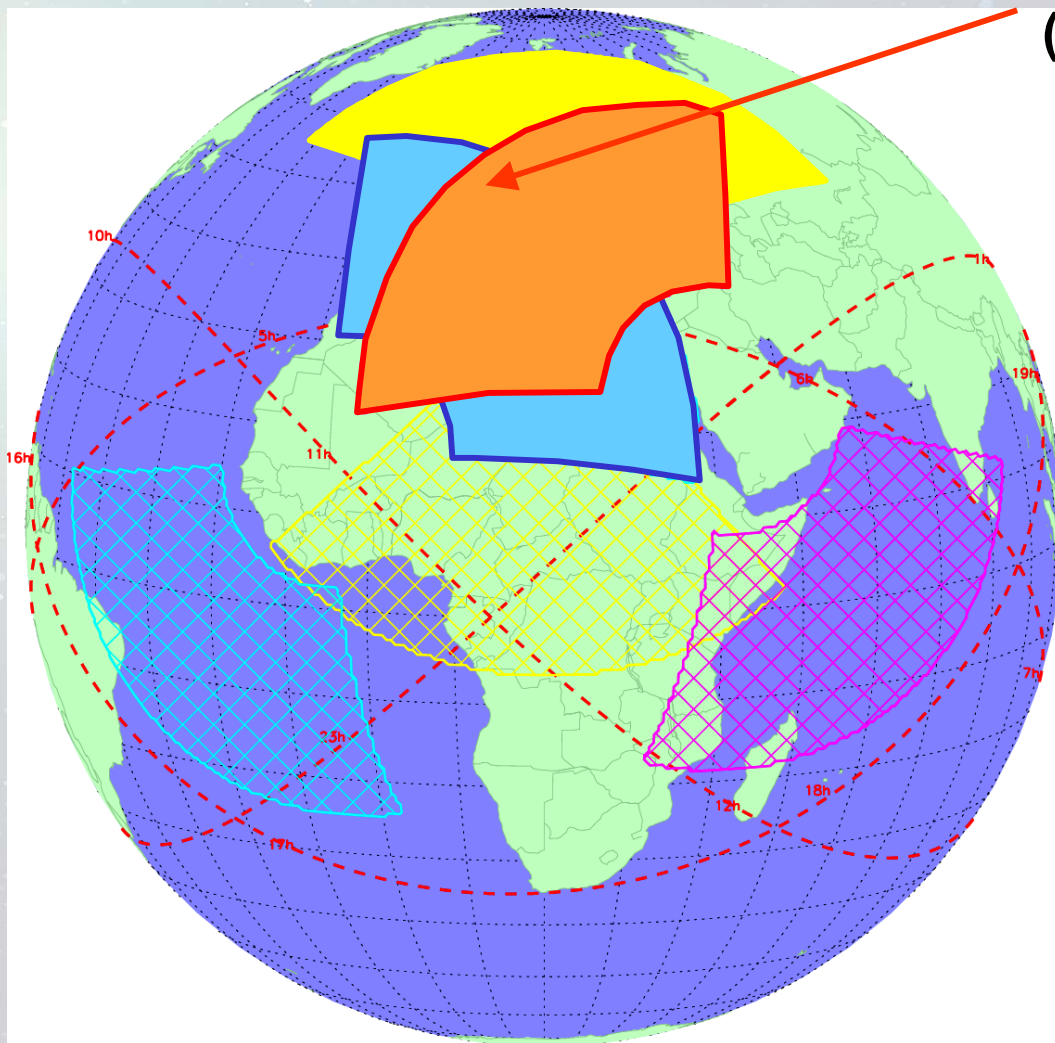




Wavelength	3.1 cm
Max. Bandwidth	300 MHz
Average Transmit Power	1000 W
Antenna Size Tx	100 m ²
Antenna Size Rx	6 m ²
Noise Figure + Losses	5 dB
Receiver Altitude	400 km
Ground Range Resolution	3 m
Max. Res. Cell Diameter	6 m



**3 revisits / day
(for 1 satellite)**



- Example: Orbit with an inclination of 35° and 5 cycles/day (altitude: ~ 8000km)
- Integration time for 3m azimuth resolution: ~ 10 sec
- NESZ: ~ -30dB (X-Band, $P_{TX} = 1 \text{ kW}$, $A_{TX} = 100 \text{ m}^2$)
- Huge simultaneous access area
- Long observation times enable change detection in one pass

MEO SAR Constellations are an interesting alternative to the geostationary illuminator concept

- ***Multi-static SAR***

- enables smaller and cheaper satellites
- provides more information and increased sensitivity
- benefits from advanced technologies

- ***Digital beam forming***

- makes full use of information at antenna array
- improves performance

- ***Satellite constellations***

- may share common illuminator
- symbiotic use of platforms (e.g. with Nav./Com.)
- allow for decreased revisit times